AD-778 192

AN/TPQ-31 PERFORMANCE: DURING EVALUATION AS A HOSTILE WEAPON LOCATOR

Raytheon Company

Prepared for:

Marine Corps

April 1973

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SECURITY CLASSIFICATION OF THIS PAGE (When Data Z-11.0)

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM				
1. REPORT NUMBER 2. GOVT ACCES LWL-CR-03P73B	SION NO. 3. RECIPIEN I'S CATALOG NUMBER AD - 778/92				
4. TITLE (and Subtitle)	5. TYPE OF REPORT & PERIOD COVERED				
AN/TPQ-31 Performance -	Final Report				
During Evaluation as a Hostile Weapon Locat	6. PERFORMING ORG. REPORT NUMBER				
7. AUTHOR(*)	8. CONTRACT OR GRANT NUMBER(s)				
Raytheon Company Equipment Division Wayland, Massachusetts 01778	Contract M00027-68-C-0047				
9. PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TASK				
HQ, United States Marine Corps Washington, D.C. 20380					
11. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE				
US Army Land Warfare Laboratory	April 1973				
Aberdeen Froving Ground, MD 21005	13. NUMBER OF PAGES 220				
14. MONITORING AGENCY NAME & ADDRESS(II different from Controlling	Office) 15. SECURITY CLASS. (of this report)				
	UNCLASSIFIED				
	15a, DECLASSIFICATION/DOWNGRADING SCHEDULE				
16. DISTRIBUTION STATEMENT (of this Report)					
Approved for public release; distribution unlimited					
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, If di	Sterent from Report)				
	D D C				
18. SUPPLEMENTARY NOTES	MAY 10 1978				
19. KEY WORDS (Continue on reverse side if necessary and identify by block	19. KEY WORDS (Continue on reverse side if necessary and identify by block number)				
NATIONAL TECHNICAL INFORMATION SERVICE U.S. Department of Comparise Springfull VA 20151					
20. AGSTRACT (Continue on reverse side if necessary and identify by block number)					
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AN/TPQ-31 RADAR SYSTEM SUPPORT PROGRAM

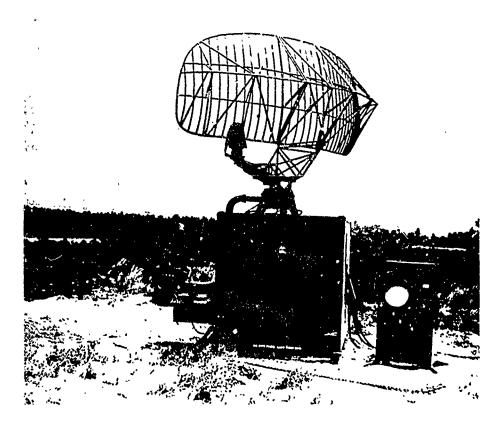
CLASSIFICATION:

Unclassified

Project Number:

Subject:

AN/TPQ-31 Performance, Report of



ABSTRACT

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CLASSIFICATION:

Unclassified

PROJECT NO.:

SUBJECT:

AN/TPQ-31 Performance, Report of

REFERENCE:

- (a) USMC Contract No. M00027-68-C-0047
- (b) U.S. Army Land Warfare Laboratory, Test Plan,
 December 1972

ANNEXES:

- A AN/TPQ-31 Radar Set Equipment Characteristics Radar Performance Parameters
- B Background Information
- C AN/TPQ-31() Radar Set Equipment Performance
 Objectives
- D Wang Computer Program
- E Dahlgren Test Data Analysis
- F Fort Sill Test Data Analysis

1. INTRODUCTION

a. <u>Purpose</u> - The primary purpose of this report is to provide a record of the performance of the AN/TPQ-31 during test firings, provided for the field evaluation, of similar type equipment being used as a hostile weapon locator. In addition, the report will also provide data for future comparison with the performance of the AN/TPQ-31's currently being middlified by Raytheon Company under Contract M00027-72-C-0098.

b. Description ..

(1) The AN/TPQ-31 Radar Set was designed to provide for direct interpretation by operators from a first round track of blips painted on a PPI scope by incoming mortar projectiles the location of the hostile weapons firing the

incoming mortar rounds. The AN/TPQ-31 Radar Set utilizes a doppler processor which eliminates stationary targets, and has a continuous 360 degree azimuth coverage at a range of 1 to 24 kilometers on targets of 0.015 sq. maters or larger. Target range and azimuth data is displayed on a 16 inch arimuth range indicator (PPI) for the operators. This indicator can be operated up to 100 feet from the radar set equipment shelter. The AN/TPQ-31 utilized by the Marine Corps during these firings included the 1967 modifications plus changes to the receiver chain to improve rain clutter and to the antenna system to increase its rotation speed from 15 RPM to 30 RPM. Additional details are contained in Annex A to this report.

c. Background -

- (1) In October 1967, the Raytheon Company received a quick reaction contract from the Marine Corps to provide a 360° counter mortar radar for use in Viet Nam. Ten AN/UPS-1 air search radars designed in the mid 1950's were provided to Raytheon by the Marine Corps for limited modifications that would permit the permit the permit the permit in the counter mortar role. Various features which may have been incorporated in a complete "design-for-the purpose" were of recessity not pursued. In February 1969 deployment of active units was started in Viet Nam. Careful siting of the radar system and constant vigilance on the part of the radar operator were found to be essential to the effectiveness of the system. Additional background details are contained in Annex B to this report.
- (2) Currently the Marine Corps has a limited modification program with Raytheon to update the 1967 version of the AN/TPQ-31 using modern technology in specific sections of the radar. The transmitter-receiver chain is being modernized and current solid state signal and target data processing techniques introduced. These modifications are expected to significantly improve target detection and processing with the provision of manual, semiautomatic and automatic hostile

weapon location readout. The data collected by this report will be used in the field tests of this improved AN/TPQ-31(). Additional performance objectives are contained in Annex C to this report.

2. DISCUSSION

- a. This report is limited to the performance of the AN/TPQ-31 during firing exercises at Dahlgren, Va. and Ft. Sill, Oklahoma.
- b. The objective of these firings was to demonstrate the feasibility of using radars similar to the AN/TPQ-31 as hostile artillery locators and/or designators, and to gain experience in their use for these purposes, in order that recommendations could be made for improving such equipment and for improving weapon locating methods. The philosophy of the firing tests was to progress through a series of tests beginning with projectiles of opportunity and culminating in firings which simulate as closely as possible actual operating conditions.
- c. The evaluation firings were divided into two phases; the first phase at Dahlgren, Virginia, and the second at Fort Sill, Oklahoma.
- d. During the evaluation, the AN/TPQ-31 was operated entirely by Marines. The Marine operational personnel were used to run the equipment, collect data, keep logs, and analyze the data.
- (1) For information purposes, the following is supplied on the eight Marine Personnel involved:

NUMBER	RATE	MOS	DUTIES
1	S Sgt	5934	MCDEC Project NCO
1	Gy Sgt	5934	Radar Technicians
2	L/Cpl	5934	Radar Technicians
4	Pvt	0842	Radar Operators

- (2) The project NCO and the team chief had prior experience with the AN/TPQ-31. The operators had no previous experience with the equipment.
- (3) Raytheon contractor personnel were available at the site for the purpose of assisting the Marine technician in repairs that might be beyond his experience.
- e. The firing point predictions listed in the data of Annexes E&F (Dahlgren and Ft. Sill, respectively) were arrived at through a backtrack procedure utilizing two methods:
- (1) The basic AN/TPQ-31 manual method involving map plotting. This procedure starts with plotting, in radar coordinates, the first and last position of the radar track on a map. The distance between the two points is measured and divided by 2(N-1) where N is the number of antenna scans over the time of flight. The result is then plotted backward along the line of flight established between the two previously plotted points and establishes the firing point. This point is then read out in map coordinates. The operator derived firing point locations in the accompanying data to this report were based on a 25,000 to 1 scale map with no mask angle correction being utilized.

A plotting pin approximately 1/30 inch diameter may lead to an error of approximately 50 meters. With five measurements or plots made as described above, an expected error would be $50\sqrt{5} = 112$ meters per coordinate, assuming no additional human error. Typically plotting errors in the range of 100-200 meters were noted.

(2) The second method is semi-automatic and uses a desk type exctronic calculator with the incorporation of a correction for mask angles.

In this method, backtrack firing point positions were calculated using a Wang Model No. 700A. The formulas used are:

$$X_{F} = R_{1}SINA_{1} - 1/2 \frac{\left(R_{2}SIN A_{2} - R_{1}SIN A_{1}\right)}{N - 1} - \frac{0.21\left(R_{2}SIN A_{2} - R_{1}SIN A_{1}\right)}{T^{2}}$$

$$(R_{1} TAN \phi - \Delta H)$$

$$Y_{F} = R_{1}COSA_{1} - 1/2 \frac{\left(R_{2}COSA_{2} - R_{1}COSA_{1}\right)}{N - 1}$$

$$- \frac{0.21\left(R_{2}COSA_{2} - R_{1}COSA_{1}\right)}{T^{2}} (R_{1}TAN \phi - \Delta H)$$

where

R₁ is first range

A₁ is first azimuth

R₂ is last range

A2 is last azimuth

 ϕ is mask angle

 $\Delta extsf{H}$ is radar HT - weapon HT at first detection

N is no. blips

T is a corrected time between first and last looks:

$$T = 2(N - 1) + \frac{A_2 - A_1}{3200}$$

Annex D gives the Wang program which was implemented to solve the above equations. The last term for X_F and Y_F above are mask angle corrections. The most important feature in using machine assists is that the additive error of the procedure is zero unless an error is made in the entry of a number. Another feature is the ease and speed and the elimination of plotting and reading map values.

A miniature Hewlett Packard calculator (HP-35) was also used to compute firing points and is believed to be a valuable time saving asset for such application. See Appendix 1, Annex D, this report for HP-35 program.

- f. There are three prime contributors to the backtrack prediction error:

 Antenna speed variation, range error and azimuth angle error.
- (1) Antenna speed variations are most likely to be caused by wind which results in an effect that repeats each 100°. This would have virtually no effect on either the backtrack or mask angle corrections. If the antenna ran at the wrong speed continuously the effect on backtrack would be self-can elling because the backtrack would now be based on the new spacing between the blips; however, the mask correction would suffer. Typically, for a speed error of 10%, which is unlikely, and 10 blips and a masking situation leading to a 300 meter correction, the error will be about 60 meters; however it is not suspected that speed error was significant for these tests.
- (2) In the 24 km range mode, the range gate width is 462 meters. Since the transmitted pulse is relatively narrow (approximately 100 meters wide) little gain is achieved by gate splitting; and the peak system range error is approximately 231 meters. A variable which is uniformly distributed between specified peak values has a standard deviation of $\beta/2\sqrt{3}$, where β is the peak to peak value of error. For this case, the standard deviation is:

$$\sigma = \frac{462}{2\sqrt{3}} = 133 \text{ meters RMS}$$

Thus, range error may be considered to have a one-sigma value of 133 meters and a probable error of approximately ±115 meters.

(3) Although no conclusive data is available, a probable readout error of ±0.5 degree is assumed to exist resulting from system non-linearities, and operator readout error; however, the Fort Sill data shows the possibility of readout errors nearly an order of magnitude greater which may be due to different operators aligning the electronic strobe line to different edges of the target point. It had been observed that under some conditions the strobe was placed at the lagging edge of the target paint instead of the leading edge. A study of the polar plots shows the amount of azimuth off. This situation can be observed in Fort Sill test 5 Seq 3, Test 7 Seq 3, Test 7 Seq 4, Test 13 Seq 3, and Test 15A Seq 2, which show strong biases one or the other side of the correct azimuth angle. See Appendix 1, Annex Febris report for polar plots.

g. Another factor that is considered to be most important in packtrack computations is mask angle corrections.

It has been recognized that a potential improvement in backtrack results is possible by correcting the computation of firing point arrived at in paragraph e. (1) above, according to the mask angle (elevation angle to a terrain feature) if any, which exists in the vicinity of the first sighting. The mask angle from the radar is determined by observing whether the computed firing point is in a clear line of sight from the radar or masked by a terrain feature of a determinable height. A close approximation to a parabolic correction was then computed for each corrected firing point axis (S_X and S_Y) equal to:

$$S_{Y_{i}} = \frac{0.21 (X_{2} - X_{1}) (R_{1} TAN \phi - \Delta H)}{T^{2}}$$

$$S_Y = \frac{0.21 (Y_2 - Y_1) (R_1 TAN \phi - \Delta H)}{T^2}$$

vdiere

 ϕ is the mask angle and

All is the height of radar minus the height of the backtracked point.

The results of applying mask angle correction to the operator manual map plotted firing points showed an average improvement of 37.1 percent in virtually all cases of mask angle as tabulated below. Further mask angle correction showed a marked improvement as the mask angle increased. (See Table I).

- h. <u>Dahlgren Tests</u>. The AN/TPQ-31 arrived at Dahlgren on 20 November 1972 and was fired up and operated at the main range site until 29 November. Targets of opportunity were used during this period to familiarize the operators with the operation of the radar system. The firings at Dahlgren were primarily used for equipment performance checkout and operating procedures refinement. Therefore, only the firings where data of significant nature was collected have been processed, namely, firings on 6 and 7 December. Firing data collected on the other dates are included in this report for record purposes. The firings where the radar painted a high frequency of prints, and where data (such as location of firing point, type of weapon, and QE were provided) was considered significant enough to be processed.
- (1) The firing data collected at Dahlgren is contained in Annex E which addresses the reduction of the data collected on 6 and 7 December. The firings on these dates were divided into four groups depending on their QE, and then miss distances were computed. For the 59 rounds processed, with bias removed, a CEP of 105.2 meters was attained on a 81 mm mortar located approximately 6830 meters from the radar.

EFFECT OF MASK ANGLE CORRECTION

% Improvement	59.5%	1	30.8%	63.7%	79.3%	17.7%	28.5%	37.7%	21.2%	56.5%	9.3%	19.0%	22.2%	44.0%	13, 1%	24.0,	ı	r	67.4
(Included Mask Corr.). Machine Miss Dist.	537 meters	r	1173	787	230	836	630	820	1584	449	298	3032	1324	532	591	332	467	,	136
(No. Mask Angle Corr.) ask 1 Deg Map Plotted Miss Dist.	1293 meters	ε	1693	2163	1106	1015	880	1316	2008	1030	955	3741	1700	950	089	440	450	r	416
() Mask 1 Deg N	1	1	1	, - -	1	p=4	-	1	1	1	0.5	0.5	1	0.5	0.2	0.2	0.2	0.5	0.5
Weapon	105 mm How	105 mm How	105 mm How	105 mm How	105 mm How	105 mm How	105 mm How	105 mm How	105 mm How	155 mm How	105 mm How	105 mm How	105 mm How	105 mm How	155 mm How	155 mm How	155 mm How	8" How	8" How
Test No.	5/1	2/5	5/3	5/4	5/5	7/1	7/2	7/3	7/4	∞	11/1	11/2	12/1	12/2	13/1	13/2	13/3	16/1	16/2

Table I

37.1%

- (2) During the Dahlgren test program, the radar functioned exceptionally well. A few minor problems were experienced, but none affected the operational readiness of the radar. The operational performance during the actual firings was 100%. It is to be noted that the antenna pedestal of the radar had accumulated 3,000 hours of life to t prior to being deployed to Dahlgren.
- i. Forts Sill Tests The firings at Fort Sill began on 23 January 1973 and ended on 7 March 1973. Each firing consisted of a number of sequences, with a specified number of rounds fired during each sequence. Data on firings 2 through 18 were collected with the AN/TPQ-31. The weapons used during these firings consisted of the 81 mm and 4.2 mortars, the 105 mm, 155 mm and 8.0" howitzers and a 175 mm gun. Predicted firing points were derived from the collected data.

THE PARTY OF THE PROPERTY OF T

(1) The values of accuracy for the predicted firing points were processed at the radar site manually and later by machine. It is difficult to compare both methods, in that, the true firing points used, for computation, were different for each of the methods. The accuracies indicated for the types of weapons, by each method against its assumed true firing point, are as follows:

OPERATOR CALCULATED

Weapon	No. Rounds (Detected)	Avg. Miss Distance All Rounds	No. Sequences (Detected)	Miss Distance Best Round each Sequence	Miss Distance Best One Round
81 mm	20	237 m	2	90	80
4.2 mm	86	498 m	17	369	20
105 mm	160	1538 m	18	770	100
155 mm	46	431 m	7	270	20
811	23	720 m	2	240	140
175 mm	10	280 m	1	140	60

MACHINE CALCULATED

Weapon		Avg. Miss Distance All Rounds	No. Sequences (Derected)	Miss Distance Best Round each Sequence	Miss Distance Best One Round
81 mm	20	244 m	2	150	148
4.2 mm	86	601 m	17	487	51
105 mm	160	1252 m	18	655	135
155 mm	46	331 m	7	259	67
811	23	476 m	2	312	93
175 mm	10	155 m	1	128	128

(2) The machine calculated firing point accuracies can be further refined by screening out non-valid erroneous tracks such as on weapons firing from points not associated with the test program. In eliminating these tracks, the predicted firing point accuracies listed below were attained. See Appendix 2, Annex F, this report, for further discussion on erroneous tracks.

Weapon	1. To. Rounds (Detected)	Avg. Miss Distance All Rounds	No. Sequences (Detected)	Miss Distance Best Round each Sequence	Miss Distance Best One Round
81 mm	20	244	2	150	148
4.2"	86	601	17	487	51
105 mm	144	905	15	415	135
155 mm	42	320	7	259	67
811	23	476	2	312	93
175 mm	10	155	1	128	128

(3) The firing sequences explored the capability of the system beyond as well as within the designed operating limits. Unfortunately, a very high percentage of the rounds were fired at QE's that placed their track above the antenna

beam, at velocities that fell in the blind speed notch, and at distances where the size of the target was not compatible with the detection range of the radar. A situation could have prevailed where all the rounds were out of the AN/TPQ-31 antenna beam and also in its blind speed notch, wherein very little data would have been collected. In this context the 105 mm sequences were examined and fell into five categories:

(a) Rounds not detected (radar inoperative) 1 Valid rounds (correct projectile tracked) initial velocity of (b) trajectory detectable Valid rounds, but initial velocity of trajectory not detectable (c) 10 Valid rounds, but trajectory above beam and initial velocity (d) not detectable (e) Questionable rounds (wrong velocity for scheduled refereces) 1 see Appendix F-2-1

Thus, four cut of 23 of the 105 mm howitzer sequences are expected to yield meaningful results, mainly because of the crucial dependence on seeing the initial part of the trajectory. This helps account for the poor results of the 105 mm rounds where only tests 7/1, 7/3, 9/2 and 10/1 wer usable rounds within the radar design parameters. Considering these sequences, the average miss distance for the 105 mm rounds becomes 733 (4 sequences) meters vice 1139 meters (22 sequences).

(4) The AN/TPQ-31 has the capability for detecting projectiles fired anywhere within 24 km provided that various parameters of the round fall within the design ratings of the radar. The probability that a "detectable" round is actually detected is a significant performance indication. This probability is available from the accumulated data by noting which rounds had detection curves which did not break through the threshold at any point. The reasons for this are examined in Appendices 3 and 4, Annex F but generally the projectile may have been in the blind velocity notch, or was too distant, or too high above the radar beam. Also, those rounds for which the radar was inoperative are not counted here.

Total No. of Theoretically Detectable

Rounds Fired 353

Total No. of Above Rounds Seen 306

Probability of Detection of all "Detectable"

Rounds for All Fort Sill Data $\frac{306}{353} \times 100 = 87\%$

Actual detection without regard for theoretical detectability on each weapon was shown to be:

Weapon	Sequence with Detections	Rounds Scheduled	Rounds Detected	Rounds Missed (No. Detec.)	Rounds Missed (Radar Problem)	% Detected*
81 mm	2	20	20	-	-	100
4. 211	17	178	86	9	83	90
105 mm	18	263	160	80	23	67
155 mm	7	54	46	-	8	100
811	2	50	23	27	-	46
175 mm	1	73	10	55	8	15

^{*}Percentages do not reflect rounds missed due to radar problems.

The worst case was the 175 mm gun which, due to its high muzzle velocity and long range, placed shells too high for the beam, too distant or in the blind speed notch on numerous firings.

(5) Apart from the problems recorded on data sheets in Appendix 5, Annex F, the AN/TPQ-31's performance record was satisfactory. It is difficult to foresee a problem such as the one experienced with the antenna. It had accumulated 3500 hours of fail-free operation at the time the problem was encountered. It is also felt that the Short Range Console problem would not have occured had more reliable power generators been available.

3. CONCLUSIONS

- a. The majority of rounds fired during the tests were outside the designed operating limits of the AN/TPQ-31.
- b. The following processing and/or inherent system errors can adversely.

 affect weapon firing point determinations:
 - (1) Manual plotting technique errors 112 meters per coordinate
 - (2) Probable range error ±115 meters
 - (3) Probable azimuth error ±0.5 degree
- c. Under the prevailing test conditions the AN/TPQ-31 attained the following firing point accuracies and percent of detections:

Weapon	No. Rounds	Miss Distance	% Detection
81 mm	59	105 meters (Dahlgren)	100
4.2"	178	542 meters (Fort Sill)	90
105 mm	263	956 meters (Fort Sill)	67
155 mm	54	361 meters (Fort Sill)	100
811	50	476 meters (Fort Sill)	46
175 mm	73	155 meters (Fort Sill)	15
81 mm	20	244 meters (Fort Sill)	100

- d. When applicable, the introduction of mask angle data into the back track computations significantly improves the accuracy of predicted firing points.
- e. The bias errors resulted primarily from an inability to accurately boresight the radar antenna during azimuth orientation.
- f. The AN/TPQ-31 detection performance was very successful against mortar type rounds, for which it was designed, and it had a limited detection success against artillery type rounds.

g. The data collected in this report is suitable as a basis for developing a Field Test Plan for evaluation of the AN/TPQ-31().

4. RECOMMANDATIONS

- a. That the data in this report be used as a basis for developing the AN/TPQ-31() Field Test Plan.
- b. That the majority of rounds to be fired during the field test of the AN/TPQ-31() be scheduled to fall within its design limits in order to provide a meaningful evaluation.
- c. That a horesight telescope be provided in order to permit accurate azimuth orientation which will help reduce the high frequency of bias errors encountered during the firings.
- d. That survey data on the true location of the firing weapons during AN/TPQ-31() field evaluation be made readily available to the test team upon completion of each firing sequence.

Annex A - (AN/TPQ-31 Equipment Characteristics - Radar Performance

Parameters) to Report of AN/TPQ-31 Performance during

Evaluation as a Hostile Weapon Locator

AN/TPQ-31 RADAR SET EQUIPMENT CHARACTERISTICS

SYSTEM CHARACTERISTICS

- 1. Total weight, Radar Set AN/TPQ-31:
 - a. Shelter: 3256 lbs.
 - b. Helicopter transport pallet A: 977 lbs.
 - c. Helicopter transport pallet B: 1515 lbs.
- 2. Shelter volume: 515 ft³
- 3. Input power requirements:
 - a. With air conditioners: Three-phase, four-wire, 400 Hz, 208/120-volts, 25-kW, power factor 0.9 lagging.
 - b. Without air conditioners: Three-phase, four-wire, 400 Hz, 208/120-volts, 10-kW, power factor - 0.9 lagging.

ANTENNA

- 1. Type: Cosecant-squared.
- 2. Dimensions:
 - a. Horizontal aperture: 15 feet, 9 ± 2 inches.
 - b. Vertical aperture: 6 feet ± 2 inches.
- 3. Beamwidth:
 - a. Horizontal: 3.7 degrees.
 - b. Vertical: 11 degrees.
- 4. Sidelobe and back radiation:

	Side	Back
Azimuth	25 dB	30 dB

ANTENNA (Cont.)

- 5. Gain
 - a. Greater than 27 dB over a frequency range of 1300 to 1350 MHz.
 - b. Not less than 26 dB over a frequency range of 1250 to 1300 MHz.
- 6. Scan rate:
 - a. Azimuth: 30 rpm
 - b. Elevation: Tilt -2 to +5 degrees (manual adjustment).
- 7. Drive system:
 - a. Continuous search operation: Servo-operated dc motor through a gear train having a reduction ratio of 233.3 to 1.
 - b. Manual searchlight operation: Manual rotation of handwheel.

RADAR TRANSMITTER T-1111/TPQ-31

- 1. Frequency Range: 1250 to 1350 MHz.
- 2. Wavelength:
 - a. 24 cm at 1250 MHz
 - b. 22.2 cm at 1350 MHz
- 3. Transmitter and duplexer tube types:
 - a. TR (V1851 type 7166/MA 3770) and ATR (V1852, V1853, and V1854 RCA type 8270926-1) gas-filled tubes in duplexer.
 - b. Magnetron (V204 type QK 358), gas-filled thyratron (V201 type HY-15), hard vacuum diode (V203, type 561), and gas-filled diode (V205, type KU52).
- 4. Average power: 1.0 kW nominal
- 5. Peak power: 1.0 MW nominal

RADAR TRANSMITTER T-1111/TPQ-31 (Cont.)

- 6. Pulsewidth:
 - a. 0.7 microseconds at 1600 PRF
 - b. 4.2 microseconds at 267 PRF
- 7. Pulse repetition rate frequency: 1600 or 267 pulses per second (pps)
- 8. Power requirements for operation:
 - a. Input system trigger: 40-V, positive pulse
 - b. Filament voltage: 6.3 Vac
 - c. Magnetron filament transformer: 130 Vac
 - d. Magnetron tuning: 28 Vac
 - e. Pulse generator: 350 Vdc
 - f. High B+: 1000 Vdc
 - g. Magnetron modulator plate power: 8500 Vdc

RADAR RECEIVER R-1585/TPQ-31

- 1. Stalo, five crystal-controlled frequency channels
- 2. Noise figure (including duplexer): 9 dB
- 3. Intermediate frequency: 30 MHz
- 4. IF bandwidth:
 - a. 1.5 MHz in the 20- and 40-mile ranges
 - b. 0.45 MHz in the 275-mile range
 - c. 1.5 MHz in the HFSR mode
- 5. Normal video minimum discernible signal (mds): 104 dBm minimum
- 6. Coherent video minimum discernible signal (mds): 100 dBm minimum

INDICATORS

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1. Azimuth-Range Indicator IP-954/TPQ-31

a. PPI scope:

10-inch crt, type 10WP7

b. A-scope:

3-inch crt, type 3SP1

c. Range scales:

20, 40, and 275 miles

2. Azimuth-Range Indicator IP-953/TPQ-31

a. PPI scope:

16-inch crt, type 16M59P33M

b. Range scales:

0 - 12 and 0 - 24 km (1.54 - 6.47 and 0.54 - 12.95 nmi)

SIGNAL DATA CONVERTER-SYNCHRONIZER CV-2578/TPQ-31

1. Range-gated doppler processor

2. 52 range-gated signal detectors

3. Synthetic video outputs to both indicators

AN/TPQ-31 RADAR PERFORMANCE PARAMETERS

RADAR SENSITIVITY

The received signal to noise ratio of a radar target observed for T_{o} sec is:

REC SNR =
$$\frac{(P_{TR-PK}) \quad \tau \quad (PRF) \quad G^2_{ant} \quad \lambda^2 \quad T_o \quad \sigma_T}{(4\pi)^3 \quad R^4 \quad KTF \quad L_R L_T}$$

If the required detection SNR is S, and processing loss is D, (departure), then

REQ REC SNR = (S) (D) for detection

Solving for target size:

$$T = \frac{(S) (D) (4\pi)^3 R^4 KTF L_T L_R}{(P_{TR-PK}) (\tau) (PRF) G^2 \lambda^2 T_o}$$

S (Prob Det = 0.8, 1 F.A. per scan)

KT

204

F (9 dB assumed) G_{ant}^{2} (G = 27.5 dB) λ^{2} (λ = 23 cm)

12.6 $(4\pi)^{3}$

$$R^4$$
 ($R_{max} = 24 \text{ Km}$) 175.2
 $L_R L_T$ 2
 P_{TR} (10⁶W) 60
 τ (0.7 x 10⁻⁶) 61.55
PRF 32.05
 T_o (21 millisec) 16.8
D 11.5

Total = -18.3 dB =
$$\sigma_{T}$$

 σ_{T} = 0.015 sq meters
@ 24 Km

The radar cross sections (RCS) of various projectiles from 81 mm to 155 mm have been found to average about 0.018 sq meters. Applying a case I fluction loss (at Pd = 0.8) of -5 dB gives an equivalent size of nominally 0.005 sq meters required for satisfactory detection sensitivity.

Assuming all radar parameters at the design value, and a target on the nose of the beam, the maximum detection range for a 0.005 sq meter target is

$$R = \sqrt{\frac{0.005 \text{ sq m}}{0.015 \text{ sq m}}} \times 24 \text{ Km} = 18.6 \text{ Km (see fig. 1)}$$

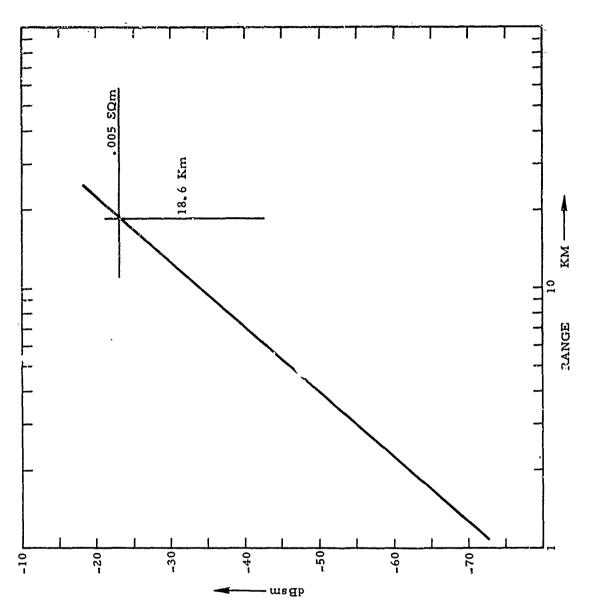


Figure 1. AN/TPQ-31 MDS vs. Range

Antenna Pattern

Fig. 2 shows the vertical gain for various degrees of antenna tilt. Fig. 3 shows the coverage for a target of 0.005 sq meters based on e above pattern and the system parameters presented earlier.

Velocity Notch

Fig. 4 shows a plot of velocity notch versus speed. As is characteristic of single PRF doppler radars, doppler velocities at or near multiples of the PRF are attenuated by the clutter filter. The amount of attenuation is a function of velocity and may be read directly from the plot of Fig. 4. The wide notch is used for particularly severe clutter or weather.

Sensitivity vs Time Control (STC)

STC is a technique to reduce the system gain for targets at short range to reduce the effect of clutter which is most bothersome at low ranges. The STC gain variation produces most attenuation (60 dB) at zero range and is reduced by approximately the fourth function of range until full gain is achieved at about 6 Km. The effect of STC, when considering a target in the maximum gain portion of the beam, is to hold the required minimum target to an approximately constant value while cutting clutter by an amount which is an inverse function of range. However, a shortcoming is that most desired targets which are at short range are also at high angles where the beam coverage has low gain. The STC in this case only makes matters worse.

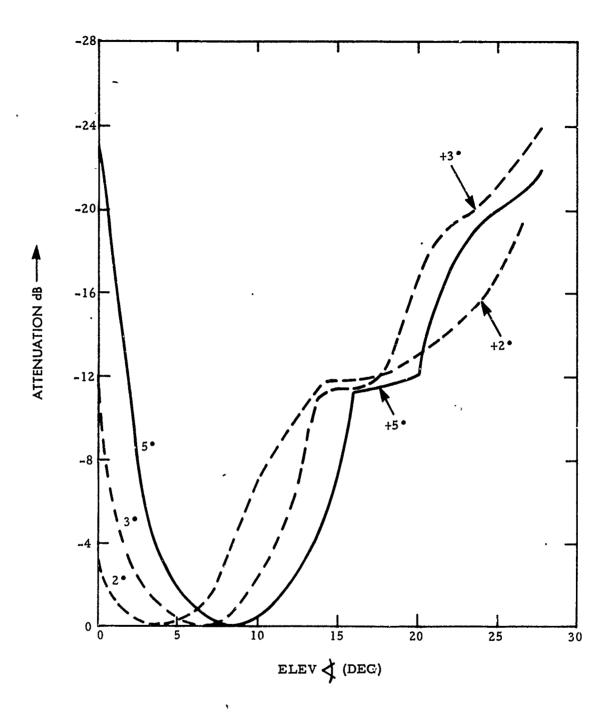


Figure 2. AN/TPQ-31 Antenna Vertical Pattern

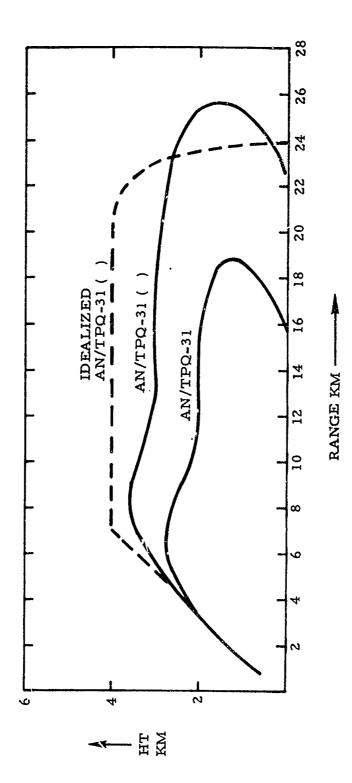


Figure 3. Radar Coverage for σ_{T} = 0.005 sq meter (no STC)

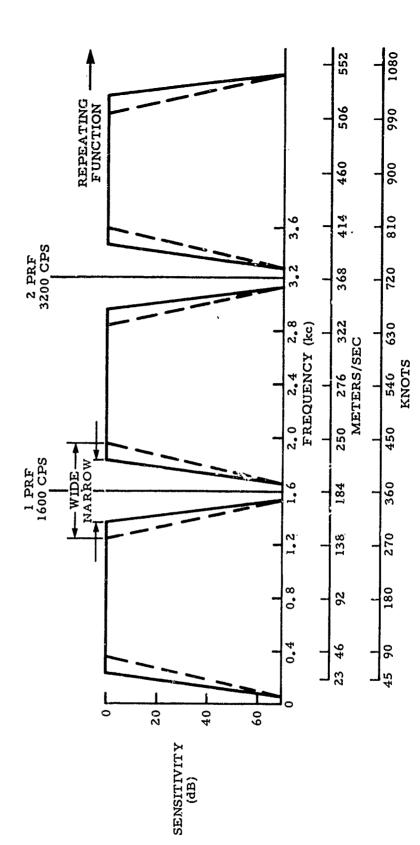


Figure 4. AN/TPQ-31 Velocity Response Characteristic Curve

Annex B - (Background Information) to Report of AN/TPQ-31 Performance during Evaluation as a Hostile Weapon Locator BACKGROUND INFORMATION

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In May 1967 the United States Marine Corps, to meet an urgent Vietnam requirement for a 360° counter mortar radar capability, awarded Raytheon a contract to test the basic capability of the AN/UPS-1 air search radar and evaluate its potential in a two dimensional mortar detection role. The test and evaluation program was undertaken at the Marine Corps Development Center, Quantico, Virginia and the basic capabilities and deficiencies of the AN/UPS-1 radar were established. The result of these tests showed a definite capability of the radar to provide the basic functions required to fulfill the new mission. In October 1967 Raytheon received a quick reaction contract from the USMC for the development of 10 AN/TPQ-31 2D counter weapon radars, the first to be delivered for test in 9 months. In order to provide AN/TPQ-31 equipment in Vietnam in the minimum possible time, existing AN/UPS-1 air search radars, developed during the mid 1950's, were provided from available Marine Corps assets for modification. Further, because of the time restrictions, only those modification's essential to the basic function of the AN/TPQ-31 were undertaken. Various features which may have been incorporated in a complete "design-forthe-purpose" were of necessity not pursued. The major modifications made to the 12 year old design of the AN/UPS-1 in 1967, to develop the AN/TPQ-31, were limited to the following:

Transmitter: Increase PRF to avoid blind speeds associated with mortar rounds. Increase output frequency stability and decrease time jitter to improve SCV.

Receiver: Modify stalo and colo for improved SCV.

Indicator: Provide 16" PPI with selectable 24 KM and 12 KM ranges.

MTI: Replace MTI delay line with a range gated doppler processor to

provide synthetic video to the PPI indicator and further improve SCV.

In June of 1968, the AN/TPQ-31 system was tested at Camp Lejeune, North Carolina. Analysis and evaluation of the test results confirmed that the desired capabilities were successfully incorporated into the AN/TPQ-31. The tests performed at Camp Lejeune were similar to those conducted with the AN/UPS-1 at Quantico except they were more detailed in nature and extensive in scope. In addition to demonstrating its ability to acquire and track mortar shells, the tests were extended to include howitzers and guns with emphasis placed on the radar's ability to determine the source of hostile firing with sufficient accuracy to permit direct assignment of retaliatory action without the use of other precision tracking devices.

Results of the tests were highly gratifying because the demonstrated capability significantly exceeded the performance specifications, despite the fact that the design was based on the mortar threat only. Helicopter gun ships were directed by the radar operator in simulated counter strike operations. The location accuracy obtained was generally better than ± 250 meters. No significant terrain features were encountered at Camp Lejeune, the firing range was generally flat, therefore possible mask angles were very small. The firing points and impact area were visible to the radar and the clutter from trees presented no problem to the radar.

In September 1968 the first units were deployed in Camp Pencleton,

California and active training of operator and maintenance crews commenced.

The terrain at Camp Pendleton is generally mountainous and significant mask

angles were encountered. However, although this presented problems in predicting the actual position of the weapon firing from defilade positions, the results in these adverse circumstances were considered of real value. Various methods of extrapolating the observed projectile track back to the point of origin were developed but none were implemented as a formal procedure. In February of 1969 deployment of active units commenced in Vietnam. The initial locations were: (1) Southwest of DaNang at Hill 55, (2) Northwest of Dong Ha at Fire Base C2, 7 KM from the DMZ and (3) in Saigon. The Saigon operation was not successful because the radar had to compete with excessive clutter returns from local buildings, etc. and the targets were not always detected. The site was on top of one of the tallest buildings in the city - while this would appear to be a vantage point - it was in fact a serious disadvantage because the high clutter returns could not be masked from the radar. Brief deployment outside the city at ground level produced more satisfactory results and the radar was later moved to LZ Sharon East of Dong Ha. Of the various systems ultimately deployed in combat locations, the best results were obtained from Hill 55, Da Nang and Fire Base C2 Dong Ha, partly because the enemy was particularly active in these areas. Careful siting of the radar system and constant vigilance on the part of the radar operator were found to be essential to the effectiveness of the system.

Annex C - (AN/TPQ-31 () Performance Objectives) to Report of AN/TPQ-31 Performance as a Hostile Weapon Locator

Raytheon, under Contract No. M00027-72-C-0098 with the U.S. Marine Corps, is conducting a modification program to further enhance the existing performance and versatility of the current models of the AN/TPQ-31 Radar System. Under this program the following desired design objectives are being pursued.

AN/TPQ-31 () PERFORMANCE OBJECTIVES

		Description	Objectives
A.	Envir	onmental	Allow 20% degradation in 4 mm/m of
			rain. Operate at sea level to 3000 m.
			alt.
3.	Relia	bility	90% Prob. of operating 24 hr/day for
			30 days. Down time 1 hr/day.
c.	Prob	Detection and Location	80%
	(0.00	14 sq. meters at 25 Km)	
D.	Accus	racy:	
	1.	Cannon Location	100M c.e.p. for 10 Km weapon at QE
			200 Mils. to 150M c.e.p. for 20 Km
			weapon QE 300 Mils.
	2.	Rocket Location	100M c.e.p. or c.e.p. = 1% range
		(122 mm or greater)	(whichever is greater) for QE -
			400 Mils.
	3.	Mortar Location	80M c.e.p. for 10 Km weapon at QE
		(60 mm or greater)	1000 Mils 90% within 160M at 12 Km
			range.

Objectives Description E. One Round Location Yes F. Radar-Weapon Operating Range Howitzer, 75 mm and 105 mm 1-25 km 2. Howitzer, 155 mm 1-25 km Gun, 175 mm 1-25 km 3. 1-25 km 4. Rockets (122 mm) Rockets (200 mm) 1-25 km 5. 0.5-25 km (QE = 1000 Mils)6. Mortar (60 mm) G. Enemy Weapon Characteristics and Conditions Max. Projectile Velocity 1000 m/sec from 5 km to max. ranges and 500 m/sec from 0.5 to 5 km 2. Minimum Weapon QE for 300 Mils 'rom 10 km to max. ranges Cannon and Rocket in (F.) and 200 Mils from 2 to 10 km. 3. Weapon QE for Mortar 500 Mils to 1500 Mils. Terrain Masking of weapon -50 Mils to +200 Mils. at radar 5. Radar-Weapon Aspect Angles 0 Mils to 1600 Mils. 6. Radar-Weapon Height $0 \text{ to } \pm 1000 \text{ Meters}$ Differential

H. Sectors of Search Track

I. Adjustment of Artillery

- 1. Simultaneous Search and 360° to 25 km range
- Locate
- 1. Fire-for-Effect Missions Yes

Description

Objectives

	2,	Fir	ing Corrections by	Yes
		Reg	gistration	
J.	React	ion (i.e., set-up time from	
	start	of O	ff-load to Operational)	
	1.	Sys	tem not emplaced	30 mins.
	2.	Sys	tem moved by air	l hour
	3.	Dis	placing	15 mins.
K.	Mobi	lity,	tracked or wheeled	Yes
	vehic	les.	No degradation of	
	vehic	le fo	ording or swimming	
	capal	oility	•	
L.	Pres	entat	ion	
	1.	Aut	omatic Print-out of hard	Yes
		cop	y:	
		a.	Weapon Loc. in UTM	Yes
			cords	
۸,		b.	Weapon Traj. back	Yes
			azimuth	
		c.	Time of Location	Yes
		d.	Identity of radar	Yes
		e.	Altitude of weapon locat.	Yes
		f.	Indicate whether weapon	Yes
			is cannon or rocket	

launcher

2. P.P.I. Az. vs Range

Yes

		Description	Objectives
M.	Orie	entation	
	Rapi	id visual or electronic means of	Yei
	orie	nting the radars will be provided.	
N.	Rate	e of computing weapon locations	4 per min.
0.	Air	portability via C-130 and Heli-	Yes
	copt	er Air Lift	
P.	Surf	ace Transportability by Truck or	Yeş
	Trac	ck Vehicles	
Q.	Num	ber of simultaneously-firing	10
	wear	oons which can be located	
R.	Man	ual (back-up) Method of Weapon	Yes
	Loca	ation	
s.	Othe	r Capabilities	
	1.	Ability to choose between weapons.	No
		That is, select a new, previously	
		unlocated weapon in each of 3 bat-	
		teries rather than 3 weapons in one	
		battery.	
	2.	Provide historical information on	No
		activity of located weapon.	
T.	Stora	age under extreme conditions without	l year
	subs	equent perf. degradation.	
	1.	Time to restore radar to full	8 hours
		capability.	
υ.	Total	l life expectancy	10 years

PHYSICAL CHARACTERISTICS

\ <u></u>		Description	Objectives
A.	Desig	n Considerations	
	1.	No. of vehicles required	2
	2.	Emplace and operate on 10° slope	Yes
	3.	Adequate shelter for operator,	Yes
		comm. equip., plotting board, etc.	
	4.	Provision for remote operations	100 ft.
В.	Weigl	ht	
	1.	Curb weight	4,560 lbs.
	2.	Combat loaded	5,000 lbs.
C.	Confi	guration:	
	1.	Operational height	168"
	2.	Surface mobility	
		a. Railway height	92"
		b. Highway and cross country height	114"
		c. Width	106"
	3.	Dimensions to conform to Berne	Yes
		International clearance diagram	
	4.	Air Portability:	
		a, Height	102"
		b. Width	106"
D.	Other	Requirements	
	1.	Withstand exposure to normal ocean	3 months
		beach atmosphere	
	2.	Fungus-Proof in operating and trans-	Yes
		port configuration C-5	

Appendix 1 - (Comparison of AN/TPQ-31 () to AN/TPQ-31) to Annex C

of Report of AN/TPQ-31 Performance during Evaluation as
a Hostile Weapon Locator

AN/TPQ-31 () COMPARISON TO AN/TPQ-31

Coverage

The increased sensitivity of the AN/TPQ-31 () due to higher average transmitter power, more sensitive received and better processing leads to the coverage pattern for a 0.005 sq meter target shown in figure 1. The AN/TPQ-31 () improvement is conservatively +7 dB and would have allowed at least some detections for tests 14, 15/1, 15/2 and 16/1 at Ft. Sill which were beyond the range of AN/TPQ-31.

Velocity Notch

The AN/TPQ-31 () is designed for a PRF of 6000 Hz which places the first blind velocity at 4 times the value for the AN/TPQ-31. Thus the numerous cases, such as tests 7/2, 7/4, 9/3, etc. at Ft. Sill, where the radar was blind to the shell at the start of the trajectories, would have been properly tracked. Successful track, from the point the salve the radar horizon, is necessary for minimum miss distance. None of the many cases of velocity notch blindness (see detection curves, Appendix 3, Annex F) for the Ft. Sill tests would have presented a blind condition for the AN/TPQ-31 ().

Detectability

Based on the relative coverage and velocity notch improvement of the AN/TPQ-31 () over the AN/TPQ-31, virtually all of the Ft. Sill firings would have been detected by the AN/TPQ-31 ().

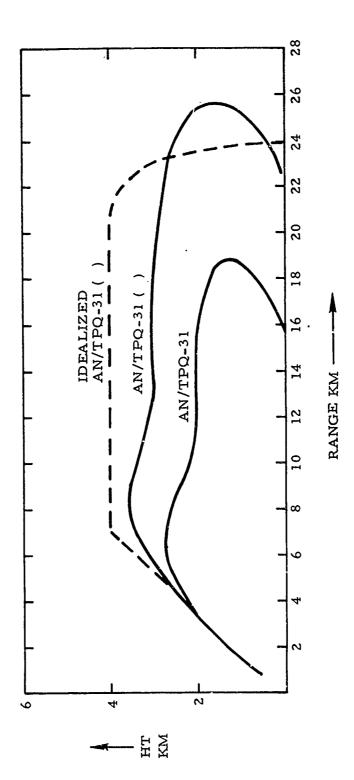


Figure 1. Radar Coverage for σ_{T} = 0.005 sq meter (No STC)

Annex D - (Wang 700A Calculator Program) to Report of AN/TPQ-31 Performance during Evaluation as a Hostile Weapon Locator

WANG 700A CALCULATOR PROGRAM

STORAGE REGISTERS USED

- 03 Height Difference (WEAPON RADAR) in meters
- 04 Actual Weapon Azimuth in mils
- 05 Actual Weapon Range converted to meters
- 06 Mask angle in degrees
- 07 1st Det. Az converted to degrees
- 08 1st Det. Rg converted to meters
- 09 Last Det. Az converted to degrees
- 10 Last Det. Rg converted to meters
- 11 Number of hits minus one
- *12 Number of rounds calculated in sequence
- *13 \(\Sigma\) calculated Az in mils, Avg Az in mils
- *14 \(\Sigma\) calculated Rg in meters, Avg Rg in meters
 - 15 X₂ X₁, Avg Az error in mils
 - 16 Y₂ Y₁, Avg Rg error in meters
 - 17 Cartesian coord. $X_L(X_2, X_1, X_0, X_L)$
 - 18 Cartesian coord. Y_L(Y₂, Y₁, Y₀, Y_L)
 - 19 T²

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- 20 Calculated Az in mils = A_{F}
- 21 Mils to Degrees conversion
- 22 Kilometers to meters conversion

^{*}Accumulating Registers.

- 23 Calculated Rg in meters = R_F
- *24 Σ/Az error/, Average/Az error/
- *25 \(\Sigma/\text{Rg error}\), Average/Rg error/
- 26 Working Register
- *27 \(\Sigma \text{RMS Az error} \), Average RMS Az error
- *28 \(\Sigma \text{RMS Rg error} \), Average RMS Rg error
- *Accumulating Registers.

NOTE: Registers 00, 01, 02, 120 & 121 are used with the trig package.

FUNCTION KEYS USED

01 00	Clear storage register and store radar/weapon height in meters
01 01	Store actual azimuth and range of weapon in mils and KM
01 02	Store mask angle in degrees
01 03	Store 1st detection azimuth in mils and range in kilometers
01 04	Store last detection azimuth in mils and range in kilometers
∪∄ 05	Store number of scans
01 06	Calculate correction factor T ² (display shows T ² in Y and impact from
	C in the X register)
01 07	Calculate origin and errors
	Display shows Az in mils/Rg in meters
	Press GO and display shows:
	Cartesian coordinates Y/X in meters

Press GO and display shows:

Error from actual Az in mils/Rg in meters

Press GO and display shows:

Miss distance in meters

01 08	Calculate averages - display shows average the same as 01 07
01 09	Calculate absolute averages - dimplay shows absolute error Az in
	mils/Rg in meters
01 10	Store each individual Az and Rg error and compute difference from
	average
01 11	Calculate RMS average - display shows Az RMS error in mils/Rg RMS
	error in meters
	Press GO and display shows miss distance in meters

PROGRAM MARKS USED

0002	Sin X
0003	Cos X
0004	Tan X
0007	Tan-1X
0100 0111	As described for function keys
0308	Quadrature II correction
0309	Quadrature II and III correction
0310	lst half of miss distance problem
0311	2nd half of miss distance problem
0312	Compute Az & Rg error
0313	Compute Mask correction
0314	Quadrature II and IV correction
0315	Clear registers 03 to 29
0801	Add 2 to value of T
0802	Subtract 2 from value of T

1513 1514 Used in trig program 1515

TOTAL NUMBER OF STEPS

699

VERIFICATION NUMBER

7097

COMPUTATIONS

R₁ = First range in meters

A₁ = First azimuth in degrees

R₂ = 2nd range in meters

A₂ = 2nd azimuth in degrees

N = Number of points from 1st to last

 ϕ_{m} = Mask angle in degrees

H_R = Radar height in meters

Hw = Weapon height in meters

 $H_D = Difference height in meters (H_W - H_R)$

Polar to Cartesian Coordinates

$$X_1 = R_1 \sin A_1$$

$$Y_1 = R_1 \cos A_1$$

$$X_2 = R_2 Sin A_2$$

$$Y_2 = R_2 \cos A_2$$

Origin if mask = 0°

$$X_0 = X_1 - 1/2 \left(\frac{X_2 - X_1}{N - 1} \right), Y_0 = Y_1 - 1/2 \left(\frac{Y_2 - Y_1}{N - 1} \right)$$

Fartial Time Correction (C)

$$C = \frac{A_2 - A_1}{180} \text{ if } C \begin{cases} <-1, T = 2 (N-1) + C + 2 \\ -1 \le C < 1, T = 2 (N-1) + C \\ \ge +1, T = 2 (N-1) + C - 2 \end{cases}$$

Mask Correction

B = 0.21 (R₁ tan
$$\phi_{\rm m}$$
 - H_D)/T²
S_X = (X₂ - X₁)B, S_Y = (Y₂ - Y₁)B

Computed Origin

$$X_L = X_0 - S_X$$
 $Y_L = Y_0 - S_Y$

$$A_F = \tan^{-1} \frac{X_L}{Y_L}$$
 $R_F = \frac{X_L}{\sin A_F}$

AVERAGES

Accumulator adds Az in mils and range in meters.

Cartesian coordinates are computed from average polar coordinates.

Error subtracts actual from average polar coordinates.

Miss distance is computed from average error and average range.

Absolute error is accumulated from each individual error.

RMS error subtracts the average error from each individual error, squares it, accumulates sum of individual squares, takes average and the square root.

RMS miss distance is computed from RMS Az and Rg error and the average calculated range.

DATA SHEET

A sample data sheet is attached.

PROGRAM STEPS

Data sheets with the individual program steps are attached.

OPERATING PROCEDURE

- 1. Load and verify program.
- 2. Place toggle switch 10 in On position.
- 3. Key Radar Height in meters and raise to Y.
- 4. Key Weapon Height in meters.
- 5. Press function key 00 to enter height difference and clear all other storage registers.
- 6. Key actual weapon azimuth in mils and raise to Y.
- 7. Key actual weapon range in kilometers.
- 8. Press function key 01 to enter position.
- 9. Key mask angle in degrees.
- 10. Press function key 02 to enter data.
- 11. Key 1st detection Az & Rg as in steps 6 & 7.
- 12. Press function key 03.
- 13. Key last detection Az & Rg as in steps 6 & 7.
- 14. Press function key 04.
- 15. Key number of hits.
- 16. Press function key 05.
- 17. Press function key 06 to calculate factor T².
 (NOTE: Display shows T² in Y and in X either 0, +2, or -2 depending on the value of C.)
- 18. Press function key 07 to calculate origin.
- 19. Record azimuth in mils from Y and range in meters from X.
- 20. Press Go and record cartesian coordinates in meters.
- 21. Press Go and record azimuth error in mils from Y and range error in meters from X.

- 22. Press Go and record miss distance in meters from Y.
- 23. For each round in the sequence repeat steps 11 to 22.
- 24. When the last round is complete press function key 08.
- 25. Record average values for each column as in steps 19 to 22.
- 26. Press function key 09 and record the absolute error for azimuth and range.
- 27. Key the azimuth error from the 1st round and raise to Y.
- 28. Key the range error from the 1st round.

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- 29. Press function key 10 to store RMS in an accumulator.
 (NOTE: The display shows the square of the difference between the error and the average error.)
- 30. Repeat steps 27 to 29 for each round in the sequence.
- 31. When the last round is complete, press function key 11 to compute the RMS values.
- 32. Record the RMS azimuth error in mils and the RMS range error in meters.
- 33. Press Go and record the RMS miss distance in meters.
- 34. When starting the next sequence be sure to start again at step 3 in order to clear the accumulators.
 - NOTE: When storing information from individual rounds in steps 11 to 16, it is not necessary to re-enter information if it was the same as the previous round. All other steps should be performed.

Miss Dist. Meters 0 Error from Actual Rg Meters Km, Mask Angle Az Mils X Meters Avg. Absolute Error mils, Rg RMS Error Y Meters SEQUENCE # Calculated Origin Rg Meters Az ź. Az. Mils Actual Weapon Location: Ht. TEST # No. of Hits Average R R Last Detection Az. Xi1\$ 'n. AN/TPQ-31 EVALUATION Radar Location 88 E 1st Detection Radar Ht. Az. Miis

Sup	Key	Code	Comment
0	MARK	0108	Sin(x) Start
1	0002	0002	
2	1	0604	
3	9	0709	Requires Cos(x)
4	0	0700	
5		0601	
6	1	0605	
í	MARK	0408	Cos(x) - Start
8	0003	0003	
9	1	0604	
01 0	3	0703	
1	6	0706	
2	0	0700	
3	÷	0603	
4	1	0605	
5	INT X	0608	
6	-	0601	
7	4	0704	
8	x	0602	
9	1	0605	
02 0	INT X	0608	
1		0601	
2	WRITE A	0412	
3	√x	0612	COSINF TEST
4	π	0609	
5	<u>x</u>	0602	
6	2	0702	
7		0603	
8	<u> </u>	0605	
9	x2	0713	
0.3 0		0404	ļ
'	RFG 120	1200	\sqrt{2}
7	<u> </u>	0701	ļ
3	1.	0706	
1	ļ <u>. </u>	0604	
5		0701	
6		0401	
7	1	0000 -	Ans
8	T	0403 -	
9	1514	1514	<u> </u>

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Step	Key	Code	Comment
01 0	RI-DIR	0405	
1	RIG 120	1200	
2	\ DIR	0402	
3	RFG 00	0000	
4		0605	
	* DIR	0403	
6	RFG 00	0000	
7	1	0701	
. 8	• • • • • • • • • • • • • • • • • • • •	0601	*** * * * * * * * * * * * * * * * * * *
9	1	0605	
05 0	CHSIGN	0711	
1	÷ DIR	0403	
2	REG 00	0000	
3	l	0701	
4		0601	
5	+ DIR	0400	
6	RIG 00	0000	
7	WRITE A	0412	
8	WRITE	0411	Y = 0
9	SEARCH	0407	
06 0	1514	1514	
1	RE Y	0415	
2	RFG 00	0000	
3		0712	
4	5	<u>0</u> 70 <u>5</u>	
5	SET EXT	0710	
6	CHSIGN	<u> </u>	
	<u> </u>	07 <u>01</u>	
8	1	0701	
9		0601	<u> </u>
07 0		[0e01	
<u> _'</u>	ļ	0605	
	WRITEA	0112	
3	I SPPROG	0512	Set Sign
4	SPARCH	0407	
5	1515	1515	Sm(x), Cos(x)-1:nd
6	MARK	0403	Tan 1(x)-Start
7	0007	0007	
8		0412	
9	CHARX	0715	Arc Lan Fest 90°

			·
Step	Key	Cotic	Comment
08 0	[.'	0001	
1		0712	
2	5	0705	
3	SKIP IF Y> X	0507	
4	WRITE A	0412	
5	χ,	0713	Arc Tan Fest 45"
6	1	0701	
7		0600	
8	STY	0114	
9	RFG 00	0000	
0 0	2	0702	
1		0601	
2	REDIR	0405	
3	RFG 00	0000	
4	÷	0603	
5	1	0605	
6	ST Y	0114	input/ans.
7	RIGOL	0001	
8	\	0602	
9	SFY	ल्यान	angle x2
10 0	RFG 00	0000	
1	1	0701	
2	ST DIR	0404	partial prod.
3	REG 120	1200	
4	1	0701	
5	5	0705	
G	1	0604	
	8	0708	
8	ST DIR	0404	inner prod.
9	REGOT	000	
11_0	MARK	0108	
1	1513	1513	
2	RE DIR	0405	
3	RI G 00	0000	
4	Z DIR	0402	
5	RIG 120	1200	
6	REDIR	0405	
	RI G 02	0002	
0	X DIB	0402	
9	REG 02	0062	

		,	T
Step	Key	Code	Comment
15 0	} ~~~~	0406	
	RFG 02	0002	ļ
2	X DIR	0402	
3	RIG 120	1200	
4	†	0605	
5	+ DIR	0100	
6	RFG 120	1200	
7	2	0702	
8		0601	
9	1	0701	
13 0	DIR	0401	
1	R1'G 02	0002	
2	EX DIR	0406	
3	RFG 120	1200	
4	÷ DIR	0403	
5	RI G 120	1200	
6	RF DIR	0405	
7	RFG 02	0002	<u></u>
8		0412	
9	100,3	0611	Skip X ≈ 0
14 0	SLARÇII	0407	
1	1513	1513	
2	REY	0415	
3	REG OI	0001	
4	REDIR	0405	
5	RIG 10	1,200	
6	X	0605	
7	WRITE A	0412	
"	GO	0514	180/π
9	`	0602	
15_0	4	0701	
1	5	0705	
2	KSHTV.	0.11	***************************************
3	10,	06[3	Are Tan Set
1	1	0605	
5	WRIII A	0413	
6	I ND PROG	0512	Set Sign
/	SEARCH	0407	
- 8	1515	1515	l'an ⁻¹ (x)-End
9	MARK	0408	Tan(x) -Start

Step			<u> </u>
	Key	Code	Comment
160	0004		
1	ST DIR		
2	01		
3	SR0002		
4	2 DIR		
5	01		
6	5 R 0003		
. 7	+ DIR	<u></u>	
8	01		
9	REC DIR		
170	01		
1	RETURN		
2	MARK		
3	1515		
4	REC Y		
5	121		
6	RETURN		END OFTIRE PK
7	MARK		CLEAR AND
8			CLEAR AND RADAR/WEAD HT.
9	17		
180			
1	STY		
2	03		
3	0		
4	4		
5	1		
6	SR 0315		
7	0		
8	1		
9	STOP		
190	MARK		WEAPONACTOAL
1	0101		WEAPONACTUAL AZ/RG
2	ST DIR		
3	05		
4	,		
5	0		
6			
7	6		
8	5 0 2		
	5		

			
Step	Key	Code	Comment
200	60		
1	57 Y		
2	04	<u> </u>	
3	STDIR		
4	21		
5			
6	0		·
7	0		
8	0	<u> </u>	
g	X DIR		
2/0	05		
1	ST DIA		
2	22		
3	0		
4	1		
5	STOP		
6	MARK		
7	0102		MAGKX
8	ST DIR		
9	04		
220	0		
1	STOP		
2	MARK		IST DET.
3	0103		Az/Ra.
4	STDIR		
5	08		
6	REC DIA		
7	21		
8	X		
9	STY		
230	07		
1	REC DIA		-
2	27		
· з	22 X DIR		
4	08		
5	0		
Ģ	1		
7	STOP		
8	MARK		LAST DET.
9	0104		LAST DET. AZ/RA

D-11

· · · · · · · · · · · · · · · · · · ·			
Step	Key	Code	Comment
240	ST DIA		
1	10		
2	REC DA		
3	21		
4	X		
5	ST Y		
6	09 REC DIA		
7	REC DIA		
8	22		
9	X DIR		
250	10		
1	0		
2	^		
3	STOP		
4	MARK		
5	0105		No OF HITS
6	1		
7			
8			
9	57 /		
260	11		
1	0		
2	1		
3	STOP		
4	MARK		COMPUTE T2
5	0106		
6 7	REC Y		
7	1)		
8	1] 2		
9	✓		
270	ST / 19 19 19 09 REC DIA 07		
1	19		
2	PEC. Y		
3	09		
4	REC DIA		
5	07	,	
6	-		
7	1		
8	. &		
9	0		

Step	Key	Lode	Comment
280	÷		
1	V		
2	+ Din		
3	19		
4	,		
5	CHE SIGN		
6	Saurif YZX		
7	SEARCH		
8	0801		
ຄ	CHG SION		
	SKIP IF YCX		
1	SEARCH		
2	0802		
3	REL DIR		
4	19		
5	XZ		
6	ST DIR		
7	19		
8	1		
9	0	L	
300	STIP		
1	MARK		
2	0801		A00 2 TO T
3	REC Y		
4	19		
5	2		
6			
7	72		
8	XZ		
9	ST DIR		
3/0	19		
1	77		
2	STOP		
3	MARK		
4	0802		SUB 2 FROM T
5	REC Y		
6	19		
7	2		
8	CHG SION		
9.	·	Ī	I :

Same !	Key	Code	Comment
Step		Code	, community
320 1	1 J X-2		·
2			
3	ST DIR		
4	19		
- 5			
6	STOP		
7	MARK		COMPUTE ORION
8	0107		CIAPUTE UNITA
9	REC DIA		A ₂
	09		772
	SR 0002		
1	1.		
2	REC DIA		
3	10		Rz
4	X		
5	9T Y		
6	17		Xz
7	REC DIA		
8	07		Α,
9	SR OCO2		
340			V40-1-W-1-0-1-0-1-0-1-0-1-0-1-0-1-0-1-0-1-
1	REC DIA		
2	08		R ₁
3	<u> </u>		
4	<u> </u>		
5	R DIM		
6			Х
7	17		ļ
8	- 1	<u> </u>	
9	ST Y		
350	15		×2 - ×1
1	RFC Din		ļ.,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
2	11.		(N-1)
3	-	<u> </u>	
1	ス ユ エ		
5			
6	Ψ.		
7	-DIR		ļ
8	17		×o
9	REC DIA		

D-

Step	Key	Code	Comment
340	09		AL
1	5/10003		
2	1		
3	RFC DIR		
4	10		RL
5	10 X St Y		
6	57 Y		
7	18		72
:	REC DIR		
9	07		A,
370			
1	1		
2	REC DIA		
3	08		R ₁
4	Χ		
5	V		
6	2 DIR		
7			3,
8	18 12		•
9			
380	·Sr y		
1	16		13: -31
. 2	REC DIA		· · · · · · · · · · · · · · · · · · ·
3)/		(N-1)
4	7.		
5	2		
6			
7	v		
8			
ġ	18		Yo
390			, , , , , , , , , , , , , , , , , , ,
1	06		MASKA
2	S 1 0004		ton 4
3	1		
4	0		
5	SHIPIFY-X	0509	
6	Go		
7	SR 0313		CURRECTONFACTOR
8			L. P. M C. VVI MEIDKI
9	1		Y ₄
<u> </u>	' D	L	/ <i></i>

/ ********			
Step	Key	Code	Comment
0			
1			
2	ADD ITIO	NAL	STEPS
3	REDVIR	F7 T	O CORNECT
4	FOR T	4N-1	OF ANGLES
5	IN THE	= 2	40 4 3 KD
6	QUAD		
7			
8			
9			
400	CLEAR X		
1	ST DIR		
2	20		
3			7 941715
4) 9KIPIF } Yin +
5	GO		
6			OUAN II9II
7			
8			XL
9	17		
410			
1	Ψ		
2	50 0007		700 -1
3	1		
4	0		
5	SHIP IF YZX		
6	60		
7	SR 0314		QUADIT STE
8	V		
9	+ DIR		
420	20		
1	PEC DIA		
2	20		Ax
3			
4			
5			
6			
7			
ß			
9			

Step	Key	(.ode	Comment
0			
1			
2			
3			
4			
5		L	
6			
7	ļ		
:3			
9			
0			
1			
2			
3			
4			
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6			
7			
8			
9			
0			<u> </u>
1			
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3			
4			
5			,
6			
7			
8			
9			
0	,		
1			
2			
. 3			
4			
5			
6			
7			
8			
9			

D-14

Step	Key	Code	Comment	Î	Step	Key	(.ode	Comment
r					45 0	STOP		
1					1	RECY		
2					2	20		AF (MILE)
3					3	Rec Din		
4					4	23		RF (HEORA)
5					5	SR 03/2		COMPOTE FARAR
6					6	STOP		
. 7	·×				7	1×1		
8					8	+ DIR		
Ç.					9	25		EIRGEORES
Ú					460	12		
1					1	. x		
2	, , , , , , , , , , , , , , , , , , , ,				2	+ DIR		
423	SR 0002		si 4		3	24		E /AZ FREEN
4	Rec Y	,			4	7.)		
5	17	~	X۲		5	SR 0310		CONPOTE MISS
6	÷					SR 0004		ton 4
7	5-1				7	REC Y		
8	23		R _F		8	23		R=
9	REC DIR				9	500311		COMPUTE MISS
430	21		MILG -> DEG		470	. 0		
1	1/x		DEG -> MILS		1	STOP		* *
2	X DIR				• 2	GO		
3	20		AF (MILS)		3	GO		
4	1				4	GO		
5	+ DIR				5	MARK		COMPUTE
6	12		COUNTER		6			AUFPAGES
7	V				7			
8	+ DIR				8			No.
9	14		RE Accum.	j	9	+ DIR		
4401	REC DIA				480			Ava Ra
1	20		AF (ris)		1	+ DIR		•
2	+ DIR				2	13		AUG Az
3	13		Ar Accum		. 3	REC Y		
4	12		, , , , , , , , , , , , , , , , , , ,		4	13		
5	STOP				5	REC DIR		
6	REC Y				6			
7	18		YL		7	STOP		`
8	REC Din				8	REC DIR		
9	17		ΧL		9	21		MILG =7 DEG.
·				D-1				

Char	K	Code	Comment		Step	Key	Code	Comment
Step	Key	Code	Comment		530	+ DIR		Comment
490	×				2)0	25		Ava Ra man /
2					2			700 11000
3	ST DIR				3	24		
4	26				4	REC DIR	<u> </u>	
5	5 R 0002		sin &		5	25		
6	1				6			
7	REC DIR		0 0	-	7	STOP		EN+FR
	14		AVL Ra			MARK	<u></u>	
8	X				8	0110		ERRIR DATA
9	V				9			
500	2 DIR				<u>540</u>	26	<u> </u>	
1	24					REC DIA	 	
2	5R0004		lan 4	,	2	15	<u> </u>	AVE AZ FORM
3	+				3			
4	REC DIR				4	<u> </u>	<u> </u>	
5	26		\		5	X2		
6	STOP		AUG CARTESIAN CO	PRP.	6	+ DIR		
7	. RFC Y				7	27		ERMS Azwa
8	13		Aug Az		8	2 Din		
9	NEC DIR				9	24		
5(0	14		Ava Ra		<i>55</i> 0	··· •		verst 5th
1	SR 0312		COMPUTE ERROR		1	RECDIA		
2	STY				. 2	16		AVL RO FPRIA
3	15		Ava Az Ennon		3	****		
4	ST DIR				4	V		
5	14		Ave Ro. EANON		5	XZ		•
6	STOP				6	+ DIR		
7	SR 0310		COMPOSE MISS		7	28		E RMS RO FARIK
8			Han &		8	RECY		
9			,		Ž,	24		
520	,		Ava Ra		560	STOP		
1	SR 0311		COMPUTE MISS		1	MARK	•	CAMPUTE
2	0.				2	0111		RMS ERNOR
3	STOP	-			3	REC DIA		, A.
4	MARK		AVERAGE		4	12		No.
5			ABSULETE ERFOR		5	+ DIR		
6	NEC DIA		10 10 20 1 2 2 10 10 1		6	27		Ava RMy Azen
7	12		No.		7	+ DIR		· vu IIII FTE
8	+ Din		7		8	28		Ava RMg Ra en
9			Ava (Azera)		9	REC DIA		TIVE NATO NA JON
	· · · · · · · · · · · · · · · · · · ·		LIAN INEXAM	-16	ار ا	1.86 1/11		

	-		<u>'-</u>		1	Step	Key	Lode	Comment
	Step	Key	Code	Comment				Code	Connient
	5 70	27				610	REC DIR	-	
		77				2			A
	2	1				3	04		ALTONAL AZ
	3	REC DIA				4			
	5	28		,		5	<i>₩</i>		
	6	ママ	· 			6			
-	7			0		7	24		
	8			COMPLETE MISS		8			
	8 9			Tan 4		9			Actual Ra
		REC Y.		Ava RG			05		ACTUAL RE
	5 80	14				620	RECDIA		
	1	5R 0311		COMPUTE MISS		2			
	3	STOP		E	-	3	26 1 L	<u> </u>	
\dashv	4			FUNCTIONS COMPLE	•	4	RETURN		-
	5	MARK		COMPOTE 19T HAVE		5			44.40
		0310 X2		MISS DISTANCE		6			MAGK X
	6					7			CORRECTION
	7	ST DIA				8	08	<u> </u>	R,
	8 9	26				9			<u> </u>
	590						REC DIA		
	7 70	21		A		1	03		Н 5
	2	× '		MILS -> DEG		· 2			,15
	3	¥				3	REC DIR		
	4	RETURN				4	19		TZ
_	5	MARM		CUMPLIE 2ND HAVE		5	-		•
	6			HISG DISTANCE		6			
	7			11199 212 2116		7	æa		
	8	<i>\psi\</i>	 -			8	BI		
	9	ΧŽ				ġ	30		
	600					640			
	1	26				1	80		
	2	REC.DIA				2	40		
	3	26				. 3	X		
	4	VX			Ì	4	V		ය
	5	1				5	REC Y	-	
	6	RETURN				6	15		(× • - × ·)
	7	MARK		COMPOSE		7	×		
	8	0312		ERROR		8	12		
	9	ST DIA				9	- DIR		
•	اب سبسی			**************************************	D-17				

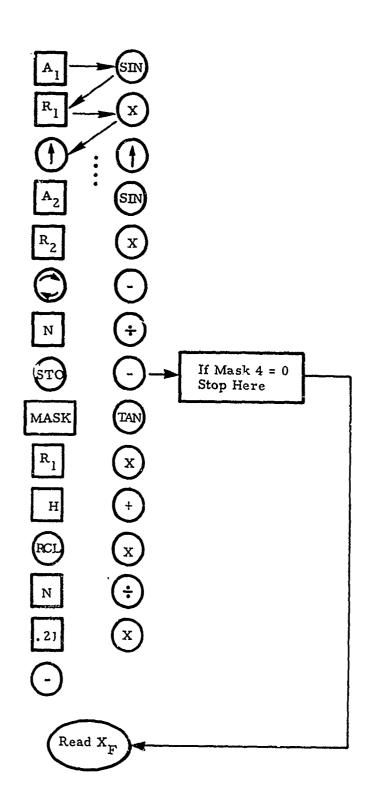
Step	Key	Code	Comment		Step	Кеу	Code	Comment
650			X۲		490	RETURN		
1	REC DIA		1		1	MARK		QUADRANT
2	16		(32-31)		2	0308		QUADRANT I
3	×				3	3		
4	V				4	' Le		
5					5	0		
6	18		. YL		6	- DIR		
7	RETURN				7	20		
8	MARK		QUADRANT		8	RETURN		
9	0314		QUADRANT II + TI.		9	END PROG		
660	3				0			
1	6				1			
2	0				2			
3	+				3			
4	RETURN				4			,
5	MARK		CLEAR		5			
6	0315		CLEAR REGISTERS		6			
7	0				7			
8	STIND				8	*		
9	1				9			
670					0			
1	t				1	-		
2	(. 2			
3	STIPIFY-X				3	-		
4	SEARCH				4			
5	0315				5			
6	RETURN				6	 -		
7	MARK		QUADRANT		7			
8	0309		正中亚		8			
9					9			
C80	8				0			
1					1			•
2	 				2			
3	20				3			
4	REC DIA				4			
5	17		ΧL		5			
6	WAITE A) X 19 -		6		<u></u>	
7	·····) X 19 -		7			
8	·····				8			
9	SR0308		QUAD II	n- 19	9			
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Appendix 1 - (HP 35 Program) to Annex D of Report of AN/TPQ-31

Performance during Evaluation as a Hostile Weapon Locator

A miniature Hewlett Packard calculator, HP-35, was also used to compute firing points and is believed to be a valuable time saving asset for such application. The program (based on certain approximations) to compute firing points on the HP-35 (N' is equal to 2(N-1)) is presented on the following page. The boxes represent data entry and the circles represent function buttons. The "Y" coordinate of the firing point, Y_F , is obtained using the same program, but by pressing COS whenever SIN is called for. The execution time for computing both X_F and Y_F was found to be 1-3/4 minutes total for someone modestly familiar with the calculator.



BACKTRACK COMPUTATION ON HP-35 HAND CALCULATOR

- Annex E (Dahlgren Test Results) to Report of AN/TPQ-31 Performance

 During Evaluation as a Hostile Weapon Locator
- 1. The raw data of the tests at Dahlgren is contained in Appendix 1 to this Annex. This data reflects target information collected on the following dates: December 1, 4, 6 and 7, 1972.
- a. Twenty-six rounds artillery were tracked on 1 December. The data is far from complete, but is included for record purposes. No back data could be obtained from the main range personnel as to weapon type, Q. E., etc.
- b. On 4 December, 18 rounds of 81 mm mortar were tracked and plotted. Although these rounds were not part of the scheduled program, weapon type and Q. E. information was made available. The predicted firing point location was estimated manually by the operator. The basic method of manual processing target information by the operator includes the plotting of mortar shells, artillery shells or missiles with a grease pencil on the face of the PPI crt. Starting at the point of origin, where the target was first detected, the operator follows the line of flight of the target by drawing a line through each of its target returns. Successive plotting of targets starting at their first point of detection helps the operator determine the point of origin of the hostile weapon emplacement. The range (distance) of the point of origin is determined by pointing the azimuth cursor to the first point of target detection and rotating the TARGET RANGE DESIGNATION control so the azimuth cursor touches the leading edge of the first target return. The azimuth of the target can then be determined by the azimuth scale on the outer edge of the PPI crt, and the range read off the digital range counter.

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c. The first half of scheduled mortar rounds to be observed from site 7 were fired from the main range firing point on 6 December and the second half on

7 December. The raw data for these dates shows the predicted origin of the firing point estimated manually by the operator. Tables F-1 through F-6 show this data when processed by a hand calculator (Wang).

2. For launch point calculation with the calculator, 6830 meters and 5655 mils were used as the best known range and bearing of the firing point from the radar. The following equations were evaluated on the hand calculator for each radar reading to yield a table of range launch point (R_0) , and azimuth launch points (Az_0) , *

$$R_0 = \frac{R_1 - R_2}{2(N-1)} 4 R_1$$

$$Az_0 = \frac{Az_1 - Az_2}{2(N-1)}$$

where for a particular round,

R₁ is the first detection range

Az₁ is the first detection Az

 R_2 is the last detection range

Az2 is the last detection Az

N is the number of scans between and including the first and last detections

These values are listed in the data column of the Tables E-1 through E-8. On line

2, Appendix E-1-1, and lines 2, 3, 4, 6 and 12, Appendix E-1-2, Annex E, are

first look range values which cannot be correct because they are less than the last

^{*}These results are approximate as the equations yield acceptable results only when the difference between the first and last azimuth is small, as was true in this case.

look value. This is assumed to be an operator error and 5 is assumed to be the correct first digit. Rounds 6, Tables E-1 and E-2 and 9, Tables E-7 and E-8, were not included in the computer runs because of no track.

GROUPING

There are 4 groups of firings:

Group 1 Ch 8 - 45°	Tables E-1 and E-2
Group 2 Ch 6 - 65°	Tables E-3 and E-4
Group 3 Ch 2 - 65°	Tables E-5 and E-6
Group 4 Ch 8 - 25°	Tables E-7 and E-8

All but group 4 were below the blind notch of the radar at PRR - 1600 Hz. Group 4 have a muzzle velocity of 248 m/s which becomes 225 m/s in the ground plane for QE = 25°. Since the system clutter notch extends from 154 to 198 m/s an occasional sighting may be expected just before the shell drops into the blind zone if the antenna rotation timing is just right. However, since the data of group 4 were beyond the design parameters of the AN/TPQ-31 they will be presented separately and not combined with the remaining data. Group 1 firings have an initial radial component of approximately 175 m/s which is still wi. in the 3 dB points in the notch, but are reasonably well detected because of the signal strength overcoming the value of the notch skirt at that point.

3. In Tables E-1 through E-8 a printout of the reduced data for groups 1, 2 and 3 are presented.

An overall bias for all the values (59 rounds) was computed to be:

Range bias 58.4 m

Az bias -2.42 mils

$$N_{\Sigma X_i}$$

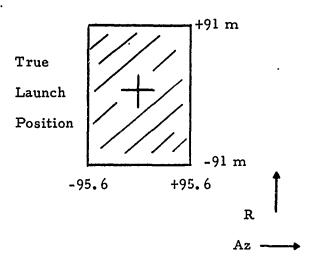
BIAS =
$$\frac{1}{N}$$
 - True Value

The average value for predicted launch range and azimuth is printed out for each group as well as the offset that value represents.

The error for each point is tabulated, as well as the error per point corrected by the 59 round bias.

The group variance for R_0 and Az_0 is printed below each group as well as the average value of the absolute error.

The prediction Footprint for the 59 Values of Groups 1, 2 and 3, with bias removed, within which 50% of all predictions fall, is presented as follows:



Area = 34,798 sq meters

Radius of circle of equivalent area:

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$$R_{CEP}\sqrt{\frac{34,798}{\pi}} = 105.2 \text{ meters}$$

4. The values in Tables E-7 and E-8 are presented separately with no further analysis because the projectile entered the radar blind zone shortly after firing. Errors of 592 m range bias and 37.6 mils Az bias resulted.

Table E-1

RANGE

RNH

6 Dec. 72 Charge 8 45°

Best known RO = 6830 meters, AZO = 5655 mils 59 round bias = 58.4 meters range and -2.42 mils AZ Avg value = 6904.8 meters Offset of avg of values = 74.8 meters

Rd.	Data R _o	Error	Error Less Bias
1	6720	-110	-168.373
2	7120	290	231.627
3	7410	580	521.627
4	6900	70	11.6271
5	6910	80	21.6271
6*			
7	6900	70	11.6271
8	6600	-230	-288.373
9	. 6900	70	11.6271
10	³ 7080	250	191.627
11	7100	270	211.627
12	6910	80	21.6271
13	7010	180	121.627
14	7010	180	121.627
15 🕟	6850	20	-38.3729
16	6790	-40	-98.3729
17	6800	-30	-88.3729
18	6590	-240	-298.373
19	6590	-240	-298.373
20	7010	180	121.627
21	6870	40	-18.3729
22	6900	70	11.6271
23	6990	160	101.627
24	6670	-160	-218.373
25	7010	180	121.627 Charge 6
26	6980	150	91.6271 65°

Variance = 182.912 meters rms Avg ABS error = 158.8 meters

^{*}Round 6 not included because of no track.

Table E-2
AZIMUTH

RNH

6 Dec. 72 Charge 8 45°

Best known RO = 6830 meters, AZO = 5655 mils 59 round bias = 58.4 meters range and -2.42 mils AZ Avg value = 5656.6 mils Offset of avg of values = 1.6 mils

Rd.	Data Az.	Error	Error Less Bias
1	5638	-17	-14.5763
2	5660	5	7.42373
3	5672	17	19.4237
4	5682	27	29.4237
1 2 3 4 5	5661	6	8,42373
6 *			
7	5640	-15	-12.5763
8	5650	~ 5	-2.57627
8 9	5670	15	17.4237
10	5643	-12	-9.57627
11	5639	-16	-13.5763
12	5638	-17	-14.5763
13	5648	-7	-4.57627
14	5647	-8	-5.57627
15	5689	34	36.4237
16	5649	-6	-3.57627
17	5658	3	5.42373
18	5654	-1	1.42373
19	5658	3	5.42373
20	5668	13	15.4237
21	5694	39	41.4237
22	5644	-11	-8.57627
23	5639	-16	-13.5763
24	5660	5	7.42373
25	5651	-4	-1.57627 Charge 6
26	5663	8	10.4237 65°

Variance = 15.4454 mils rms Avg ABS error = 12.4 mils

^{*}Round 6 not included because of no track.

Table E-3
RANGE

7 Dec. 72 Charge 6 65°

Best known RO = 6830 meters, AZO = 5655 mils 59 round bias = 58.4 meters range and -2.42 mils AZ Avg value = 6864.58 meters Offset of avg of values = 34.5833 meters

Rd.	Data R _o	Error	Error Less Bias
1	7050	220	161.627
2	6640	-190	-248.373
3	7180	350	291.627
4	6680	-150	-208.373
5	6870	40	-18.3729
6	7070	240	181,627
7	7070	240	181.627
1 2 3 4 5 6 7 8	6790	-40	-98.3729
9	6570	-260	-318.373
10	7070	240	181.627
11	6680	-150	-208, 373
12	6980	150	91.6271
13	6460	-370	-428.373
14	6870	40	-18.3729
15	6750	-80	-138.373
16	6970	140	81.6271
17	6480	- 350	-408.373
18	6960	130	71.6271
19	6970	140	81.6271
20	6970	140	81.6271
21	6970	140	81.6271
22	6870	40	-18.3729
23	6970	140	81.6271
24	6860	30	-28.3729

Variance = 190.853 meters rms Avg ABS error = 167.083 meters

Table E-4
AZIMUTH

7 Dec. 72 Charge 6 65°

Best known RO = 6830 meters, AZO = 5655 mils 59 round bias = 58.4 meters range and -2.42 mils AZ Avg value = 5654.08 mils Offset of avg of values = 0.916667 mils

Rd.	Data Az _o	Error	Error Less Bias
1	5614	-41	-38.5763
2	5643	-12	-9. 57627
1 2 3	5613	-42	-39.5763
4 5	5686	31	33.4237
5	5642	-13	-10.5763
6	5647	-8	-5.57627
7	5635	-20	-17.5763
8 9	5676	21	23,4237
9	5676	21	23.4237
10	5646	-9	-6.57627
11	5667	12	14.4237
12	5625	-3n	~27.5763
13	5677	22	24.4237
14	5667	12	14.4237
15	5646	-9	-6.57627
16	5667	12	14.4237
17	5665	10	12.4237
18	5656	1	3.42373
19	5667	12	14.4237
20	5646	-9	-6.57627
21	5646	-9	-6.57627
22	5667	12	14.4237
23	5656	1	3.42373
24	5668	13	15.4237

Variance = 19.0873 mils rms Avg ABS error = 15.9167 mils

Table E-5

RANGE

Best known RO = 6830 meters, AZO = 5655 mils 59 round Bias = 58.4 meters range and -2.42 mils AZ Avg value = 6904.4 meters Offset of avg of values = 74.4 meters

7 Dec. 72 Charge 2 65°

Rd.	Data R _o	Error	Error Less Bias
1	6943	113	54, 6271
2	6938	108	49.6271
3	6842	12	-46.3729
4	6950	120	61.6271
5	6943	113	54.6271
6	6955	125	66.6271
7	6950	120	61.6271
8	. 6844	14	-44.3729
9	6841	11	-47.3729
10	6838	8	-50.3729

Variance = 51.7672 meters rms Avg ABS error = 74.4 meters

Table E-6

AZIMUTH

Best known RO = 6830 meters, AZO = 5655 mils 59 round bias = 58.4 meters range and -2.42 mils AZ Avg value = 5638.9 mils Offset of avg of values = -16.1 mils

7 Dec. 72 Charge 2 65°

Rd.	Data AZ _o	Error	Error Less Bias
1	5635	-20	-17.5763
2	5635	-20	~17.5763
3	5645	-10	-7.57627
4	5647	-8	-5.57627
5	5636	-19	-16.5763
6	5637	-18	-15.5763
7	5645	-10	-7,57627
8	5636	-19	-16.5763
9	5637	-18	-15,5763
10	5636	-19	-16.5763

Variance = 4.50444 mils rms Avg ABS error = 16.1 mils

Table E-7
RANGE

7 Dec. 72 Charge 8 25°

Avg value = 6237.62 meters Bias of avg of values = -598.379 meters

Rd.	Data R _o	Error	Error Less Bias
1	5960	-870	-277.621
2	5608	-1222	-629.621
. 3	5410	-1421)	-827.621
` 4	5425	-1405	-812.621
5	5630	-1200	-607.621
6	5250	-1580	-987.621
1 2 3 4 5 6 7 8	7 050	220	812.379
8	6115	-715	-122.621
9*			
10	5881	-949	-356.621
11	5878	-952	-359.621
12	5894	-936	-343,621
13	6128	-702	-109.621
14	5410	-1420	-827.621
15	5825	-1005	-412.621
16	6703	-127	465.379
17	6496	-334	258.379
18	7025	205	797.379
19	6303	-527	65.3793
20	6507	-323	269.379
21	6910	80	672.379
22	6511	-319	273.379
23	7010	180	772.379
24	6104	-726	-133.621
25	6717	-113	479.379
26	6715	-115	477.379
27	6503	-327	265.379
28	6500	-330	262.379
29	6906	76	668.379
30	6507	-323	269.379

Variance = 536.068 meters rms Avg ABS error = 644,868 meters

^{*}Round 9 not included because of no track.

Table E-8
AZIMUTH

7 Dec. 72 Charge 8 QE 25°

RNH

Avg value = 5692.59 mils Bias of avg of values = 37.5862 mils

Rd.	Data AZ	Error	Error Less Bias
1	5ა89	34	-3.58621
1 2 3 4 5 6 7	5719	64	26.4138
3	5707	52	14.4138
4	5742	87	49.4138
5	5720	65	27.4138
6	5723	68	30.4138
7	5608	-47	-84.5862
8	571 4	59	21.4138
9*			
10	5711	56	18.4138
11	5683	28	-9.58621
12	5700	45	7.41379
13	5755	100	62.4138
14	5775	120	82.4138
15	5689	34	-3.58621
16	5651	-4	-41.5862
17	5655	0	-37.5863
18	5630	-25	-62.5862
19	5701	46	8.41379
20	5692	37	-0.586207
21	5704	49	11,4138
22	5670	15	-22, 5862
2 3	5662	7	-30.5862
24	5690	35	-2.58621
25	5680	25	-12.5862
26	5680	25	-12.5862
27	5691	36	-1.58621
28	56 9 2	37	-0.586207
29	5671	16	-21.5862
30	5681	26	-11.5862

Variance = 33.9721 mils rms Avg ABS error = 42.8276 mils

^{*}Round 9 not included because of no track.

Appendix 1 - (Recorded Test Data from Dahlgren Tests) to Annex E of Report
of AN/TPQ-31 Performance Evaluation as a Hostile Weapon Locator

TEST FIRING RESULTS

Radar	AN/TPQ-31	Location	Site 7	Date	Dec 1, 1972
Venar		meation.	3200 /	Date .	DCC 1, 1372

Round	Time		Last Det				_	_	_				Ţŝ	TE	eţ	T	acl	ķ_					,	_	19'2
	<u> </u>	R Az		R Az	1	2	! 3	3] 4		iļ 6	5	7] 8	8 9	1	0	11	12	!13	13	4	15	16	17	18	19'2
1		5670 6.6	5720 5.9		X	X	·																		
.2		5670 6.9																							
3		5860 4.6	1140 2.5	5425 6250	X	X	K	42	1														<u> </u>		
4		5830 6.4	1340 2.9	-5790 7200	Т	Γ	Г	Τ	Т	(X	()>	()	X)		X										
5 ;		5820 5.1:	1140 2.5		Τ	Τ	T	sec	ī																
6.		5710 6.6	1260 2.6	5990 7200	X	X	()	()		4>		<u> </u>	X	X	X										
. 7		5860 5.4	1250 ´ 2.6	5715 6050	Ī		1	1	1	İ		x ì	1								٠				
8		5840 5.4	1470 3.1	5785 5550	×	\ \	<u> </u>) ((C	()	(X		X)	(X											
9 .		5860 4.9	1550 3.1	5800 5700	X)	();	<u>ر</u> ر	ן גינ	()	K X	K												
10		5870 4.8	1220 2.7	5795 5600	X	2	1	((2 2			X													
11		5770 6.1	1170 2.5	5715 7200	X		d		x	XI)	K Z	X													
12		5740 6.1	1190 2.4	5685 7200	X				X :	X :	4	X /	X					1			_				
. 13 ~		5850 5.0	1220 2.7	5765 6050	\	(xh	<	0	V	X.	X													
14		5850 5.0	1210 2.6	5765 6026	\	6	X.	X	X	N.	X														
15		5680 6.7	1300 2.7	5640 7580 ,)		<		X			X	X	X								,			
16		5740 6.2	1320 2.6	5695 7000	1	X)	<	0	X	0	X)	X	X	X											
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TEST FIRING RESULTS

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Reder	AN/TPO-31	Location	Site 7	Date	Dec. 1, 1972

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Round	Time	1st Det	Last Det	Origin	Ļ		-	1.	1 6	1 -	. =		ar	zet	Tre	ıck		•			<u>. 1.</u>	_ (= 7	19'2
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1	1415	5730 6.4	6380 6.4		×	X		L		L	با	ur	st					L	<u> </u>					j
2	1420	5740 5.8	1500. 3.5		X	X	×	X	X	lx										İ	.	-		
3	1423	5780 5.5	1615 4.3		Ī		Г	i:	Ī	П								Γ	T	T		7		
4	.1423	5790 5.2	1510 3.8	•	V	,	C	Ţ			×									Ī		7		İ
5	1424	5730 6.5	1550 3.9			Γ	Γ	Ī	Π	Γ	Γ		_		i			Ĺ	<u> </u>	\dagger	†	7		İ
. 6	1425	5700	1640 8.6		Τ	Γ	Τ	Τ.		Ŧ	Γ	X	X			•		\vdash	 -	T		-	7	+
7	1426	5760 5.7	2000		Τ	Γ	Ī	Γ	Τ	Τ	Γ	Π		X		•	\vdash		1.	+	1		i	÷
8	1427	5730 6.1	2060 7.8		T	Π	Τ	Ī	T	Γ		X	Γ		_ <u>:</u>			<u> </u>		\dagger	-+ 	1		Ť
9	1428	5750 5.6	1960 8.4	·	Г	Γ	Γ	Į.	Γ	Τ	Γ	X			1				T	Ť	1		-	
10	1429	5760 5.9	1950 8.2			Ï	┱	Mi	T	T	Т		-						Ť	T	1	1		-
11	1430	5710 6.6	1960 8.0	•				Mi:	ss	ed														Ī
12	1431	5720 6.7	1910 7.8		×)	(1)	(<u>)</u>	(×	O	X	X	X										
. 13 ~	1432	5750 5.9	1970 8.4					MŁ	ss	ed					·		-						,	
14	1433	5740 6.3	1950 7.5		>	(h	(I)	()	()	X)	(0	X	X	X									•	
15	1434	5730 6.2	1910 7.6				Ī	Ni.	-	1														
16	1435	5710 6.5	1810 5.7		>	()	,	()	(')	XX	d >	1					Ī							
• 17	1436	5800 5.5	1910 7.8			Ì					T	1.	x											T
18	1437	5750 6.3	1950 8.1			1	ì	М	ı	- 1														
		5740 5.9	1940 7.4		_	N.	V	X	X	<u>x</u>	χļ	12	<u> </u>	X	<u> </u> x	ر ر	<u>()</u>	d_{λ}		K.	X			
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TEST FIRING RESULTS

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Annex F - (Ft. Sill Test Data Analysis) to Report of AN/TPQ-31 Performance

During Evaluation as a Hostile Weapon Locator

FT. SILL TEST DATA ANALYSIS

The prime analysis tool is the use of a computer simulation of each round fired. This allows reconstructing each shot to determine how the radar should have performed. The actual performance is then compared with the predicted performance. The statistics which were processed from the radar data provide a revealing insight to the performance of the radar for the Fort Sill trials.

Each round was simulated and the projectile aspect and pitch angle, range, range rate, azimuth, elevation and normalized target size for each 2 second interval of the shot were derived. A value for each 2 second interval representing the predicted signal level above the MDS threshold was then determined using this information. This signal level was plotted vs time and provided a basis for a comparison of radar results with expectations. See Appendix 3, this Annex, for predicted detection curves.

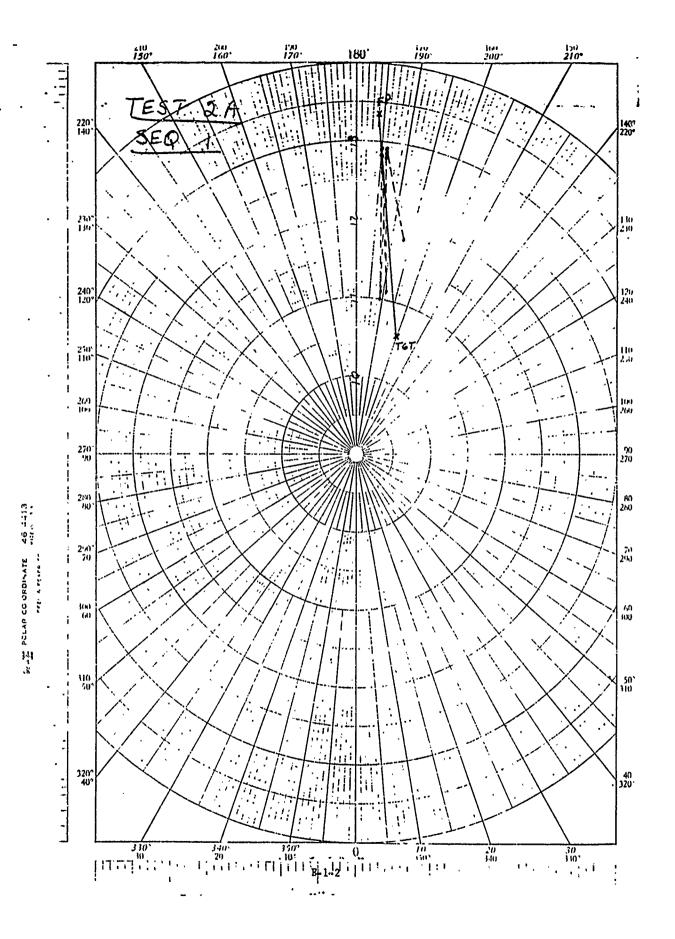
Appendix 1 - (Polar Plots) to Annex F of Report of AN/TPQ-31 Performance During Evaluation as a Hostile Weapon Locator

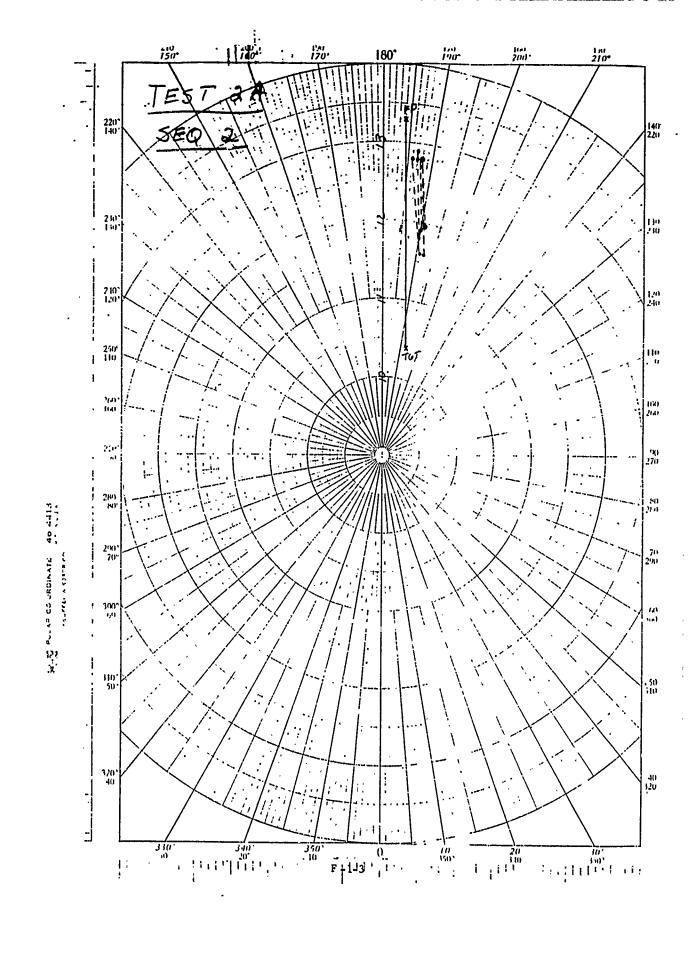
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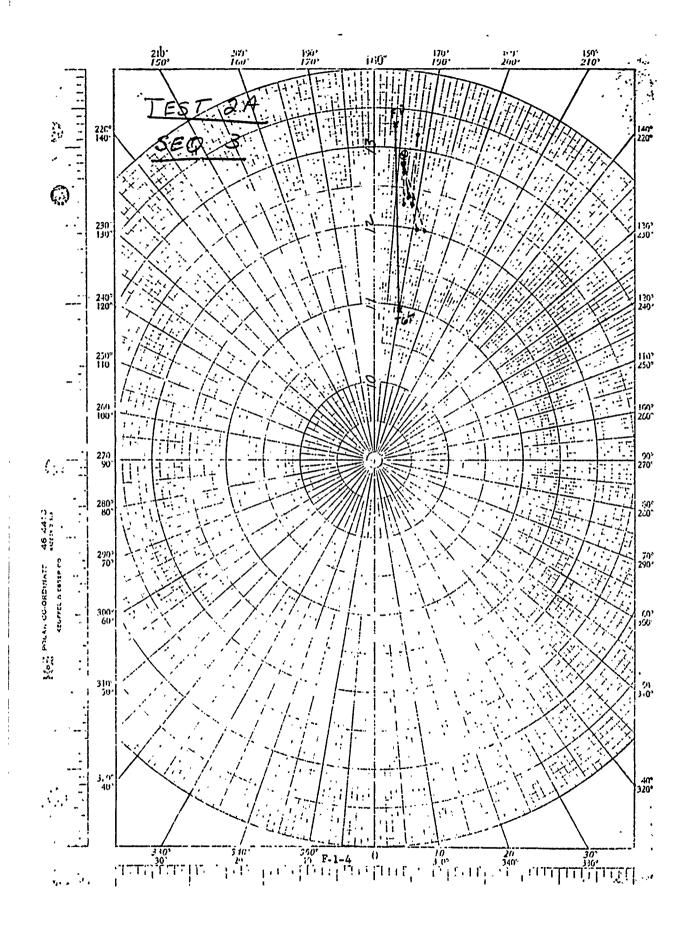
The following plots show in polar form the firing points and target points for each test. In addition, first looks and last looks are plotted with dashed lines connecting some of the points. This type of presentation allows a quick evaluation of the possibility of laving tracked the wrong projectile. These plots also readily reveal the existence of any biases present in the radar data.

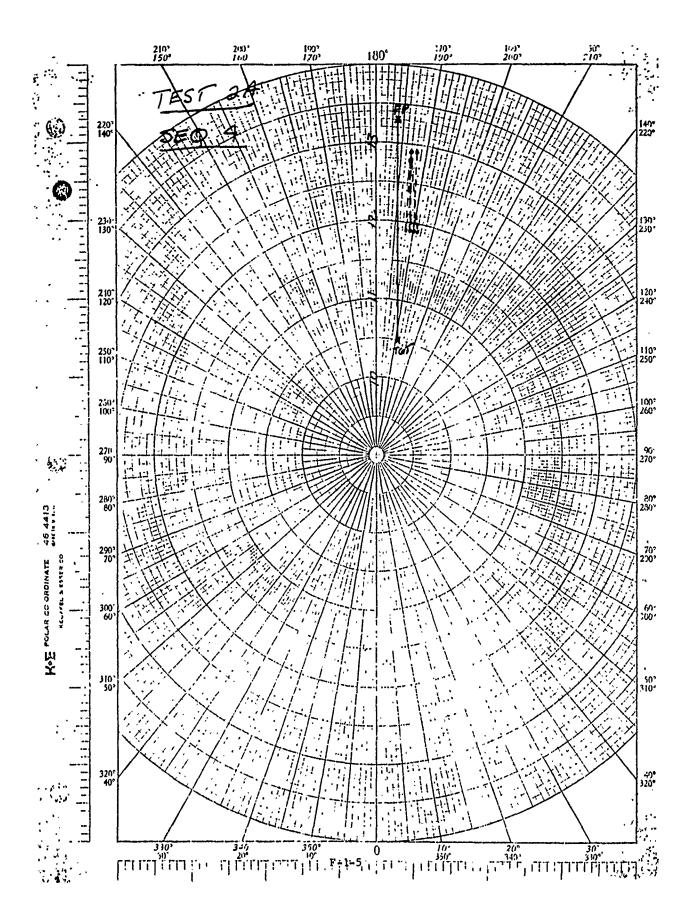
The displayed biases, in many cases, are caused by azimuth orientation errors that result from the manual boresite method used by the AN/TPQ-31. The use of an optical boresight telescope would help alleviate the biases attributable to azimuth orientation errors.

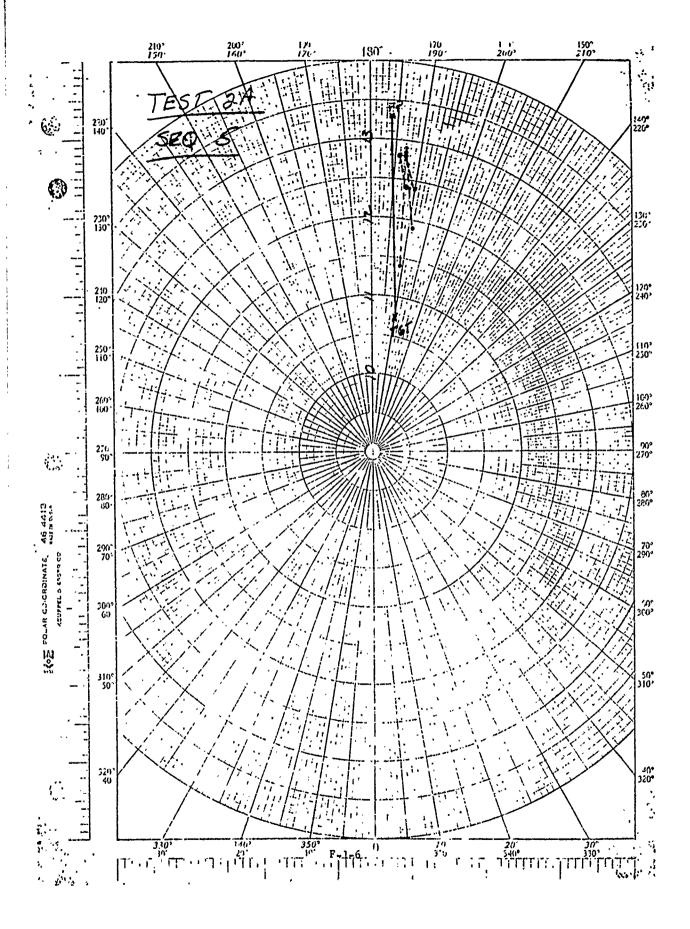


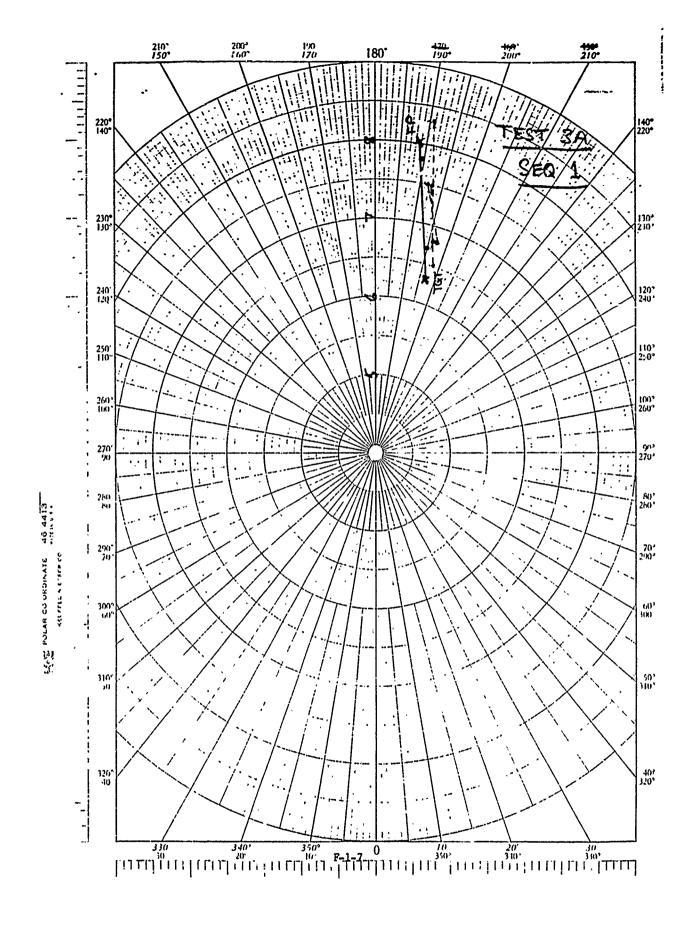


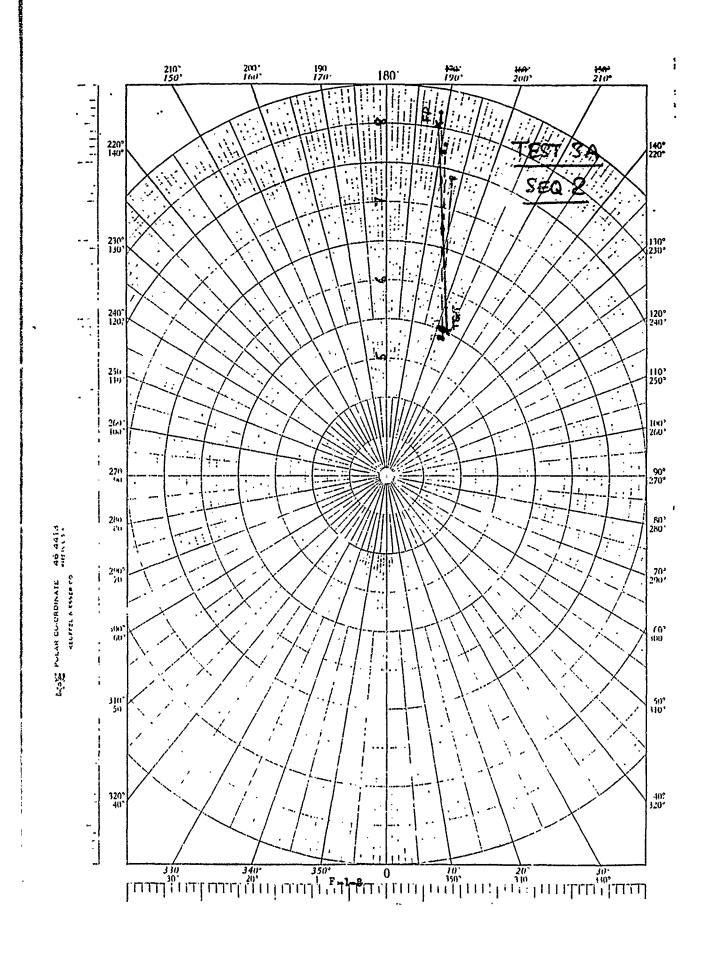
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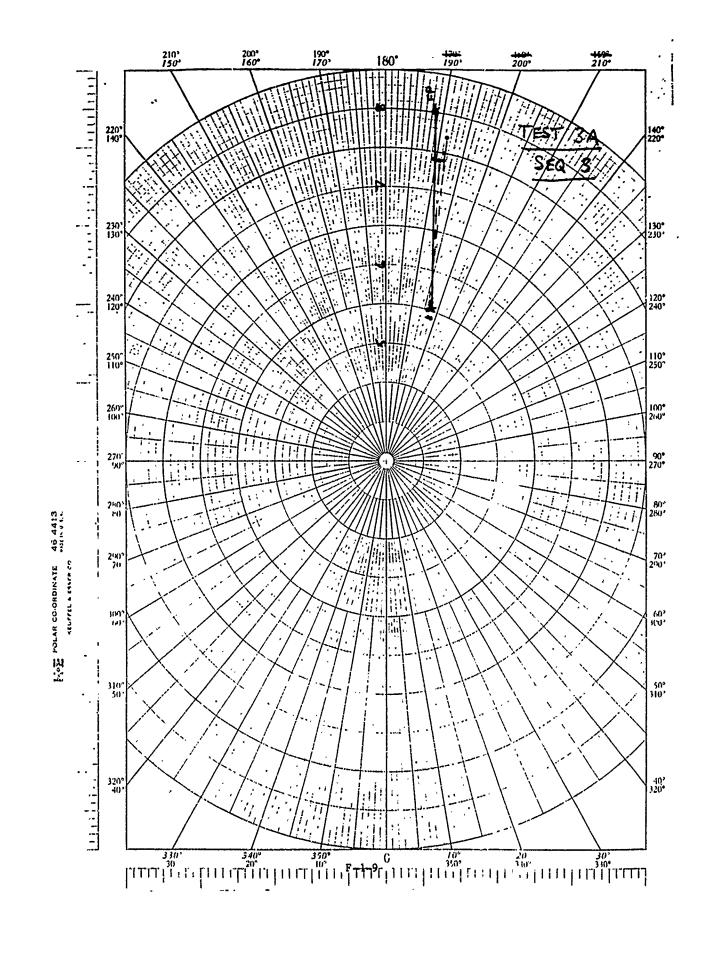


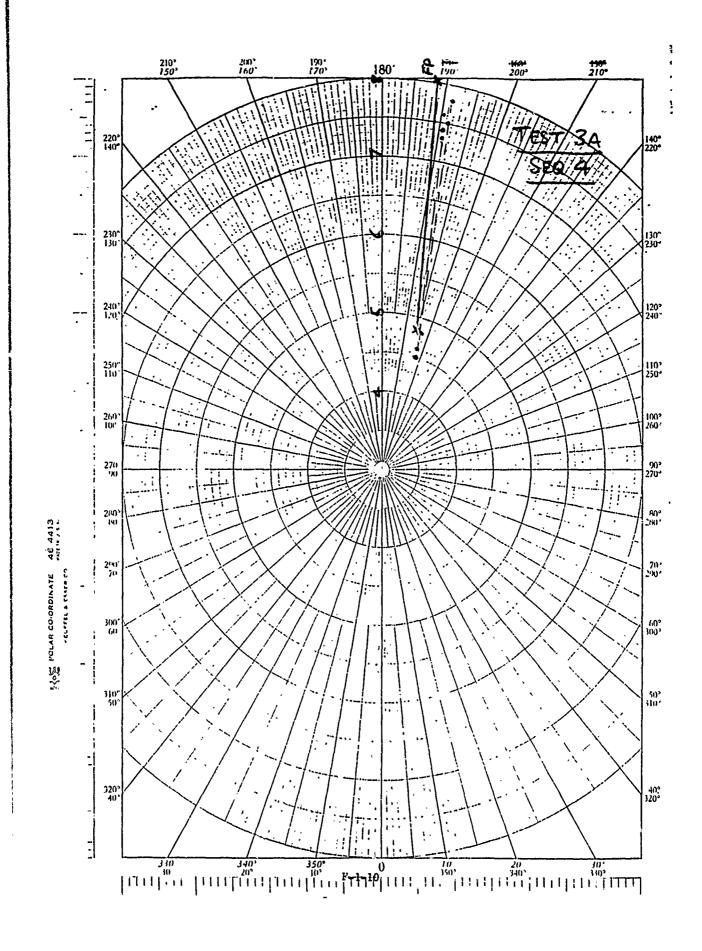


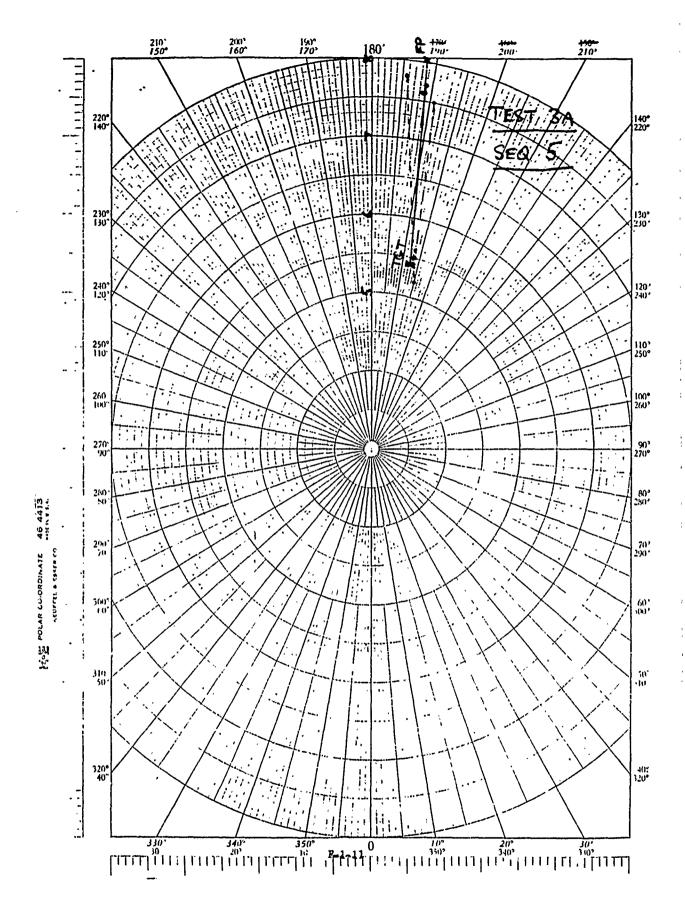


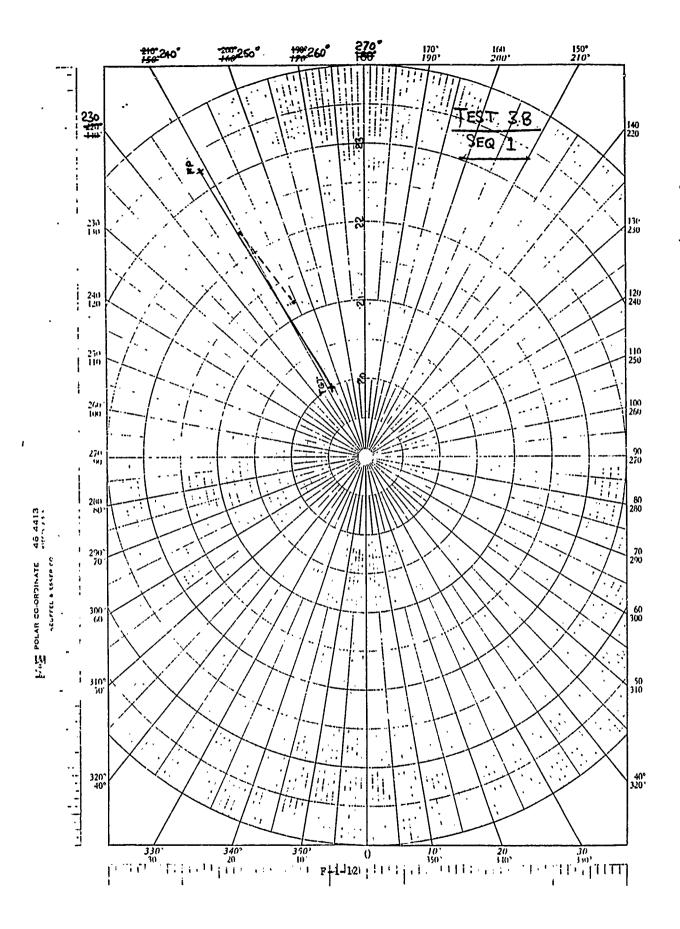


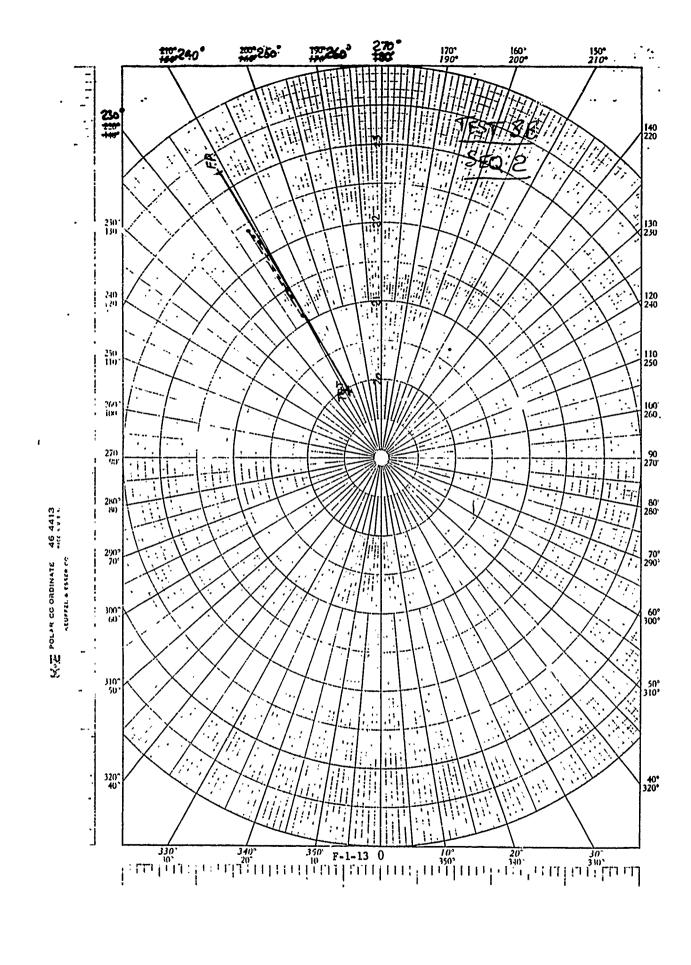


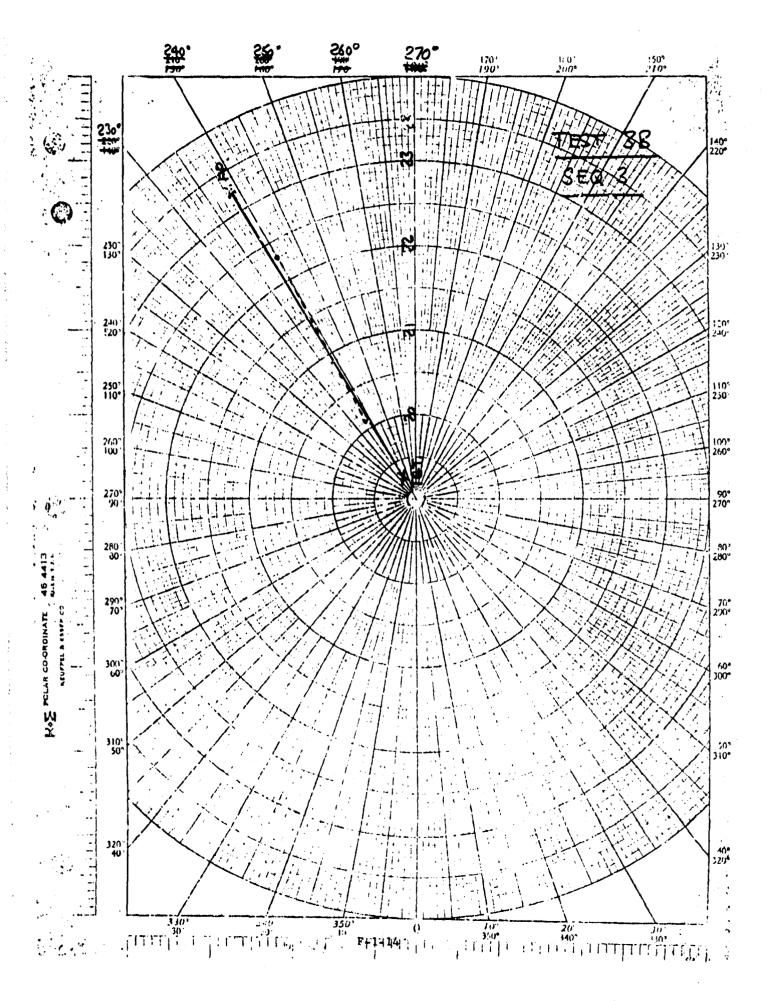


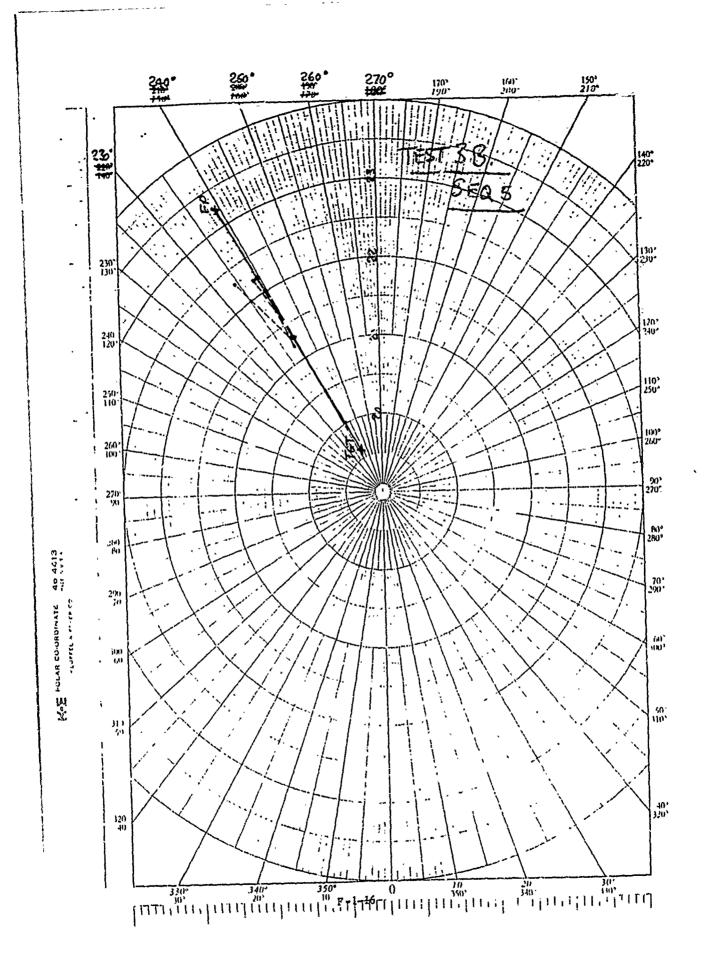


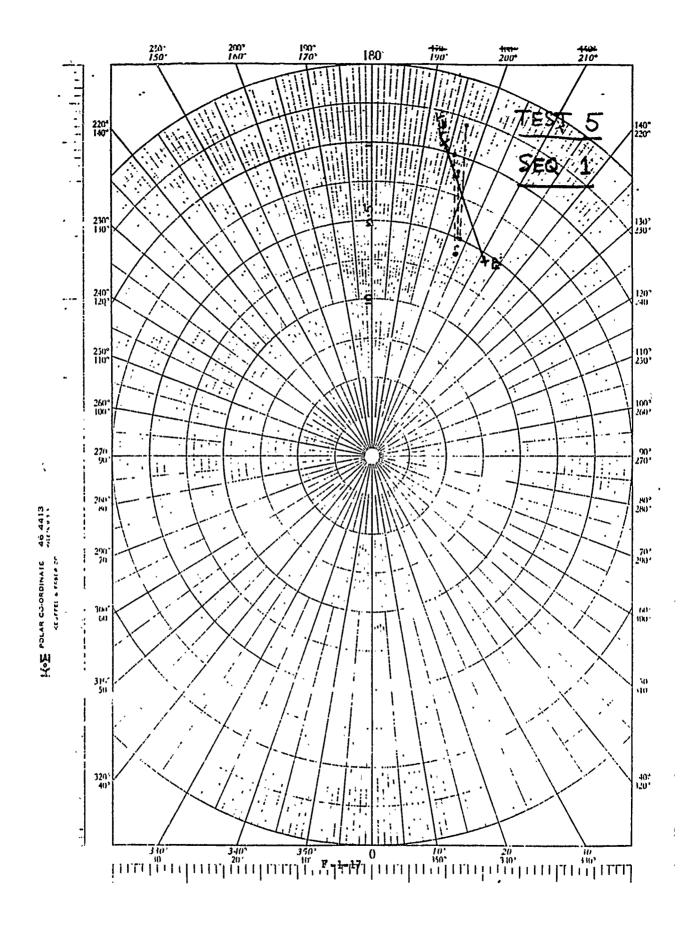


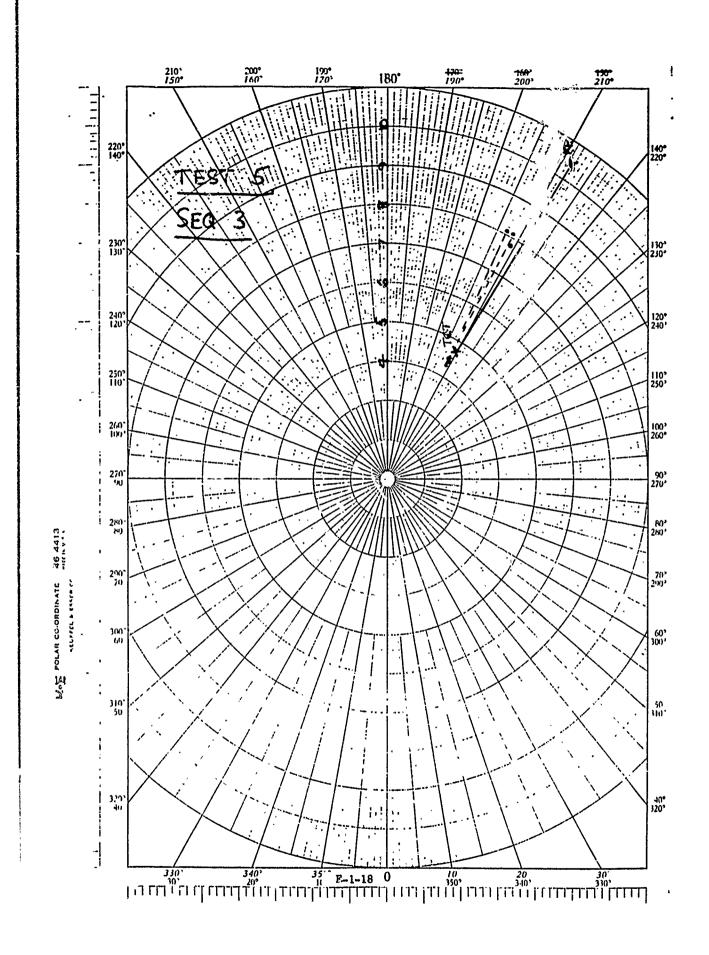


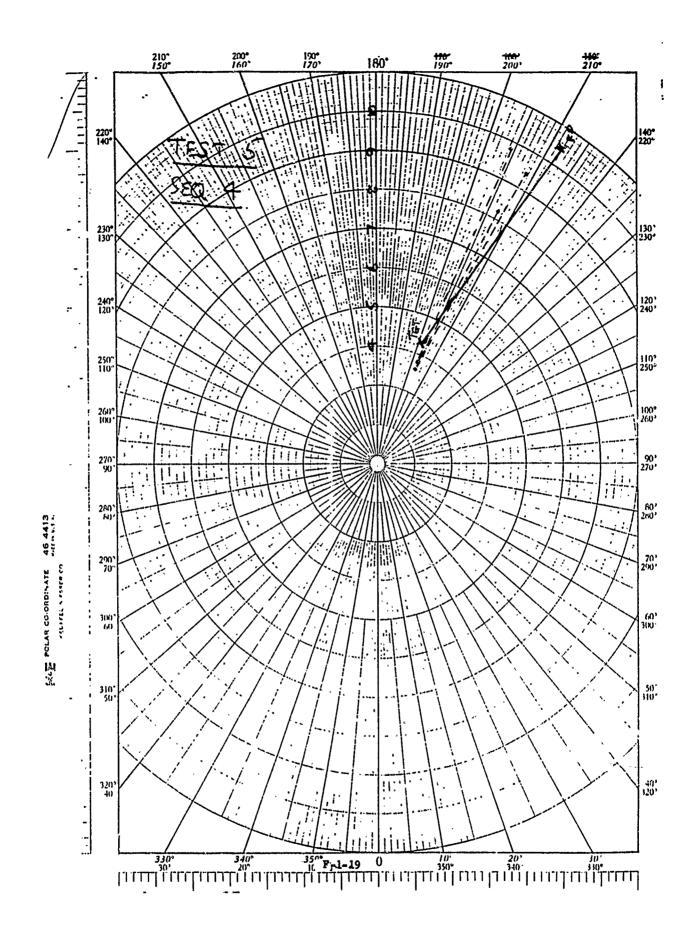


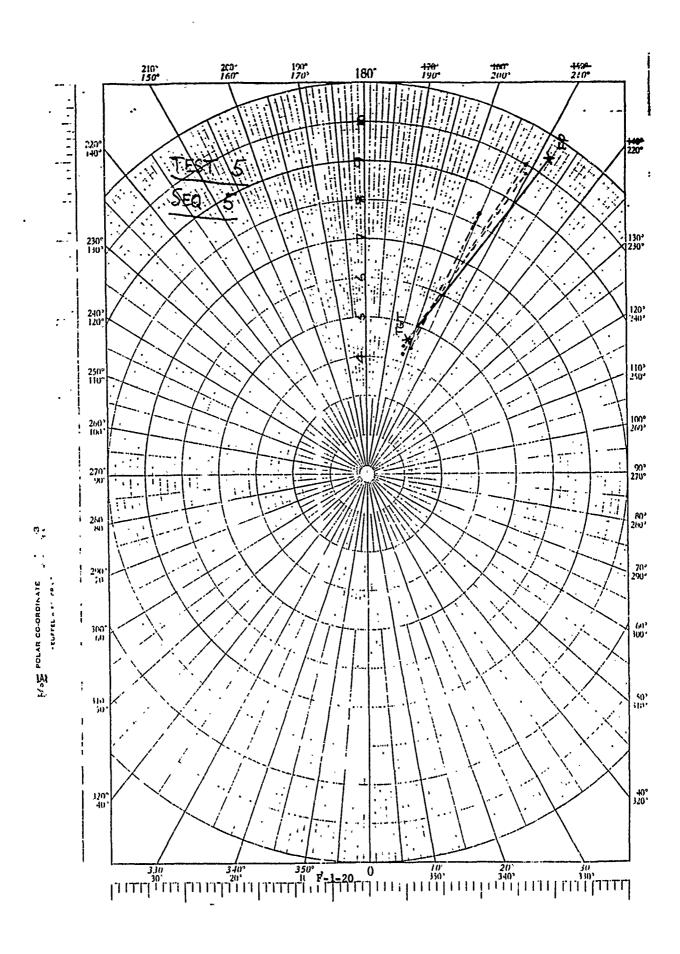


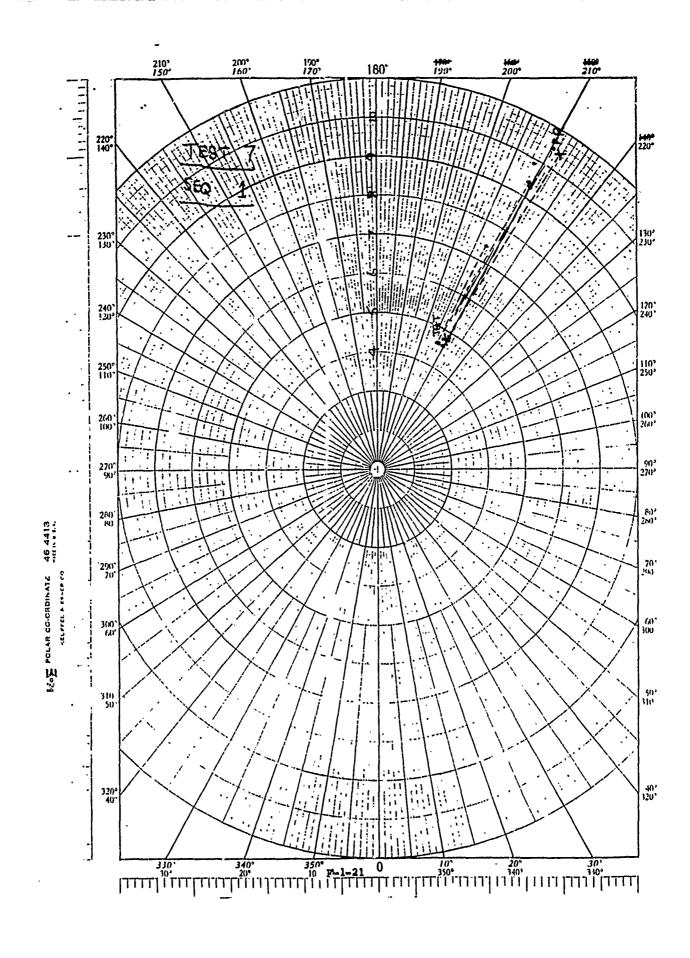


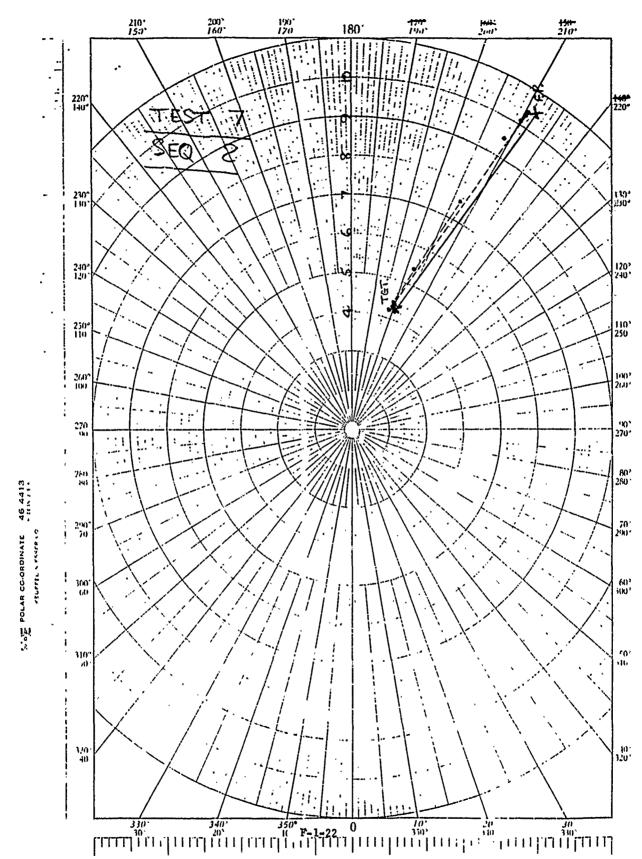






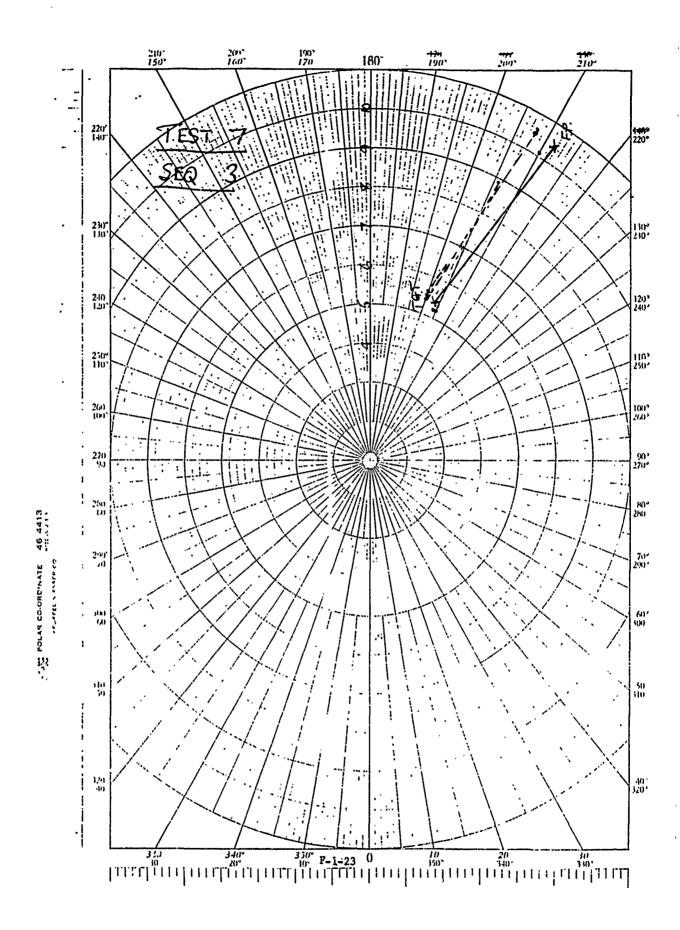


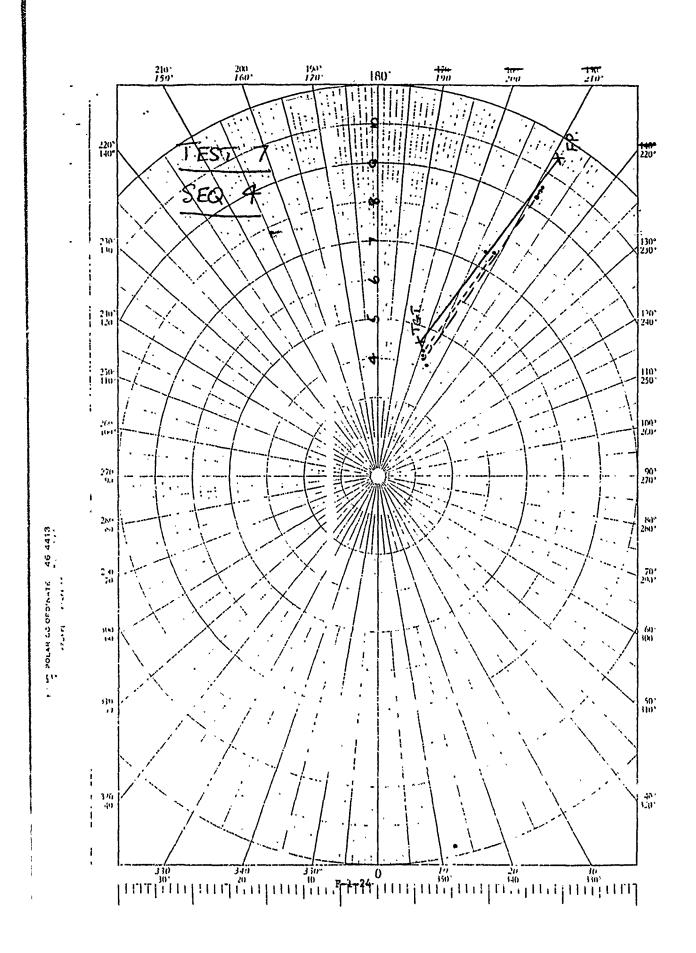


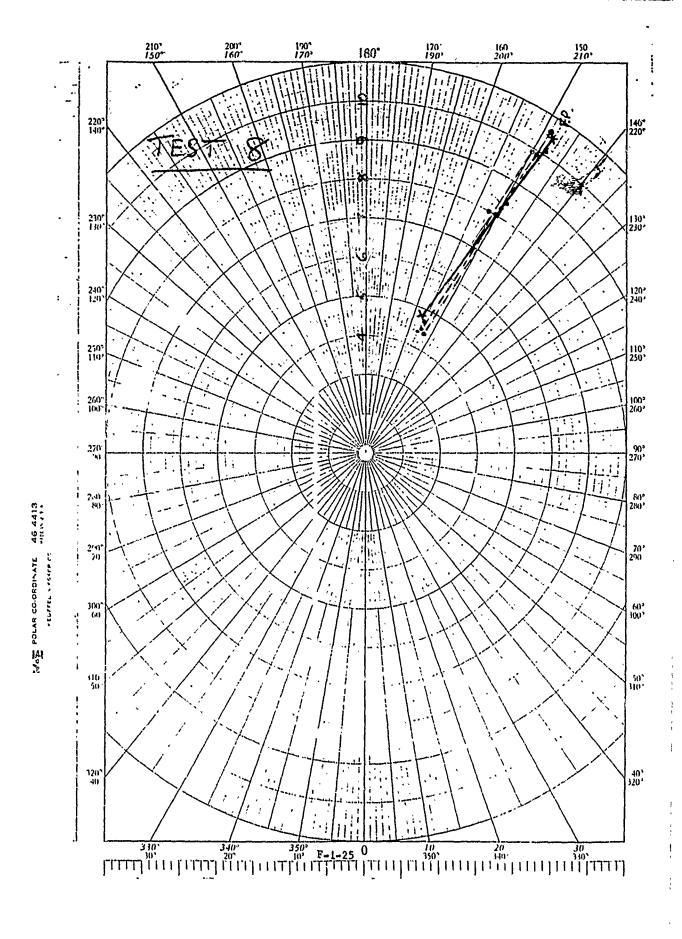


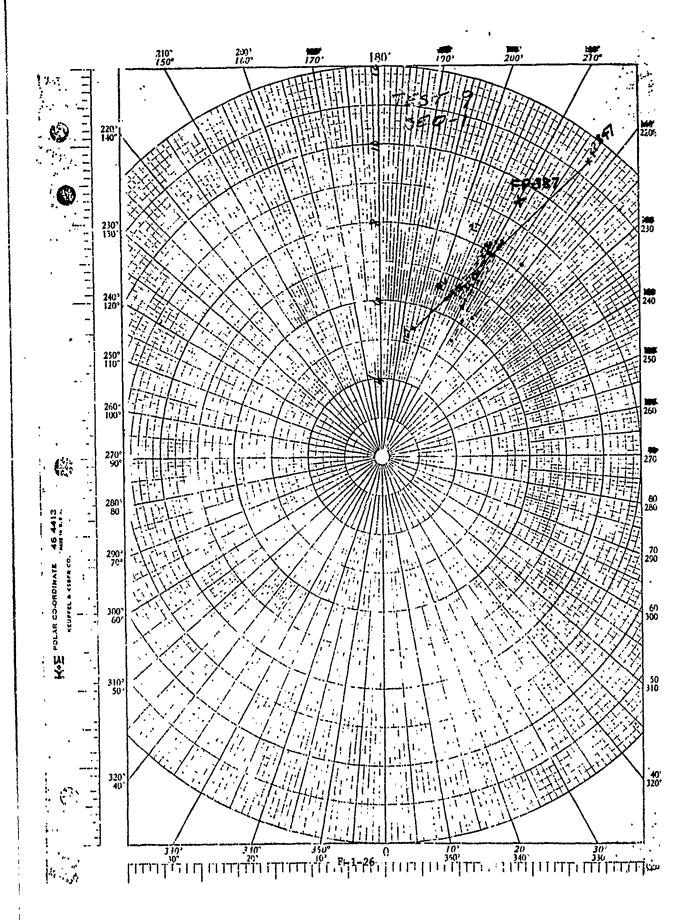
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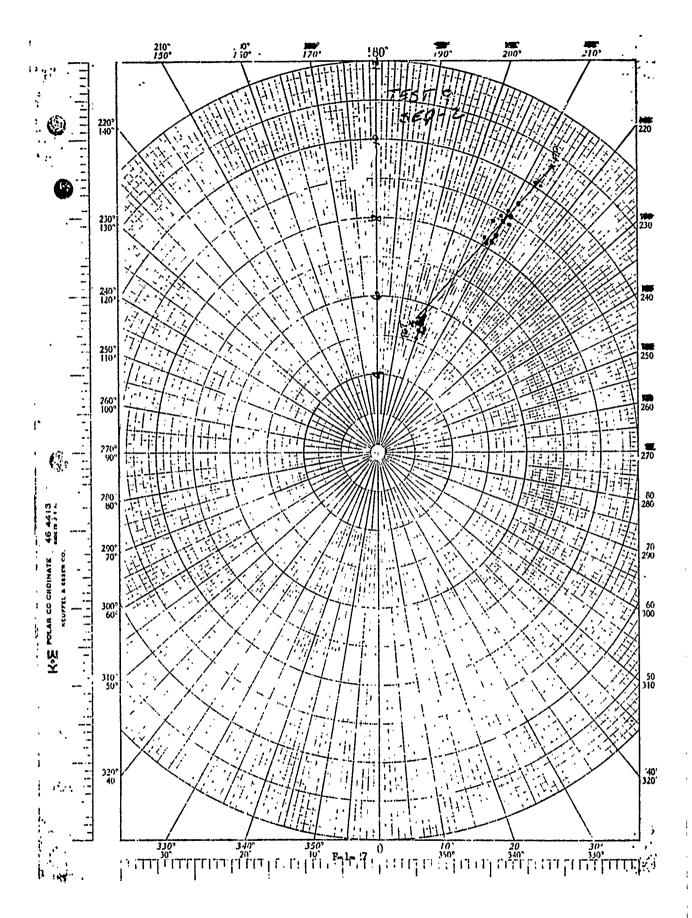
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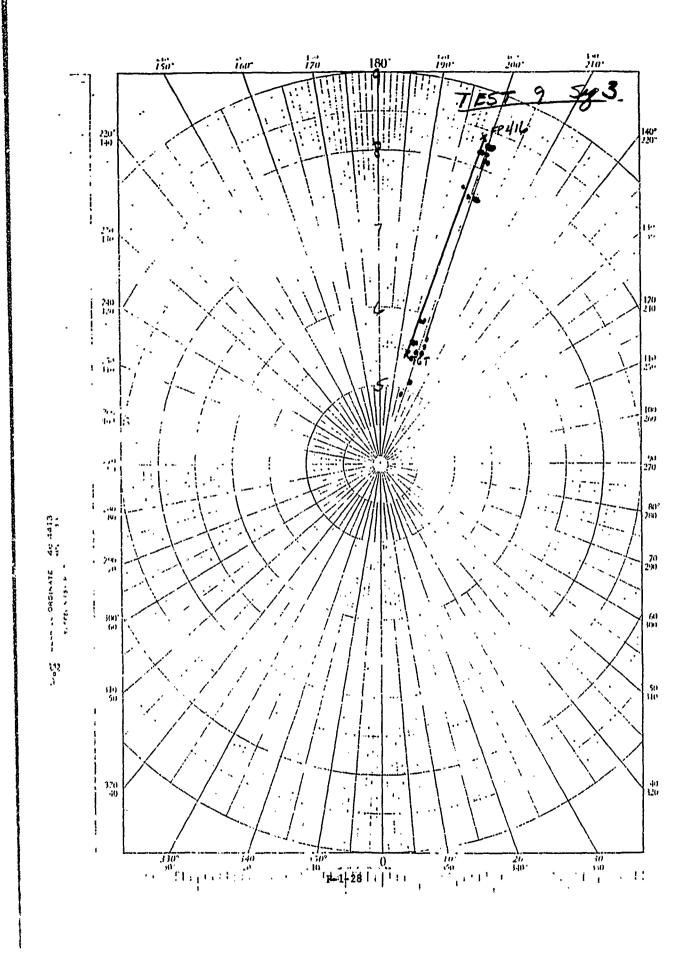


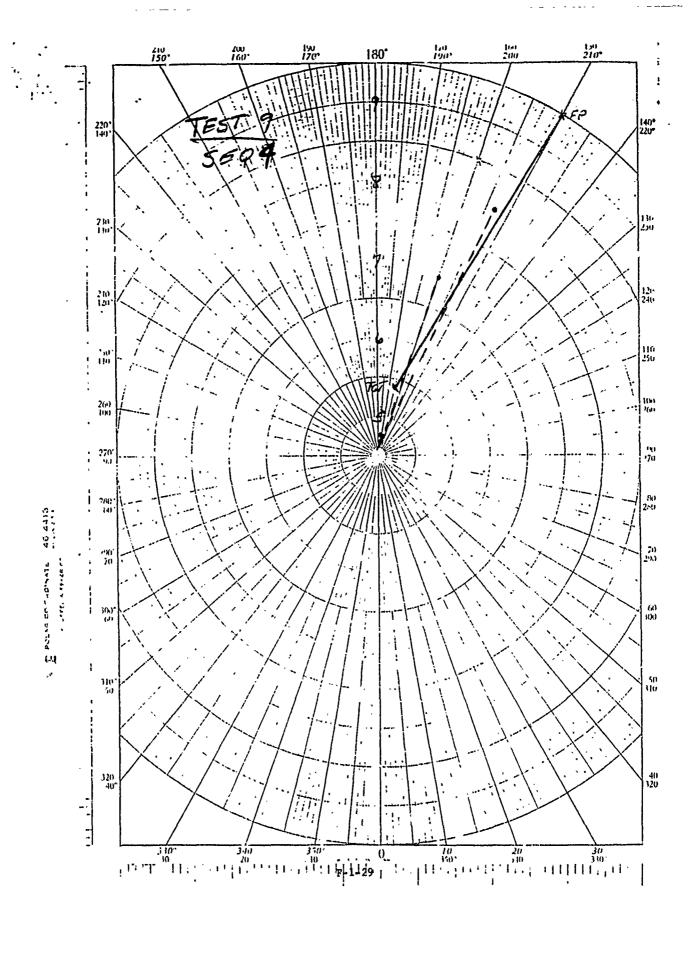


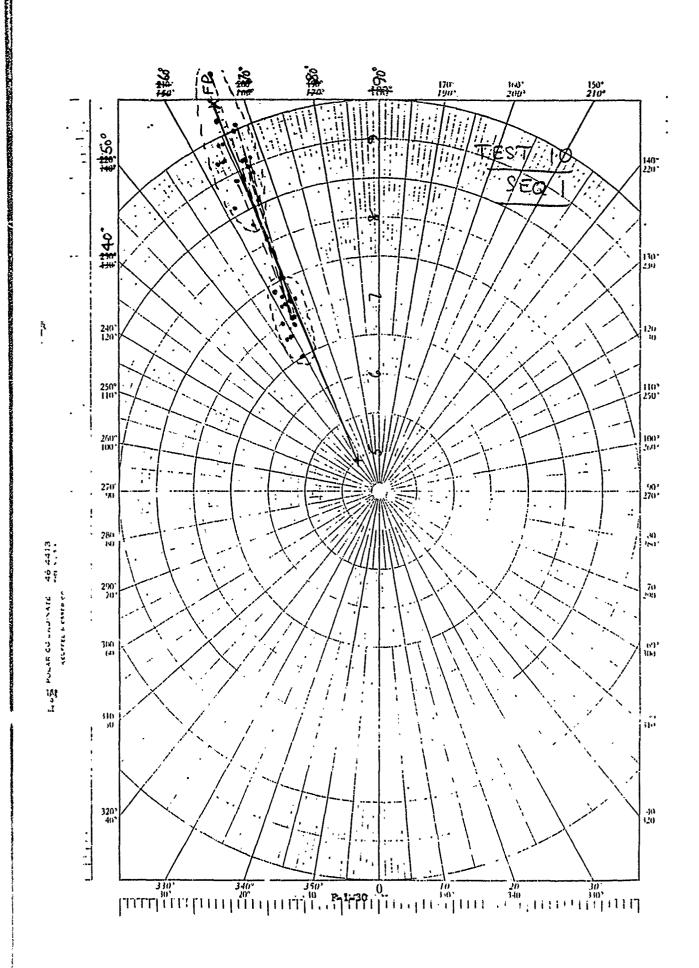


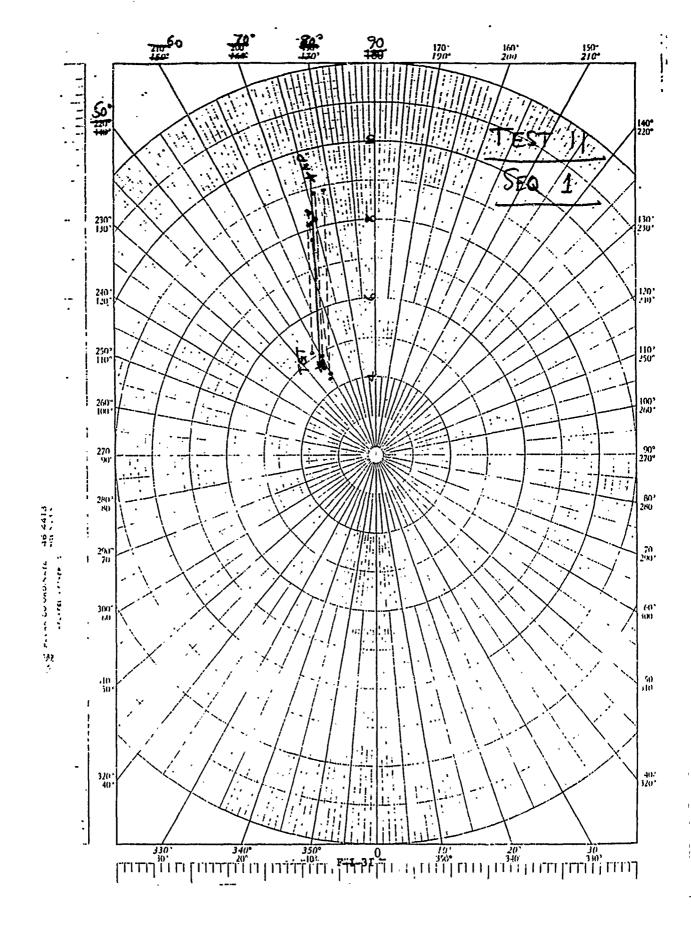


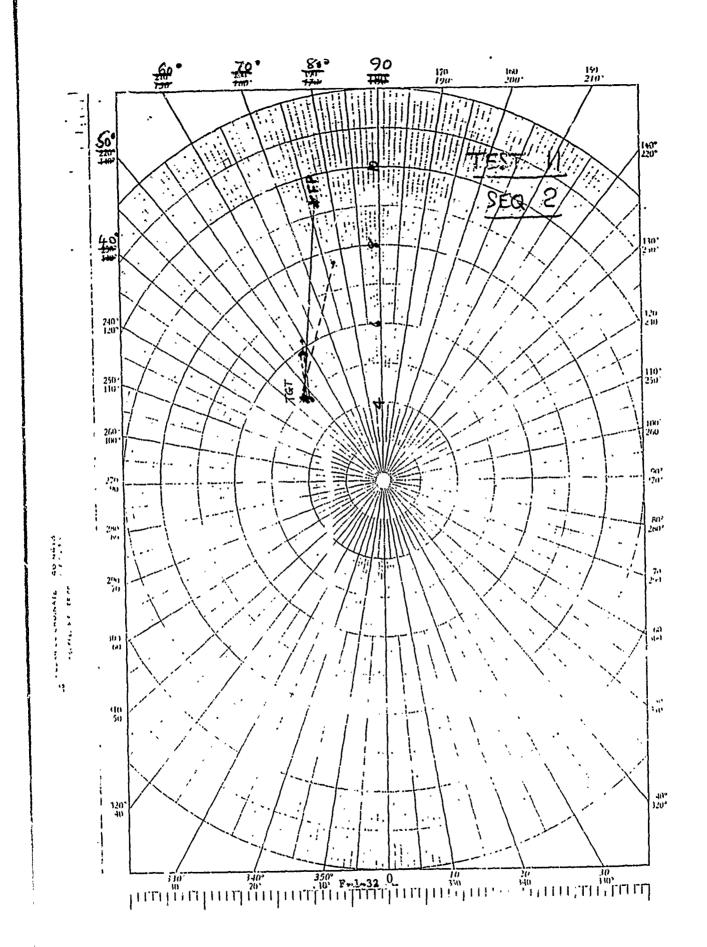


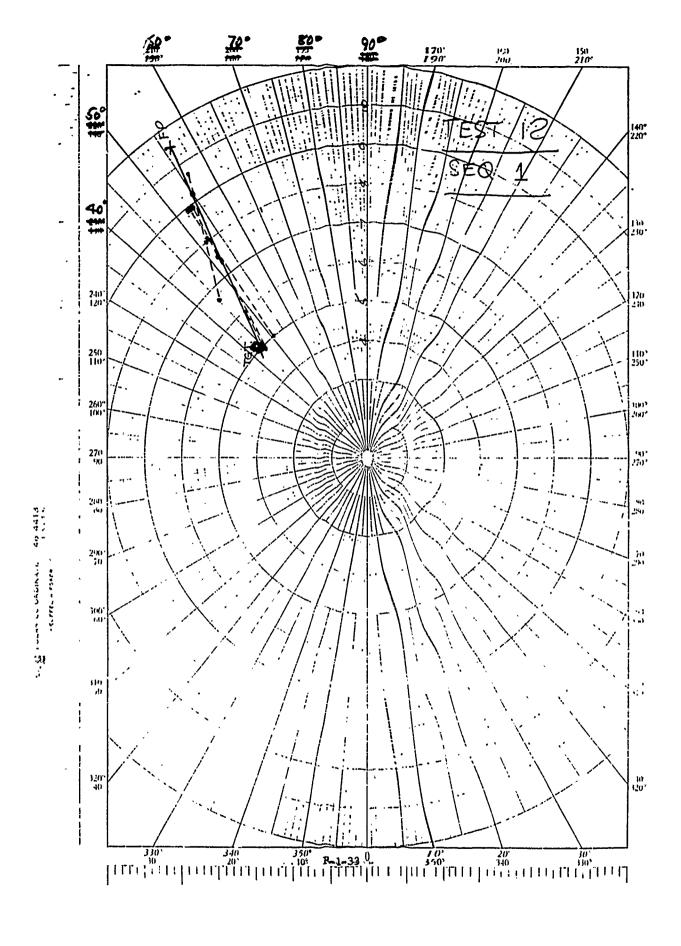


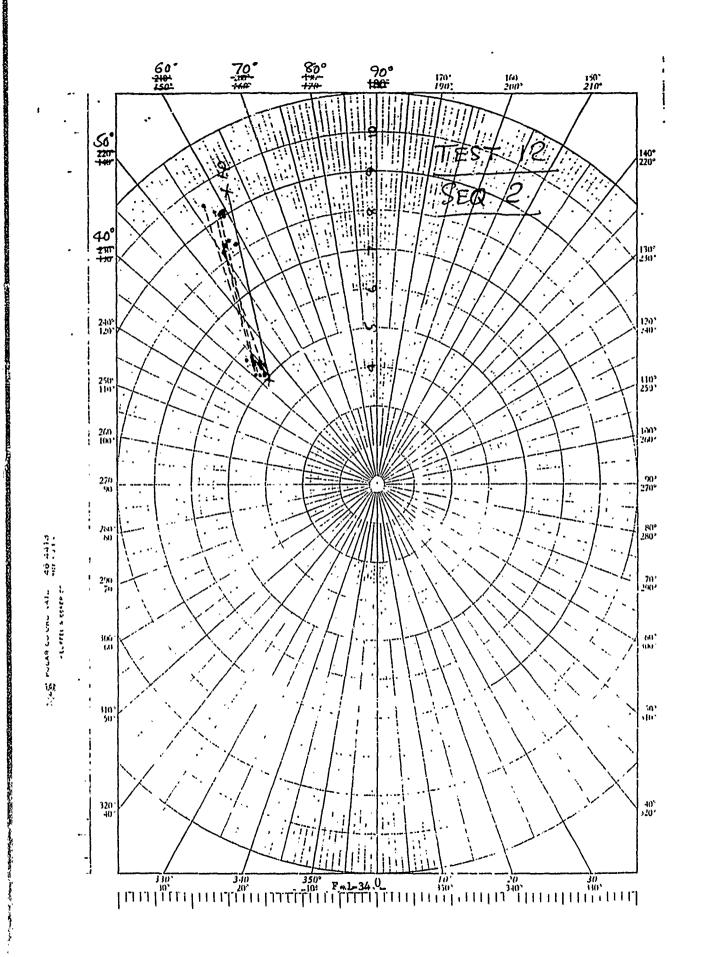




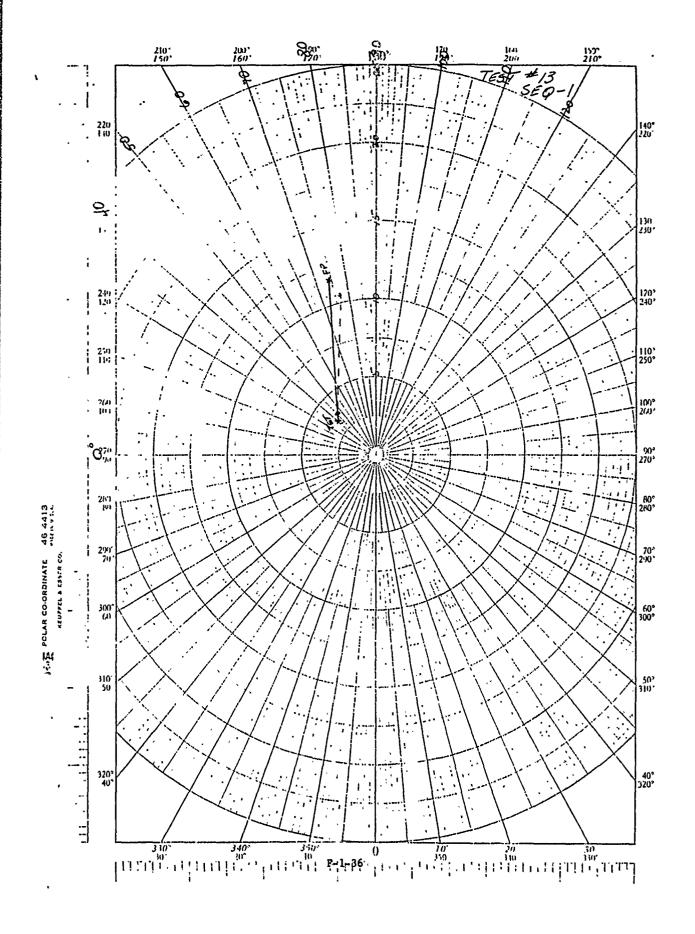


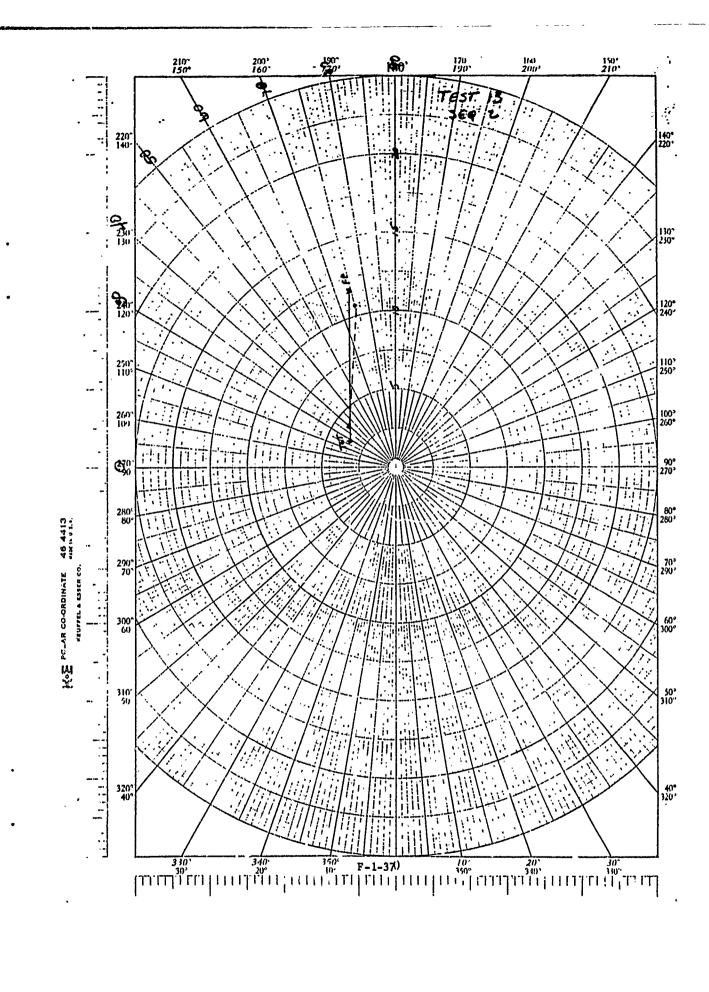


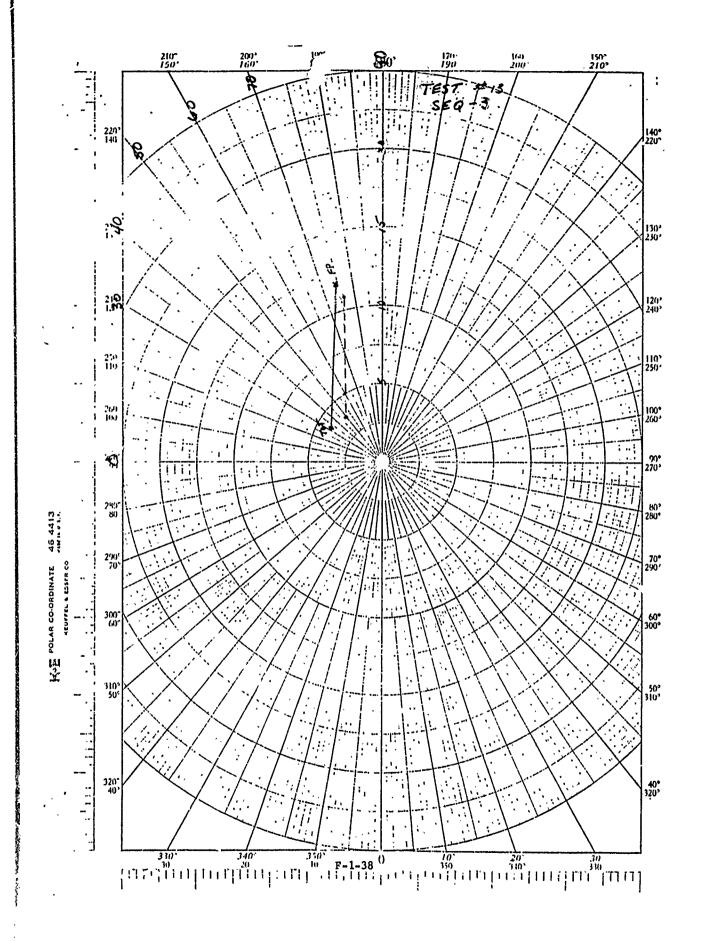


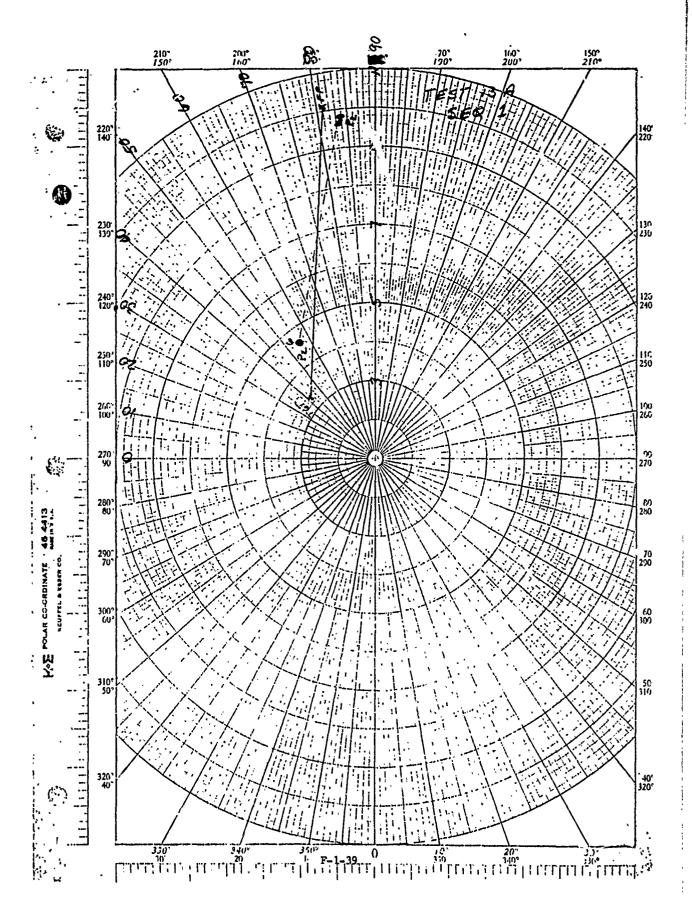


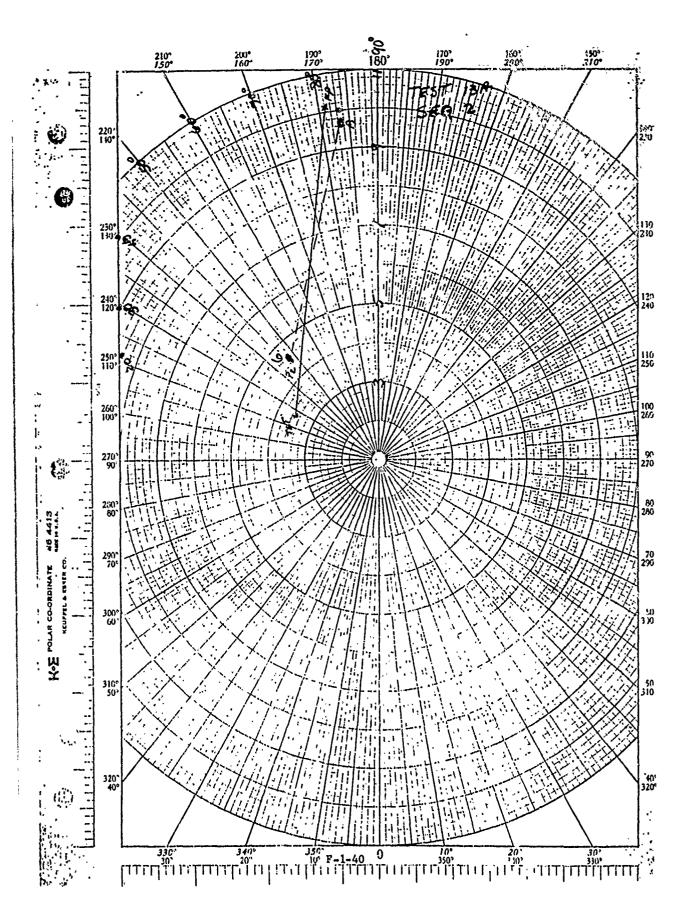
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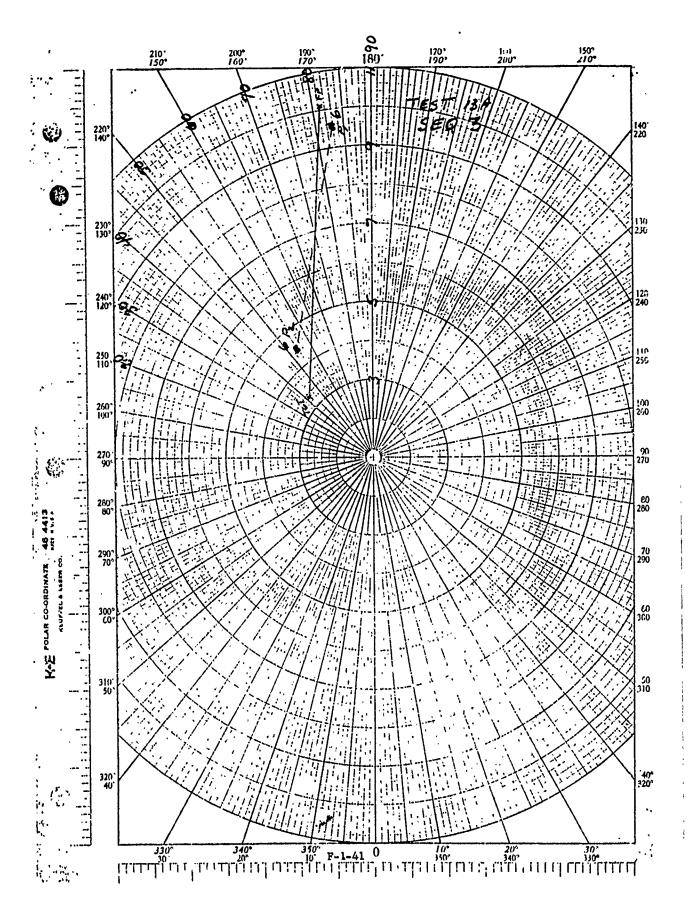


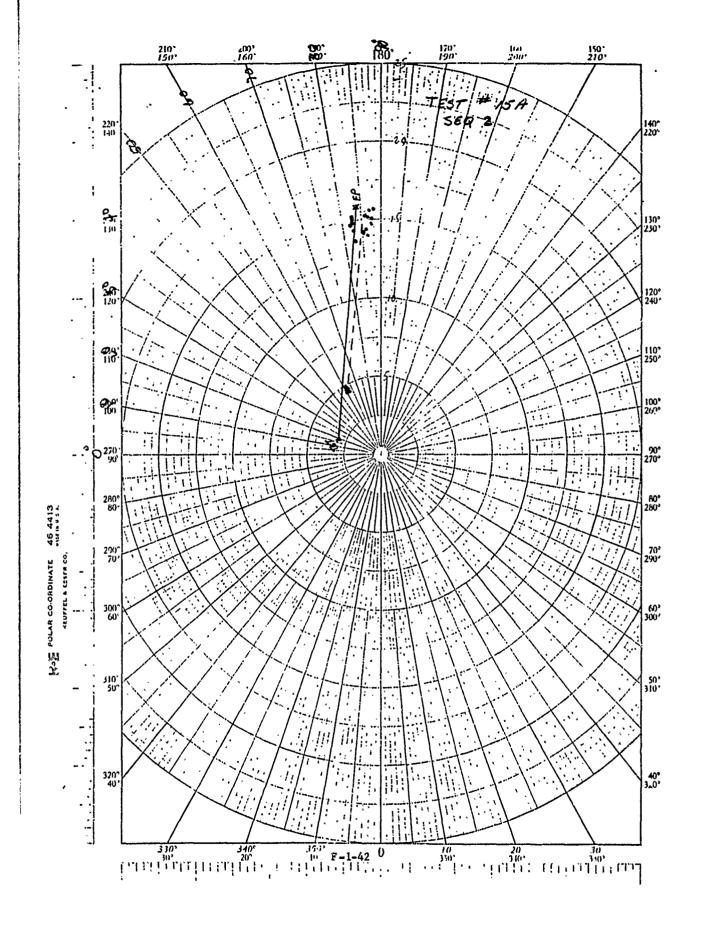


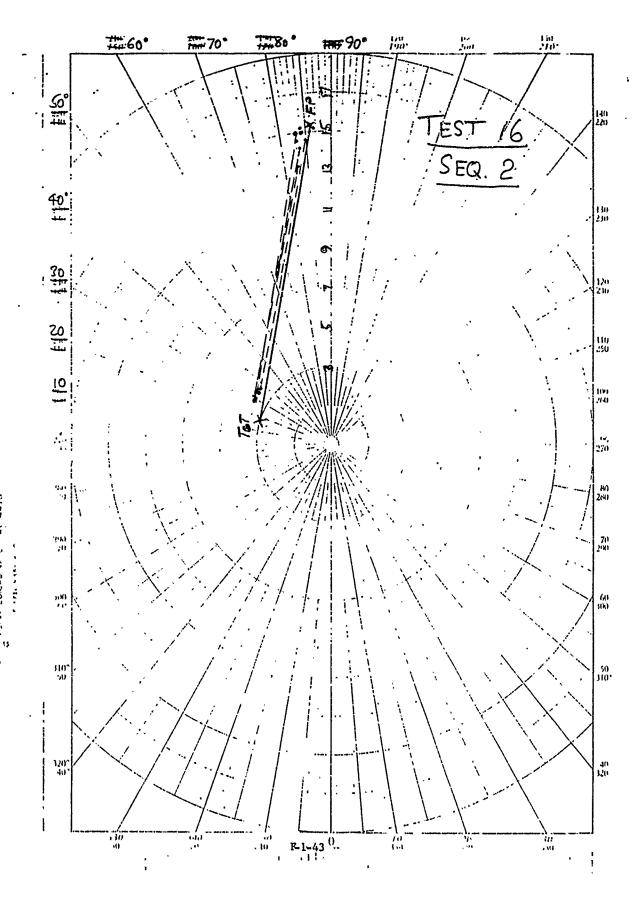


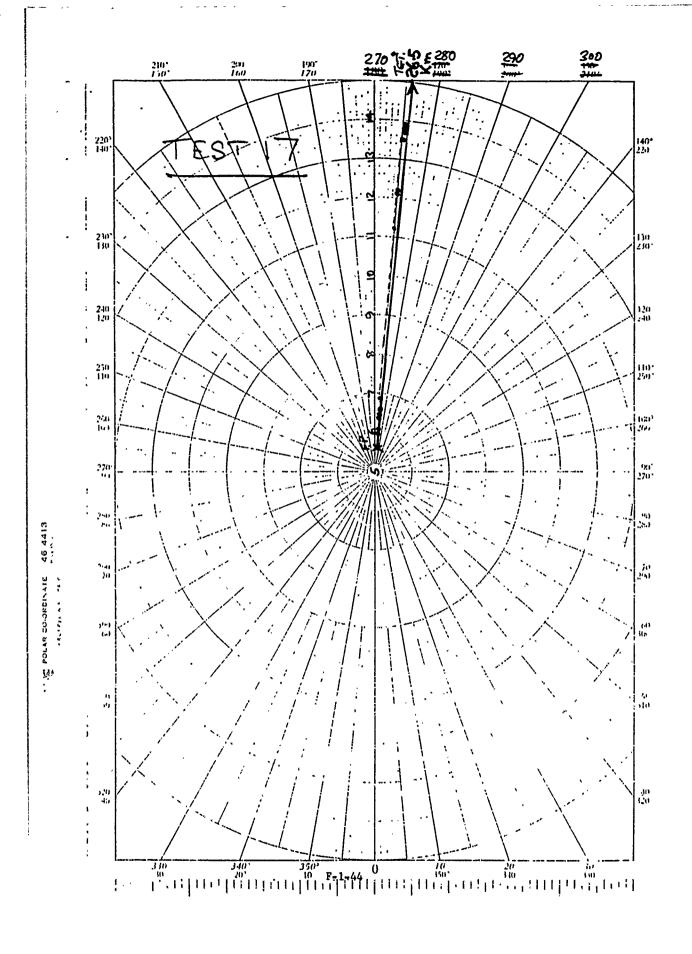


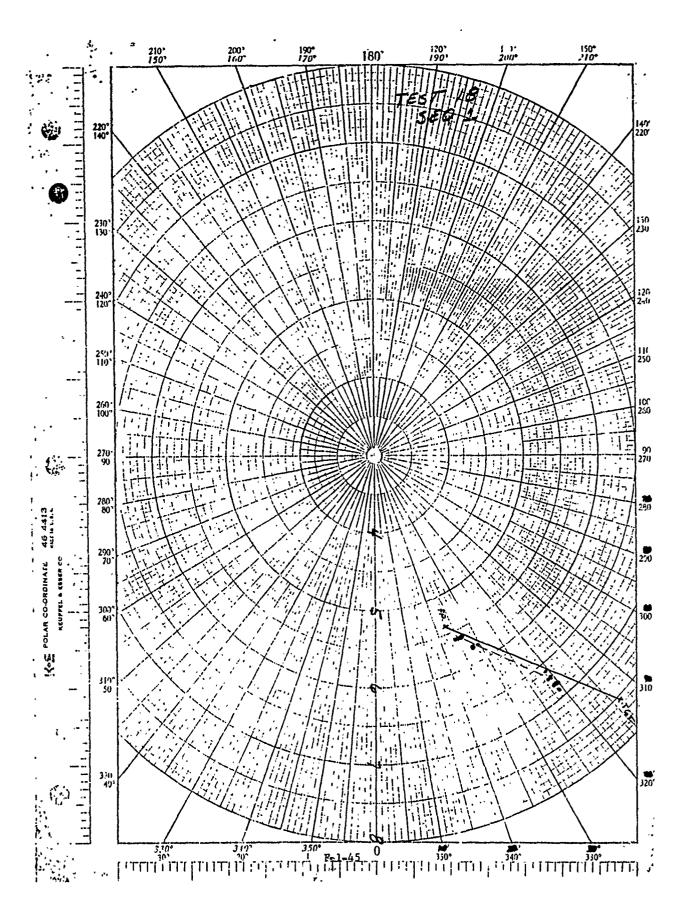


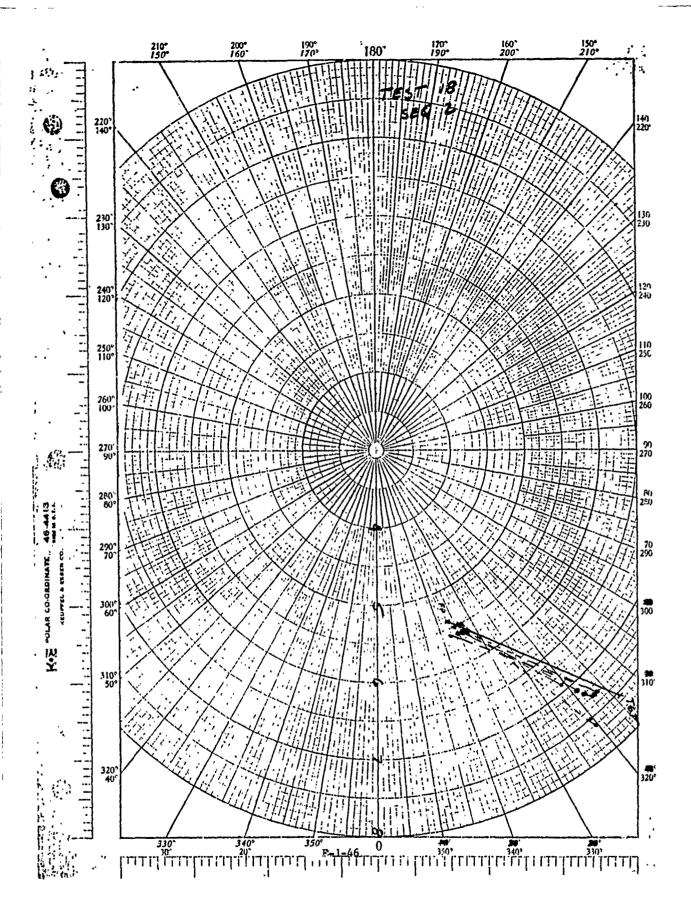


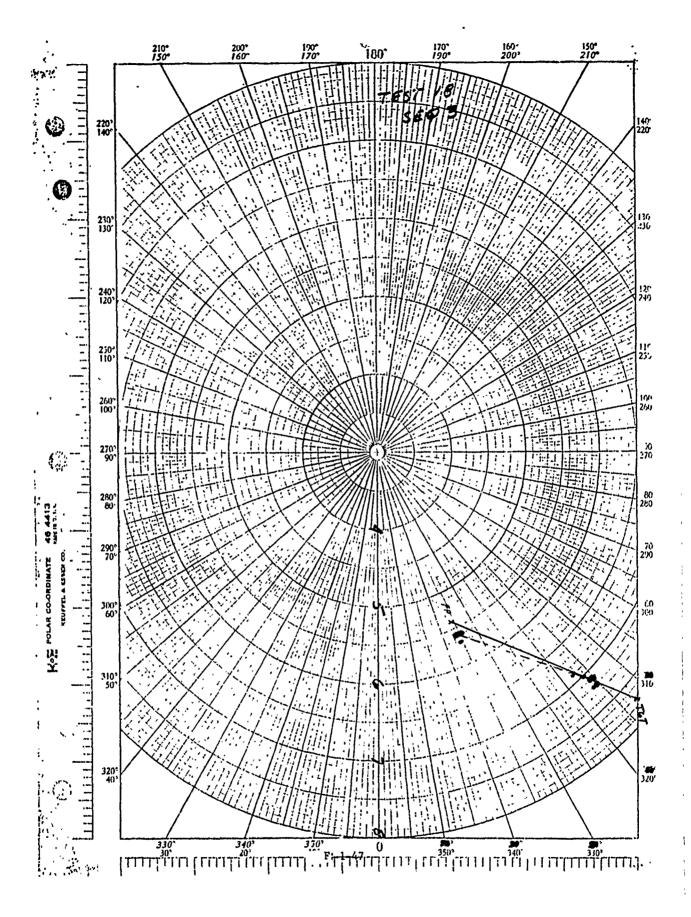


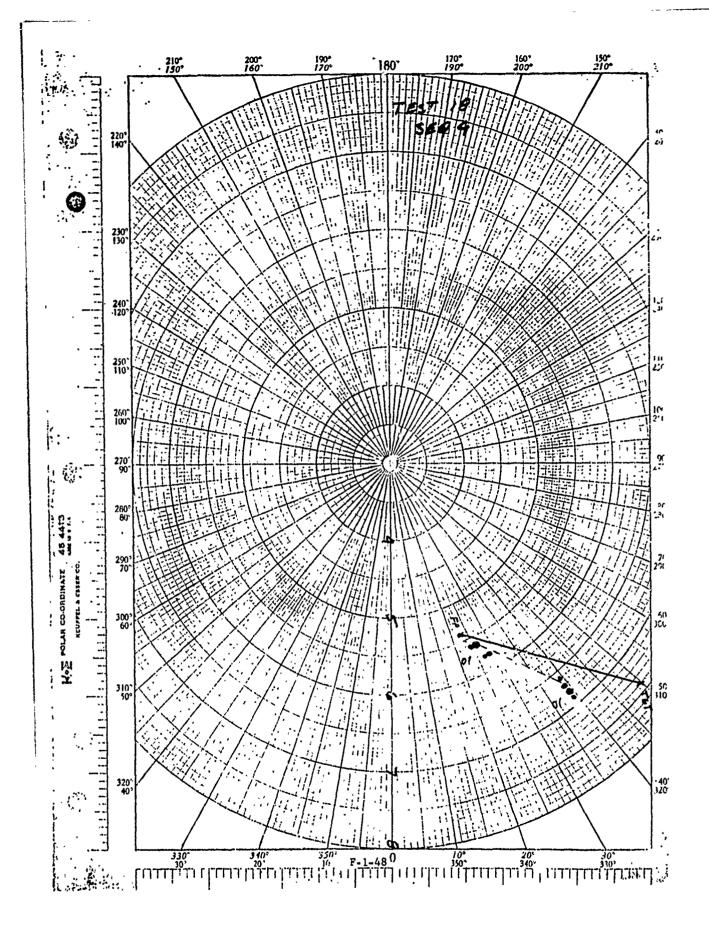












Appendix 2 - (Erroneous Tracks) to Annex F to Report of AN/TPQ-31 Performance

During Evaluation as a Hostile Weapon Locator

ERRONEOUS TRACKS

The data for Test 9/1 at first appear puzzling because the predicted detection curves indicate that good results should be obtained. However, the miss distance averages for this group are among the worst of all the data. A look at the polar plots for 9/1 reveal that there is a strong possibility that a different test shot firing from a range of approximately 8.5 km (FP-387) was observed in error. The first and last detections are typically at approximately 8.3 km and 6.6 km respectively. The true projectile should have produced 8-9 detections during the detectable flight time, four of which would be during the observed range (8.3 km to 6.6 km). The raw data gives an average of 7.3 detections during the observed range indicating that the target tracked was not the correct one, and in all probability originated at FP-387.

A similar analysis was conducted for certain other samples of data where very evident inconsistencies were noted among the detection curves, miss distances and polar plots. In those cases where the observed number of scans separating the first and last blip differed by over 50% from the true value obtained from the computer simulation, the shots were considered wrong identifications and removed from the averages for miss distances. The following rounds fall into this category:

Tests 7/1 one erroneous round 7/2 one erroneous round 7/3 one erroneous round 7/4 two erroneous rounds four erroneous rounds 9/1 all rounds erroneous 9/4 all rounds erroneous 12/3 all rounds erroneous

Appendix 3 - (Test Analysis Parameters) to Annex F to Report of AN/TPQ-31

Performance During Evaluation as a Hostile Weapon Locator

TEST ANALYSIS PARAMETERS

The following data lists some of the parameters for each test plus some pertinent comments. A brief explanation of the headings are as follows:

- 1. Test Type: Test types are either confirm or locate. For confirm tests, the general firing point and target area are known in advance. For locate tests, no information is given, and any trajectory observed is the recorded trajectory.
 - 2. Weapon Type: (Self explanatory).
 - 3. Range to Weapon: (Self explanatory).
 - 4. Miss Distance:
 - a. Operator Prediction: The miss distance is computed by backplotting the track on a map.
 - b. Machine Prediction: A calculator performs the equivalent of the above, but adds in a correction for the mask angle.
- 5. Blip Scan %: This is the observed ratio of the number of scans on which the targets were seen to the total number of scans during which the target was in the air.
- 6. Predicted Detection Curve: This is a prediction, arrived at by computer simulation, of the expected signal to noise ratio above a minimum detectable signal for the particular firing plotted for every 2 second (radar data rate) interval. Taken into account for this simulation are:
 - a. Range to target
 - b. Elevation to target and beam pattern

- c. Instantaneous aspect and pitch of the projectile and its corresponding radar cross section
- d. Range rate and the velocity notch
- e. Sensitivity vs time control (STC) attenuation
- f. Weapon type, QE, charge, initial range, initial aspect
- g. Slant range (2 second intervals).
- 7. Comments: (Self explanatory).

:						
	Comments	Firings 2/1 thru 2/5 not detected. Locally supplied Army generator failed causing problems with short range display.	•			
	Predicted Detection Curve	25 27 27 27 07 07 07 07 07 07 07 07 07 07 07 07 07	11.	C C C S		Co. Co. Co. Co. Co. Co. Co. Co. Co. Co.
Blip	Scan %	0	, 0	0	0 ,	0
stance	Mach. Scan Pred. %	ı	1	1	1	ı
Miss Di	Op. Pred.	t	1	t	ı	1
	Range to Wea.	8.2 Km	8.2 Km	8.2 Km	8.2 Km	8.2 Km
_	rea. Type	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar
	Test Type	Con- firm	Con- firm	Con- firm	Con- firm	Con- firm
	Test # ✓Seq #	2/1	2/2	F -3-3	2/4	2/5

Comments	The 2A/1 thru 2A/5 are characterized by non-detectability during the first 6 seconds of flight. The low detectability may be attributed to distance from weapon. Best M. Dist.: 149 m	This firing showed a strong azimuth bias, see polar plot, Appendix G-l Best M. Dist.: 378 m	Best M. Dist.: 508 m	Веst M. Dist.: 529 m	This firing showed a strong azimuth bias, see Appendix G-1 Best M. Dist.: 681 m
Predicted Detection Turve	00 G C C C C C C C C C C C C C C C C C C	20 20 30 30 30 30 30 30 30 30 30 30 30 30 30	02 CE CE CE CE CE CE CE CE CE CE CE CE CE	20 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	30 01 02 03 03 03 04 04 05 04 05 05 05 05 05 05 05 05 05 05 05 05 05
Blip Scan %	28%	26%	36%	37%	31%
Distance Mach. Pred.	360.8 neters	732.6 meters	596.0 meters	667.0 meters	802.3 meters
Miss Di Op. Pred.	330.0 meters	389.0 meters	462.0 meters	520.0 meters	440.0 meters
Range to Wea.	13.25 Km	13.25 Km	13.25 Km	13.25 Xm	13.25 Km
Ean. Dype	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Yor- Car
Tast Type	Con- firm	Con- firm	Con- firm	Con- firm	Con- firm
1285. 3. 18. 18. 18. 18. 18. 18. 18. 18. 18. 18	2A/1	2A/2	E/YZ 5-4	2A/4	2A/5

	Jaments	Firings 2B/1 thru 2B/5 had a radar problem - shorted wire (burnt out) in distribution box. See Appendix 4, Annex F.				Reproduced from best available copy.
	Predicted Paraction Surve	25 25 25 25 25 25 25 25 25 25 25 25 25 2		01 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		65 67 68 69 61 61 61 61 61
Stip	در	Ö	0	0	0 ,	0
Distance Blip	Pred.	1	t	•	. 1	ı
Miss Di	Pred.	1	•	1	1	
Range	to Wea.	4.45 Km	4.45 Km	4.45 Km	4.45 Km	4.45 Km
Wea	Type	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar
	r Type		Con- firm	Con- firm	Con- firm	Con- firm
Test #	/ 8	2B/1	2B/2	5-3-5	28/4	28/5

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Comments	The firings 3/1 thru 3/5 were missed because of problems with locally supplied Army generators. Problem with short range display - instability in horiz. sweep circuit was also encountered.				
Predicted Detection Curve	27 G		C) C) C) C) C) C) C) C) C) C) C) C) C) C	2 2 2 2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3	25 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
Blip Scan %	0	0	0	o .	0
Miss Distance Blip Op. Mach. Scan Pred. [2xed. %	t	t	ı	, 1	ı
Miss I Op. Pred.	1	ı	1	1	ŧ
Range to Wea.	13.4 Km	13.4 Km	13.4 Km	13.4 Km	13.4 Km
Wea. Type	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar
Test	Lo- cate	Lo- cate	Lo- cate	Lo- cate	Lo- cate
Tast A	3/1	3/2	F-3.46	3/4	3/5

	a high ection es and s were ily 90%.				
Commerts	Firings 3A/1 thru 3A/5 show a high predicted probability of detection over the complete trajectories and the observed blip scan ratios were correspondently high - normally 90's Best M. Dist.: 164 m	Best M. Dist.: 252 m	Best M. Dist: 413 m	Best M. Dist.: 337 m	Best M. Dist: 242 m
Prolibed Detection Curve	20 C C C C C C C C C C C C C C C C C C C				
Blip Scan %	79%	%06	%06	. 92%	%88
Stance Mach. Pred.	334.8 meters	379.1 meters	528.0 meters	472.0 meters	292.4 meters
Miss Distance Op. Mach. Pred. Pred.	417.0 meters	470.0 meters	484.0 meters	434.0 472.0 meters meters	314.0 meters
Range to Wea.	8.02 Km	8.02 Km	8.02 Km	8.02 Km	8.02 Km
Wea. Type	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar
Test	Lo- cate	Lo- cate	Lo- cate	Lo- cate	Lo- cate
Test *	3A/ī	3A/2	F-3-7	3A/4	3A/5

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Comments	The 3B/1 thru 3B/5 firings started at 23 km and ended at 19 km ranges. Therefore, the prediction curves are generally below the threshold. The blip scan percentages were low (generally 20%) as expected resulting in large miss distances. Best M. Dist.: 885.8 m	.: 758 m	.: 1172.4 m	.: 821.6 m	.: 804.6 m
	The 3B/1 thru 23 Km and ender Therefore, the generally belo scan percentag as expected re distances. Best M. Dist.:	Best M. Dist.:	Best M. Dist.:	Best M. Dist.:	Best M. Dist.:
Predicted Detection Carve	20 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	33	10 00 00 00 00 00 00 00 00 00 00 00 00 0	8.5 0 0 2 2 8	20 00 00 00 00 00 00 00 00 00 00 00 00 0
5; ip Scan %	23%	23%	23%	23%	17%
Distance Mach. Pred.	885.8 meters	915.8 meters	1172.4 meters	902.1 meters	858.9 meters
Miss Di Op. Pred.	640.0 meters	746.0 meters	900.0 meters	828.0 meters	792.0 meters
Range to Wea.	23.23 Km	23.23 Km	23.23 Km	23.23 Km	23,23 Km
Eea. Type	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar	4.2" Mor- tar
Test f. Type	Lo- cate	Lo- cate	Lo- cate	Lo- cate	Lo- cate
Test	38/1	3B/2	6/86 F-3-8	38/4	3B/5

Соптеп С	Problem with antenna pedestal. Bad brush, oil leak and partially slezed bearing.				
Pradicted Datation Curve	2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2	des de 13 ; 2 ; 14 mp de 15 ° 2 4 mp m	22 23 24 25 27 27 27 28 28 28 28 29 20 20 20 20 20 20 20 20 20 20 20 20 20	2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2	0; 0; 0; 0; 0; 0; 0; 0; 0; 0; 0; 0; 0; 0
Blip Scan %	0	0	0	0 .	
stand	ı	•	ı	. ,	
Miss Distance Blip Op. Cach. Scan Pred. Pred. %	ı		1	•	
Range to Wea.		14.05 Km	105mm 14.05 Km How.	175mm 14.05 km Gun	
Type		4.2" Mor- tar			
Test	Wea- pon Ident	Wea- 4.2" pon Mor- Ident tar	Wea- pon Ident	Wea- pon Ident	
Test #	4/1	4/2	£/ / 7 F-3-9	4/4	

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	Comments	Firings 5/1 thru 5/5 initial trajectories were velocity blind resulting in initial detection being too late to permit meaningful firing point prediction. Best M. Dist.: 463.9 m	These firings never were above detection threshold.	Best M. Dist.: 1250 m	Best M. Dist.: 339 m	Best M. Dist.: 622 m
	Predicted Detection Curve	00 00 00 00 00 00 00 00 00 00 00 00 00	╏╶╏╌╟╶╏╌┠╼╂╼╂ ╸ ┫╸ ╸╃═╏═╧╸╇╌ ┨╶╏	00 00 00 00 00 00 00 00 00 00 00 00 00	20 00 00 00 00 00 00 00 00 00 00 00 00 0	8 3 2 0 2 3 3
- 4.	Blip Scan %	30%	%0	29%	83%	298
	Distance Mach. Pred.	537.1 meters	,	1173.1 meters	787.0 meters	230.1 meters
	Miss Di Op.	1293 meters	1	1693 meters	2163 meters	1106 meters
	Ranse to Mou.	10.4 Km	10.4 Km	10.4 Кт	105um 10.4 Km How.	105mm 10.4 Km How.
	Wea.	105шш	105mm How•	105mm How•	105um How.	105mm How.
	Test Type	·	Confirm	Con- firm	Con- firm	Con- firm
	Tast # Seq #	5/1	5/2	£/5 F-3-10	5/4	5/5

Comments	All these firings were in velocity notch, therefore not seen.	Wrong track out Miss dist. 363 m Best M. Dist.: 259 m	Wrong track out Miss dist. 124 m Best M. Dist.: 135 m	Wrong track out Miss dist. 346 m Best M. Dist.: 155 m	Wrong track out Miss dist. 803 m Best M. Dist.: 658 m	This series of firings show good predicted detectability. Miss distance results are poor as operators tracked rounds not associated with test. Best M. Dist.: 66.7 m
Predicted Detection Curve	3 3 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2					
Blip Scan	0	87%	74%	72%	. 28%	77%
Distance Nach.	t	836.1 ameters	630.0 meters		1584.1 meters	449.3 meters
Miss Dia Op.			880.0 6	1316.0 820.5 meters meters	2008.0 meters t	1030.0 4
Range to Wea.	10.25 Km	105mm 10.44 Km 1015.0 How.	10.44 Km	105гт 10.44 Кг	10.44Km	10.44Km
Wea. Iype	105mm 90w.	105mm How.	105mm How•	105тт Ном,	105mm How.	155mm How.
Test		Con- firm	Con- firm	Con-firm	Con- firm	Con- firm
Test # Seq #	ý	1/1	⁷ / _{F-3-11}	7/3	7/4	٥

Commen ts	In firings 9/1 thru 9/2, operator tracked some rounds not associated with test. These non-valid tracks were interspersed in the collected data which led to large miss distances.	M. Dist.:	Best M. Dist.: 140 m	Firing 9/4 shows influence of non-valid rounds as the predicted detectability indicates no round should have been seen. Best M. Dist.: 1360 m	Firings 9/5 and 9/6 show detection was not possible.
Predicted Dataction Curve					
Blip Scan %	37%	777.	63%	39%	0
Distance Mach. Pred.	3180.0	1731.0 meters	384.6 meters	2050.3 meters	ı
Miss Di Op. Pred.	2870.0 3180.0 meters meters	1376.0 meters	534.0 384.6 meters	1200.0 meters	ı
Range to Wea.	11.35 Km	10.7 Km	8.36 Km	9.5 Km	8.2 Km
Wea. Type	105mm How.	105mm How•	105mm How.	105mm How•	105шп
Test Type	Lo- cate	Lo- cate	Lo- cate	Lo- cate	Lo- cate
Test #	9/1	9/2	€/ 6 F-3-12	7/6	5/6

Comments		The prediction curve shows poor detectability return & falling in the clutter notch regardless of whether wide or narrow notch was used. This condition also contributes to scatter on polar plot, Appendix 1, Annex F. Best M. Dist.: 203 m	Same as above all but last 4 seconds cut of clutter notch. The generator also stopped which killed the keep alive voltage causing loss of receiver crystals of a local radar.	Firings 11/1 and 11/2 are also almost completely in clutter notch causing low blip to scan ratios, which lead to large miss distances. Best M. Dist.: 284 m	Best M. Dist: 1104 m
Predicted Detection Curve		ACINAL MARCHANIA		S S S S S S S S S S S S S S S S S S S	
 Scan %	0	21%	0	20%	211
Distance Mach. Pred.	1	594.4 meters	ı	866.8 meters	3032.2 meters
op. Pred.	ı	563.0 594.4 meters meters	\$	955.0 meters	3741.0 3032.2 meters meters
Range to Wea.	7.4 Km	9.95 Кт	9.95 Km	9.3 Km	9.3 Km
Wea. Type	105mm How.	105mm How.	105mm How.	105mm How.	105mm How•
Test Type	Lo- cate	Con- firm	Con- firm	Lo- cate	Lo- cate
Test #	9/6	10/1	7 7 7 7 7	11/1	11/2

Comments	Firings 12/1 thru 12/3 are in the velocity notch resulting in low detectability and excessive miss distances. Best M. Dist.: 194 m	In addition to low detectability 12/2 and 12/3 displayed excessive azimuth blases on the polar plot, Appendix 1, Annex F. This is result of tracking non-valid rounds. Best M. Dist.: 172 m	Best M. Dist.: 1321 m		-
Predicted Detection Curve					
3:ip Scan %		29%	27%		
Tighnos Anch.	1323.9 65% meters	532.0 meters	o s	-	
Niss Ti Op. Pred.	1700.0 meters	950 meters	233%.0 2259.0 meters meters		
Range to Wea.	10.65 Km	9.55 Km	8.8 Km		
Wea. Type	105mm How.	105ш Ном•	105шт		
Test Type	Lo- cate	Lo- cate	Lo- cate		
Test #	12/1	12/2	£/21 F-3-14		

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Comments	Firings 13/1 thru 13A/2 all show a normal 5-7 second delay because of the velocity notch. The plotted and machine miss distances are normally comparable, as they should be. Best M. Dist.: 591 m	. Dist.: 332 m	. Dist.: 467 m	. Dist.: 126 m	Dist.: 155 m
	Firings 13 5-7 second notch. Th distances they shoul Best M. Di	Best M.	Best M.	Best M.	Best M.
Predicted Detection Curve					
Blip Scan %	36%	19%	19%		29%
Distance Mach. Pred.	591.2 meters	332.1 meters	467.1 meters	210.4 meters	149.5 meters
Miss D Op. Pred.	680.0 meters	440.0 meters	450.0 meters	216.0 reters	173.0 me ters
Range to Wea.	11.5 Km	155mm 11.6 Кm Ноw.	11.6 Кт		10.15 Km
wea. Type	155mm How	155пт Нои.	155mm How•	155mm 19.15 16w• Km	1.55mm How•
Test Type	Lo- cate	Lo cate	Lo- cate	Lo- cate	Lo- cate
76.51 Seq. 27.	13/1	13/2	F-3-15	13A/1	13A/2

	Consments	Best M Pist: 81 m	The firings 14 thru 15/2 show virtually no detections possible, and none were observed. These firings were essentially out of range of the AN/TPQ-31.			Although more than 50% of the time of flight is predicted to be detectable, none of the firings were seen because of poor signal to noise ratios.
	Predicted Detection Curve	3 0 0 S S C	20 00 00 00 00 00 00 00 00 00 00 00 00 0	20 01 02 02 03 04 04 05 05 05 05 05 05 05 05 05 05 05 05 05	50 00 00 00 00 00 00 00 00 00 00 00 00 0	
	Scan %	367.	0	0	0	0 '
	Mach. Pred.	120.0	t	1	1	1
	Miss Distance Op. Mach. Pred. Pred.	68.0	1	,		1
	Range to Wea.	10.15 Km	29.7 Km	29.7 Km	29.7 Km	16.05 Km
	wea. Type	155	Check 175mm 2	L75mm Gun	175mm Gun	8.0" How.
	Test	Lo- cate	Check Out	Con- firm	Con- firm	Con- firm
	Test #	13A/3	14	F-3-16	15/2	15A/1

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Concents	red ver Ar ibut	Best M. Dist.: 532 m Prediction detection for 16/1 and 16/2 are very poor as these rounds were in the clutter notch and high in the radar beam.	Best M. Dist.: 93 m	Although very poor detectability is indicated for these firings because of clutter notch, decent miss distances were attained helped by the remarkably blas free situation in polar plot Appendix 1, Anner. T. Best M. Dist.: 128 m
Predicted Detection Curve		30 20 10 10 10 10 10 20 30 30 30 30 30 30 30 30 30 30 30 30 30		30 04 04 04 04 04 04 04 04 04 04 04 04 04
Blip Scan		0	25%	10%
Distance Mach.	816.2	ı	136.3 meters	155.0 meters
Miss D	1024.0 meters	1	416.0 meters	280.0 meters
Range	16.05 Km	17.1 Km	15.4 Km	5.6 Km
Wea.	1ype 8.0" How.	8.0" How.	8.0" How.	175 mm How.
	Type Con- firm	Lo- cate	J.o. cate	Con- firm
		1/91 F-3-17	16/2	17

The ratio of the rest of the r

Comments	The firings 18/1 thru 18/4 show good detectability. The miss distances were adversely affected by azimuth bias particularly 18/4. See polar plot, Appendix 1, Annex F.	Best M. Dist.: 139 m	Best M. Dist.: 148 m	Best M. Dist.: 152 m	
Predicted Detection Curve	25 O O O O O O O O O O O O O O O O O O O	\$ 64 H	35 00- 1-02 01 00 00 00 00 00 00 00 00 00 00 00 00	00 00 00 00 00 00 00 00 00 00 00 00 00	
Blip Scan	63%	55%	%05	41%	
Distance Mach. Pred.	165.0 (mef.ers	161.6 55% mcters	195.0 50% meters	294.0 41% meters	
Miss D Op. Pred.	180.0 meters	130.0 meters	175.0 meters	300.0 meters	
Range to Wea.	5.4 Km	5.4 Km	5.4 Km	81 mm 5.4 Km Mor- tar	
Wea.	4.2" Mor- tar	4.2" Mor- tar	81mm Mor- tar	81 mm Mor- tar	
Test	1	Con- firm	Con- firm	Con- firm	
est in Sec.	18/1	18/2	° 6 81 F-3-18	18/4	

Appendix 4 - (Test Analysis Findings) to Annex F to Report of AN/TPQ-31

Performance During Evaluation as a Hostile Weapon Locator

TEST ANALYSIS FINDINGS

TEST 2

Radar problems - See Appendix 5, Annex F.

TEST 2A

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These firings are characterized by non-detectability for approximately the first 6 seconds after firing. The detectability was low due to the distance to the weapon. A glance at the polar plots shows strong azimuth biases for 2A/2 to 2A/5. Since there are no mask angles, the machine and operator predictions must agree within the limits allowed by map plotting accuracy. Because of such unavoidable errors, the operator predictions should have miss distances generally showing equal probability for being greater or less than the correct value. The operator values do exhibit such a distribution. The low blip-scan percentage is consistent with the generally weak detectability prediction. It should be noted that operator miss distances were based not on the number of target blips N, but on the total number of scans during the time the target was in flight. As a result, some scans are counted prior to target detection by radar. On confirm tests the blip scan count was started from the moment the crew was given the "round away" signal, and on the locate test the blip scan count was started from the radar first detection. If there was no radar detection on the confirm test a "0" was entered in the blip scan column, when the radar made a detection a "l" was entered. The "N" for computation of backtrack is the number of scans

between the first "1" and last "1", not necessarily the first scan the target is airborne to the last "1". This leads to a slight error in the operator miss distances but does not significantly affect the overall results.

TEST 2B

Antenna Problems Test 2B - This test was missed due to a problem with the Antenna. The first sign of the trouble was a burnt out wire in the distribution box. The wire was in the antenna drive motor neutral line. When the wire was replaced, the antenna would rotate but was a little reluctant to start on some occasions.

TEST 3

Short Range Display Problem, Tests 2 & 3 - Generators were a persistent problem during the test. The generator would run for a while and shut down. It was after such a generator failure that an intermittent fault in the short range display was experienced. When the system was cold the horizontal sweep was short, and unstable. After a warm-up period of approximately 30 mins., the display would settle down and function correctly. With the amount of time available to troubleshoot, the problem was not found. Cards were changed that were within the horizontal sweep and cursor circuits but the trouble remained. At the end of each work day the truck which housed the console was driven back to the motor pool at Fort Sill. This continuing agitation may have worsened the situation. The yoke driver assembly was a little burnt in the area of the 10 lW series resistors. This assembly would have been changed or the drive

transistor replaced had the test not come to a close. In all probability this assembly was the cause of the problem.

TEST 3A

The state of the s

These tests show a high p. edicted probability of detection over the complete trajectory and the observed blip scan ratios are correspondingly high - nominally 90%. The machine miss distances range between 292 meters and 528 meters, primarily as a result of a somewhat higher than average azimuth scatter. The operator predictions nominally fluctuate around the machine predictions as they should.

TEST 3B

These tests were at an initial range of 23 km ending at 18 or 19 km and therefore show prediction curves generally below the threshold. The blip scan percentages were low (nominally 20%) as expected and the miss distances were correspondingly high, nominally 800 to 900 meters. The polar plots reveal relatively low biases which kept the miss distances from being much higher. This test was adversely affected by antenna malfunction. The

TEST 4

drive motor was removed from the pedestal and inspected. The armature bearing at the brush end was partially seized. This seizure was caused by the heat generated by one of the brushes. The brush was worn and the spring tension so small that the resultant arcing caused the heat. The motor was overhauled, armature bearing replaced and the bull ring bearings cleaned. After reassembly, the antenna functioned without any further problems.

TEST 5

In test 5, all prediction curves show the initial trajectory to be velocity blind. Test 5/2 simulations never rise above the threshold. Excellent correlation was obtained in the test results. The blip scan ratio for 5/2 was 0% and between 30% and 80% for the others, with seemingly close correlation to the simulations.

machine predictions resulted in a significant improvement of miss distances. They range from 1693 (opr) vs 1173 (machine) to as good as 1106 (opr) vs 230 (machine). The firing data as listed on the operator's data sheet (F-5-22) are in reverse sequence order. Tgt 319 is Sequence 1, Tgt 307 is Sequence 2, etc.

Test 5 had a 1.0° mask angle and consideration of this fact in

TEST 6

The simulations put the test 6 trajectory in the velocity notch and this correlates with the observed blip scan ratio of 0%.

TEST 7

Test 7 results correlate well with the simulations except for 7/2 which came out better than expected. Both 7/2 and 7/4 started in the velocity notch. Once again, the machine miss distances were dramatically better than the operator miss distances because of the inclusion of mask angle correction.

Because of several different firings in the area during Test 7, operators had some difficulty picking the correct targets. This may account for some scatter in the results.

TEST 8

Test 8 simulations predict good detectability and such was the case (77% blip scan ratio). This test had a mask angle and the machine miss distance is less than half the value of the plotted miss distances. The biases were small and the corresponding

miss distances should have been low. A study of the polar plot was made, and one possibility is that wrong rounds were tracked for a part of the time. The observed projectile range rate was checked against the simulations, and a considerable discrepancy was found with four (4) rounds which could have originated from a non-participating firing point.

TEST 9 Test 9 results show peculiar discrepancies in several different ways.

- 1. The observed blip scan ratios do not correlate well with the prediction plots (except for sequences 9/5 and 9/6 where the targets should not have been detected due to blind speed).
- 2. Seq 1 shows a miss distance of 3.100 km using given FP-347, but if in reality they were fired from FP-387 the error is reduced to .755 km due in part to the masking of Dodge Hill. The polar plot shows this could well be the case for Seq 1.
- 3. Seq 2. This group shows a dispersion due to the masking of Dodge Hill during the flight. Some detections were made at the early part of the trajectory (10.2 km) but most were grouped around 8.5 km where they came out from behind Dodge Hill. The radar had approximately a 1 second look at the early part of the trajectory before the masking effect of Dodge Hill was encountered.

- 4. Seq 3 shows good results although the detection curves are marginal.
- 5. Seq 4. Only two rounds were recorded, and they are widely scattered. The detection curve is marginal and it is a good probability that none of the rounds of Seq 4 were actually detected.

It was therefore concluded that Test 9 data is interspersed with tracks of projectiles which do not correspond with the scheduled firings. Since Test 9 is a locate test, all operator miss distances are based on taking the difference between the plotted backtrack point and a best known firing point. For Test 9, this best known point never corresponded to the actual firing point and so the operator miss distances listed are not valid.

TEST 10

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The results for 10/1 show a relatively wide range and azimuth scatter in the polar plot and this is reflected in the miss distances 563 meters plotted, 594 meters machine. The prediction plots show poor detectability (due to the clutter notch) and this contributed to the scatter.

Test 10/2 is predicted to be in the clutter notch for all but the last 4 seconds.

However, the generator stopped before Seq 2, the loss of keep alive voltage led to a loss of the receiver crystals due to radiation from co-located L-band radar and no blips were observed.

TEST 11

Tests 11/1 and 11/2 are almost completely in the clutter notch and showed a blip scan ratio of 20% and 11% respectively. The correlation with the prediction curve is good. The rounds in Seq 11/1 had a better chance of being detected earlier than those in 11/2. This was indeed the case, and a corresponding improvement in miss distance was realized. Tests 11/2 was 3032 meters compared to 867 meters for 11/1. Once again a mask angle correction improved the machine results.

The narrow clutter notch was used for Test 11, and a detection curve based on the narrow notch corresponds well to the actual blip scans experienced.

TEST 12

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Test 12 detection curve predicts no detections due to velocity notch and yet the target was seen (65%, 29% and 27% of the time for 12/1, 12/2 and 12/3 respectively). Tests 12/1 and 12/2 were masked. The mask correction used in the machine prediction produced significant improvements - 1700 meters (operator) to 1324 meters (machine), and 950 meters (operator) to 532 meters (machine).

The azimuth biases for 12/2 and 12/3 appear excessive on the polar plots, due in part to operators experimenting with leading edge and trailing edge of the painted blip. A test of projectile velocities compared with simulations lead to the conclusion that at least two tracks of 12/3 were not the scheduled projectile, therefore explaining the very large miss distances of 2334 and 2259 for operator plot and machine calculations.

TEST 13, 13A Test 13, 13A detection curves show that all rounds are detectable after a nominal 5-7 second delay due to the velocity notch.

Test 13 mask angles were only 0.2° and had no significant effect on the prediction. The operator plotted and machine miss distances are nominally comparable, as they should be, and range from about 100 meters to about 600 meters.

TEST 14

The predicted detection curves show virtually no detectability.

These firings were also essentially out of range of the

AN/TPQ-31, resulting in no detections being made during this test.

TEST 15 Same as above.

TEST 15A

Tests 15/A and 16 prediction curves were based on assumed data for a 8" howitzer, which apparently generated overly optimistic possibility of detection. Test 15A/1 curves predict detections over more than 50% of the time of flight. The signal to noise ratio was low and no detections were made.

Test 15A/2 curves show a somewhat improved situation. The target was approximately 10 dB above MDS for 50% of the flight and a 28% blip scan was recorded. The polar plots show a considerable azimuth bias which contributed to the machine miss distance of 816 meters.

TEST 16 Test 16 detection curves are generally very poor due to the target being in either the clutter notch or high in the antenna beam. Thus test 16/1 blip scar is 0 and 16/2 is 25%. Test 16/2 was masked and the machine predictions were significantly better than operator plotted predictions.

TEST 17

Test 17 detection curve predicts very poor detectability (clutter notch), which was confirmed by the 10% blip scan ratio. The polar plots of Test 17, however, show a remarkably bias free situation. This, plus the early detection of the target gave machine miss distance of 155 meters, even though a small percentage of the projectile flight was detected.

TEST 18

Test 18 curves show good detectability for all sequences and blip scans ranged somewhat lower than expected. Miss distances were very satisfactory, however, between 161 meters and 195 for the first 3 sequences. The polar plot for 18/4 shows a significant bias and led to a miss distance of 294 meters.

Appendix 5 - (Recorded Test Data from Ft. Sill Tests) to Annex F of Report of AN/TPQ-31 Ferformance Evaluation as a Hostile Weapon Locator

± .		23 SAN 73	3	=	: 27	, I	23 JAN 73 PM	3	d. 11	5	-	24 JAN 73 AM	=	:	; ;	Sale.
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: .	RANGE TO WEAPON	Σ	8.2 KM	8.2 KM	8.2 KM	8.2 KM	1946 13つで以	13.25 KM	13.25 KM	13.25 KM	198.2 13.25 KM	4.45 KM	187.4 4.45 KM	4.45 KM	4-45 KM	209.0 4.45KM
	m.v.	194.6	1.591	210.0	167.0	209.0	194.6	219.8	183.8	209.0	198.2	176.5	187.4	198.5	212.6	209.0
, , , , , , , , , , , , , , , , , , ,	CHARSE	17.0	124	19.2	13 3	0-61	0.71	20\$	158	19.0	17 4	14.8	0.91	17.48	19.4	0.61
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1	TYPE	4.5"	4.2"	4.5"	4.5"	4.2,	4.2	4.2"	4.2	4.2	4.2	4.2"	4.2"	42"	4.2"	C·+
NUMBER	ROUNDS	ج	5	5.	5	5	5	5	5	5	5	5	5	5	5	5
なる		1/2	2/2	2/3	2,4	2/5	2A/1	2A/2	24/3	2A/4	2A/5	28/1	28/5	28/3	28/4	38/2

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	m.v.	169.3	.183.8	219.8	219.8	!87.4	158.5	205.4	187.4	212.6	187.4	230.6	227.0	245.0	234.2	234.2
٨٥	CHARGE	13.4	~ 41∞	20.4.	20 84	16.0	12.0	184	0.0	198	0.91.	22.0	2. 4%	24.0	52 41%	200 400 800 800 800 800 800 800 800 800 8
WEAPON	QE M/25	900	900	900	900	900	906	900	930	900	900	900	900	006	900	900
	TYPE	4.2"	4.2,	4.5."	4.2,	4.5"	4.5"	4.2"	4.2"	4.2"	4.2"	4.2,	4.5	4.5,	4.24	4,2
NUMBER	ROUNDS	5	5	5.	5	5	5	2	ζ,	5	5	5	7	7	5	6
LEST .	1 Cas	3/1	3/2	3/3	3/4	3/5	34/1	30.12	5-378	3A/4	3A15	38/1	38/2	38/3	38/4	38/5

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TEST # 3	· Hi	ring Stata	•		
FP-524	24 :	วีคม 13	. 4.	2" Moa	TAR
TGT CH I	EF RAI	WGE MC	<u> 177V</u>	QE	TOF,
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FP - 4/5 (30 ERST)	26 JAN 73	4.2" MORTAR
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TGT	<u>CH</u>	DEF	RANGE	100	Ven	QE	TOF
426	12	3003	2040			900	-24
405	184/8		3180	** ***		900	3 <i>0</i>
453	16	2863	2730		: ·	900	3 8
430	194/8	2707	3310		•	900	31
439	. 16 .	2675	2700			900	28

Lest # 38 Hiring Deats

F	P-174	(200 merses:	(ω_{est})	24 JAN	73	4.2" MORYAR	•
: : 工	GT	CH	DEF	RANGE	Atmo	mV QE	TOF
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	7 3	- 21.4/8	2669	3690	175°		33
49	11/5767	24	2653 _.	4170	4	900	35
22 년 	c) ;	2248	2710	38.60	20:5	900	3 <i>H</i>
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. F-5-18 PRINECTILE #2. "MONTHA 0 F 2000 CHAKGE.

ग्टोडीक्रीडीक्रिग्नीछि छिट्टाहर्टेट्येस्स्टिड्येट्रोक्टिक्रेक्टर्ग Tost 3\$ 100. SCAN BLIP 1101 89 567 \mathcal{Q} 4. 3 4565 3570 4580 3560 4545 3525 4240 22.3 4250 215 4235 22.6 4545 3525 3500 45-45 3590 0256 3580 4555 3525 4655.3335 4530 3570 \$X 255 -54-54 PREDICTED PREDICTED IMPACT 45.30 3550 45403535 4550 3530 4228 22.4 4555 3515 METERS AZ MĽŠ 4530 2474 4550 22.7 225 AAWGE FAROR- 456 4225 22,5 4260 22. 2 42 45 21. 2 4260 22.4 4259 22.5 22.6 4235- 22.6 4240 22.5 4270 21.3 4252 22.3 47662213 4250 22.5 RG KM 7260 23.4 22.7 4225 22.5 RIE! - ABITTO-31.0CAT: 011 646 ORIG - 92% 4250 4240 AZ ML's %35 4225 .5426 ; 4260 31.4 7220 20.4 4260 21 3 4250 21.5 FIRST LAST DETECTION DETECTION 4250 21.3 RG እ 4250 21.4 4250 211 4230 20.1 4220 20 4 4250 21.4 ALIP SCHW RATIO 4285 21.1 AZZmuTH AZ Mis 4340 22.0 22.2 23.2 4340 22.3 4230 22.4 4240 22.3 4235 22.2 4335 23.2 å∑ Ž 4350 22.3 4240 22.2 425.022.3 A-4247 33, S 4260 25.2 R-22.7 AVERAGE 22.3 8H1 dy AZ His 4260 4265 1250 4235 135dH240 7330 W47 91 869=25 02co

HEAT # H FINING BLOTON

FF-1335 2 FEB 73 WENDON CLOSSEFECATION TEST

OFFICH 1GT CH DEF RANGE DO DO DO TOF

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4.2" 320 20 3110 3420 900 31

175 422 1 3234 8080 . 236 22

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105

Main circuit breaker tripped. Visual inspection should antenna Control CB tripped. No other united symptoms in static condition. Reapplied power. With antenna turing an unusually large amount of current was drawn on the B3 input. System will not time out after 15 minutes. Inspection of the antenna pedestal should oil leakage around symplical and terminal board. Suspect ail seal sooked drive motor and terminal board. Suspect ail seal sooked drive motor and terminal board. Radar removed from the field.

Lest #5 I way Dutos	
FP-300 30:57N 73 105mm	
TGT CH DFF MINGE MO MY QI	<u> Tof</u>
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450 6 3203 6280 Sug 4 32	ર ટ્ર
410 6 3295 5750 Sy 3. 288	5 20
307 1 1817 2330 542 317	12
319 3 1708 3350 - Sig1 324	15

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PROJECTILE 10520

7EST#5 SCAN BLIP 0 00 Ō 0000 000 0 0 000000011 00000000 000 00 00 ō 0 .7 67 0 0 2345 8-10,450 A-3753 PREDICTED PREDICTED FP 300 **FOOT** 000 0 70 0 Ċ 0 IMPRCT RG KM 383me1+05 25.3 m MLS AZ RG-ML's KM 38.20 10.5 37**3**6 10.49 3645 1035 36.30 10.36 3880 8.85 3725 9.0 585 3716 10.17 3,8 3,98 ORIG 8.8 0018 ERRDE PATIO **ERROR** 2640 3710 4.2 DETECTION DETECTION 4.3 4.4 3.8 4.4 MI'S KM 9:1 3.8 7.7 AZIMUTH 4.4 RANGE LAST SCAN 4 119 NOT SEEN O.4 3450 3700 16.5 3500 36.20 3410 7.8 3220 5.8 34.20 9.8 3500 10.0 3580 Ξ, 37.0 8.3 3520 : 9.4 -W MUERAGE FIRST HIS KH AVERAGE BLIP 3600 10 13,13710 11 13,13,130 13 13,13,130 13 13,13,1370 1,53410 11, 34.20 9 32,3280 7 334 3700 13. 34.10 7

Autte Luing Bria FP-380 31 JAN 73 105mm

TGT CH DEF BANGE 100 MV QE TOF

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PROFILE 105 mm

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Jest #8 Hiring Diota FP -300 30 JAN 73 CH DEF BANGE MO MV QE TOF 437 5.860

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AZ RG AZ RG
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7.est #9. Hiring Data ALL ROUNDS TO TOT HAGE 31 JAN 73 105mm FP CH DEF RANGEATONO MV GE 4. 2758 6590 ns 332 371 2 303. 5 3161 5710 165 371M 356 21 387 4: 2776 4450 180 370 417. 21 3203 2960 55 378 276 .5. 422 3: 2752 2800 7.0 NOOMA 297 13 13 .. 121 2 3500 2430 26 330 264 12

Firing Points for Sequences 3 and 4 above are reversed.

Therefore, accompanying data sheets, Sequence 3 and 4, refer to
firing points 416 and \$87 respectively.

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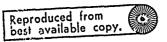
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PRODUCTIVE VOSEM FP = 15014 R= 8750m G = -437... ROWNEY. Neps: 0118 Ballelle CHR. CONTRACTOR STATEMENT STATES THE STATES CHARGES STATES JROW REDICTED REDICTED TEST:#12. + RACINE A 7007 TAACKED Ì 00 13500 5 1123 | 6860 -118 -2350 | -2650 2500 2250 0791- 381- 0601 5211 0.2 0410 - 2289acia 6570 - 227 - 2700 5,1 1/048 8540 -253 - 12300 0012 | 522- 0199 6940-1-153 1-2100 7090 -150 -2000 7250 -123 -1250 -19:94 122 670 - 221 8360 -269 : 16650 -245 6110 -333 RATIO ERBOR ERROR 1077 1032 8960 4.9 1056 11151 5.0 Prest (Met 6.5 0730 5.2 以 以 注 5.4 0800 5.2 0750 5.1 1,2 0460 AZ IMMTH 0770 5.1 1,2 02/0 8,0 0780 5. T.S.Y. RANGE SCAN 0525 0770 0510 7.9 AZ 6730 020 1030 82 8'9 0811 4 1050 6.4. 2'0 常空 1,30 6.9 7.3 8.2 03 0560 AVERAGE 2.9 050 RYFRAGE 4.1150 1050 1130 BITP 1020 1190 1001 77.5 1000 : 1



Hest # 13 Horing Rata

FP-710 27 FEB 13 155 mm

SEG# TGT CH DEF RADIGE 100 101 QE 705

1 2713/ 12° 6 4714 9220 1599 463 416 29.9.

2 2655/ 26° 6 4803 9760 1514 463 458 32

3 2700/ 10 6 4130 9370 1659 463 428 31

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SCAN R-1327 Pzwecnyt-Q E #2-458 DETECTION ORIS IMPRICT SEGIL, #2, #3 BLIP AZ RG RG RG (CONFIRM) CHARGE #2- 6 ololdolo 0001111000 0000 000 C. 13.10 AWITER-31 LOCATION 2399/3184 DATE 27/28/3 000 ololo 001 SEG ZEHI SEQ#2 10840 3.7 1360 10950 SEG#1 SEQ#3 55% <u>५०४३३६</u> WHICH IMPS 00 EACH 1330 10.7 10690 4.1 11344 11150 11210 CAUSE RATIO OF NOWY 1358 ROUNDS FIRST LAST DETECTION DETECTION AZIMUTH ERROR AMOUNT 0910 3.7 THE LS MOT WAS RECORDED SCAN EIRKI SCCURED 50 mais 350 10,5 1350 10.8 FRROR TIME RLIP CAMO! 以京 46 अनळनव 44 M

Test 13A Living Kata

FP-705 25 FEB 73 155mm \

SEGN TGT CH DIF 12:NGC MO MV QE TOF

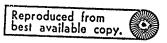
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Heat 16 Hime Plata.

ALL ROUNDS TO TGT 745 22 FEB 73 8"HOW.

SEQ# FP CH DEF RANGE MO MV QE TOF

1 25 7 4890 16,200 3198 671 55 9.5°

2 23 7 4900 14,500 2573 518 44 11°

OF 5/8 PRINECTIVE 8 HOW SCAN: PP.23 <u>ुार्डामीयित्र</u>्यस्त 0 0 1.00 81.1P 0 0 ຍ C 0 0 0 ō 0 0 0 00 0 0 Ö 0 : ٥ 0 0 00 0 0 00 0 0 0 00 DATE 22/EB 23 CHARGE 000 00 ō 0 Ō١ 0 0 0 0 7-455.#16 SEQ#2 000 00 0 0 10 0 G 0 Ö 0 0.0 0 H PREDICTED THORY AZ RG 1-520/ -380 097 -510 7360 1496 15420-14 +140 1487 14560-23 -960 30 (mierens - H. 7mils 40.20 -15 -30 46 1476 15.410 -34 1476 15,400-34 Ø 09451 0191 PREDICTED \$ 1490 14960 1516 15,000 15.620 1487 1956 RATIO FRROR ERKOR A7. CALBR BULTED-31 LOCATION X X 3 4 FIRST LAST
DETECTOR DETECTION 13,3 0620 3.7 AZIMUTH ω ∞ 3.7 3.8 RANGE SCAN RG AZ KM MIS SEEN 0550 SEEN 10550 14.6 10660 0250 0600 5050 0550 14.6 15.0 15.2 15.0 0.5/ : 19.2 **NVERAGE** SLIP 1885 1505 NOT A2 His 1475 0241 0661 NoT 1470 一十二十二 ドローのミュー <u> च्यानम्ययम्बद्धारा त्रारा स्थाप्य स</u> **200**

Lest #17 Living Data

first sequences were scheduled. Only the first sequence was fixed. Hhis was due to problems which occured with the que.

FP-30 SEQ#1 6MAR 73 175mm

TGT. CH DEF. RIANGE MO MV QE TOF

O' MASK ANGLE

TG7: 4910 31,2 Km.

O.T. ON ROWING -31. LOCATION SAN 1379 DATE MINDA 13 ICHARGE

PROJECTILE 175mm 1 8LIP SCAN A-49377 ; PREDICTED FREDICTED TEST #17
SEG = 1MPTCT SEG = 1
AZ RG RZ RZ RG - CONFERM) 1001 11070 ō 01011 00 \overline{o} <u>0</u> 01 1290 000 +310 01 <u>0</u> +290 i /!! -270 1470 051F -310 1590 +2233mETERS -19.7.H \$278.3280``` 60% -24 : #1--/6 5325 1-16. : 13.8 4905 5650 -32 : 13.9 4903 5825 -35 13.3 4920 6250 -17 13.7 4923 5800 -14 13.8 4921 5325 -16 12.1 1423 5950 -14 12.2 4913 5925 11.2 14923 6125 0025 8164 : ERROR RATIO ERROR PIRST: LAST PETECTION DELECTION PZIMUTH 14905 13.5 RANGE SCAN 26. AZ 4910 4910 1905 47 4900 5067 9.6 4900 218H 519 0064: 5.9 7900 016h 5.7 025 6.5 9.5320 6.1 13.EEV SEEN DVFRAGE DVERBGE : 16.14 ; = BLIP 775 14:5 026h 0265 MOT 4920 7 4920 NOT 4920 4905 ľ = : :

Lest # 18 Having Data.
FP-90 6 MAR 73 4.2" + 81mm
SEQ# TGT CH DEF RANGE MO MV DE TOF
1. 50.136 1918 4825 3350 1155 212.6 900 31
2 5. 136 7998 4693 3290 1130 210 900 31
3. 18 7. 4868 3580. 1000. 230 1004 .29
4 11 7 4758 3560 1000 230 1020 29
5510 310 7.42
O° MASK ANGLE

35.4 2000 08035775 H.2" 18 BLIP SCAN A-5985mils
R-5400 meres 1. C. 84/120-31_ LOCATION 512/318 DATE 644413. CHARGE 19-18. 00 0.0 11/1/100 00 | Property | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Paulities | Pauli 101 16.8 5968 5550 -70 -10 1 76 5956 5530 -79 -160 10 7.3 5671 -30 11 -136merres -23.9mils 89% 5950 5540 -35 5945 5420 -40 . : 5.6 | 5630 | 7.3 | 5976 | 5525 - 9 5650 7.3 5983 5520 -2 : 24 BPM CRITAD ERROR FAROR AZIMUTH MATENAIA LSPEED. 5.5 5650 7.1 5.6 5670 21 5.4 5620 7.6 RANGE SCAN 5690 5920 5.6 5460 5950 5.5 5670 2930 5.6 5650 THE POSE 15960 5.6 85 BLIP 5920 5950 2830 5930

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